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PROPELLANT SURVEILLANCE REPORT. LGM-30 A AND B STAGE 1. TP-H101--ETC(U)

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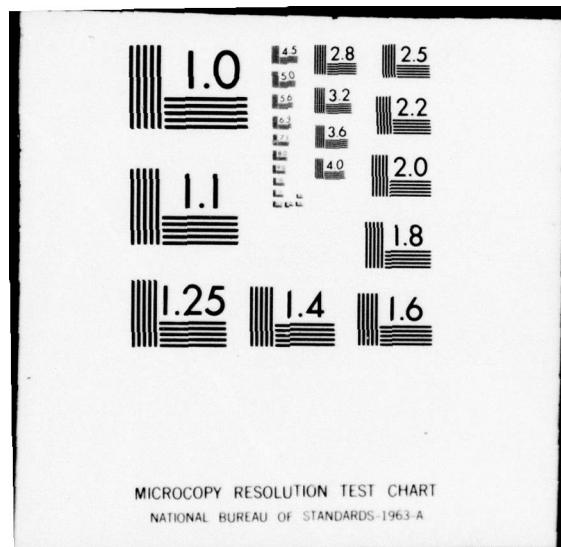
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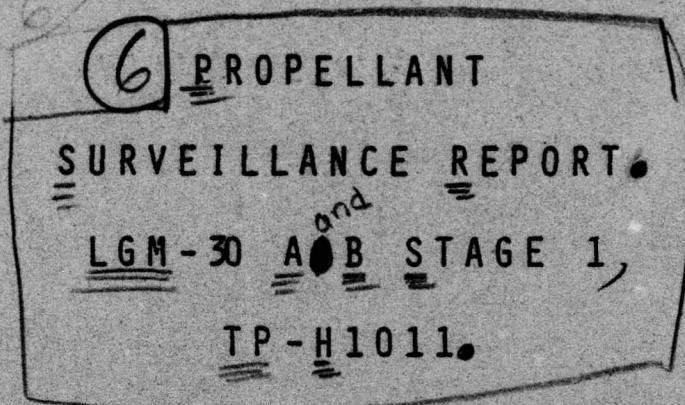
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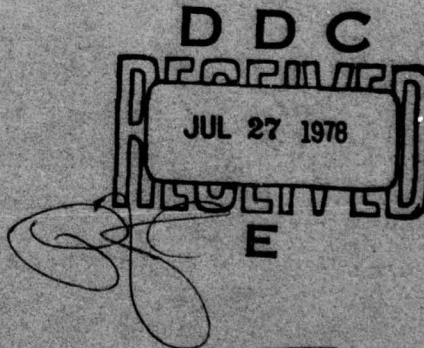
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PROPELLANT SURVEILLANCE REPORT

LGM-30 A & B STAGE I

TP-H1011

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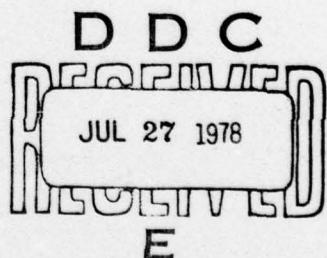
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ABSTRACT

This report contains propellant test results from cartons of TP-H1011 bulk propellant representing LGM-30 A and B First Stage Minuteman Motors. This report is the fourteenth time that a statistical approach has been used to analyze First Stage bulk carton propellant. Testing was accomplished in accordance with MMEMP Project M82934C-WNL17514.

The purpose of testing was to determine and provide early warning of any serious degradation trends occurring in the propellant for service life predictions.

An analysis of all parameters indicates that no potential problems are expected in the propellant for at least two years past the oldest data point.

Data stored in the G085 System were plotted utilizing the IBM 360-65 Computer and CAL-COMP Plotter. The data range at any age can be found by suitable inquiry of the G085 System.

Each point on the regression plot represents the mean of all samples at that particular age. The number of specimens at each point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots.

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29B	Zero Time Test Results	29 Jan 64
29C	Zero Time Test Results (Supplement 1)	30 Mar 64
29D	Zero Time Test Results (Aft Closure)	9 Jun 64
29E	Zero Time (Aft Closure Supplement 1)	24 Jun 64
29F	ATP Phase I Test Results	30 Mar 65
29G	ATP Phase I Test Results	19 Aug 65
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32A	Zero Time, Wings II-V Test Results	17 Mar 65
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61	ATP Phase I, Wings II-V (Fifth Group)	10 Jun 66
66	ATP Phase I, Wings II-V (Sixth Group)	22 Jul 66
76	ATP Phase II, Wing I Test Results	24 Jan 67
78	Zero Time, Wing VI Test Results	3 Feb 67
104	ATP Phase I, Wing VI (First Group)	12 Oct 67
118	ATP Phase II, Wings II-V (First Group)	5 Mar 68

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<u>Report Nr</u>	<u>Title</u>	<u>Report Date</u>
126	ATP Phase II, Wings II-V (Second Group)	11 Apr 68
130	ATP Phase II, Wings II-V (Third Group)	3 May 68
162	ATP Phase I, Wing VI (Second Group)	30 Sep 69
176	ATP Phase II, Wing VI (First Group)	15 Apr 70
181	ATP Phase III, Wing I	7 May 70
185	ATP Phase I, Wing VI (Third Group)	22 Jun 70
195	ATP Phase III, Wings II-V (Retest)	29 Oct 70
223	Surveillance Report LGM-30 Stage I (TP-H1011)	Sep 71
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313	Stage 1 Propellant Surveillance Report, Propellant Containing Glacial Acrylic Acid	Oct 74
315	Propellant Surveillance Report LGM-30 F & G Stage 1, TP-H1011	Jan 75
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321	Propellant Surveillance Report LGM-30 F & G Stage 1, Phase B, Series II, TP-H1011	Apr 75
325	Propellant Surveillance Report LGM-30 A & B Stage 1, TP-H1011	Jun 75
328	Propellant Surveillance Report LGM-30 A & B Stage 1, TP-H1011	Sep 75
330	Propellant Surveillance Report LGM-30 F & G Stage 1, TP-H1011	Oct 75
335	Stage 1 Motor Reliability Improvement Program	Dec 75
337	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1043	Feb 76
339	Stage 1, New MAPO & ERL-510 Qualification	Mar 76
341	Propellant Surveillance Report LGM-30 Dissected Motors, Phase VII, TP-H1011	Mar 76

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<u>Report Nr</u>	<u>Title</u>	<u>Report Date</u>
343	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Jun 76
345	Propellant Surveillance Report LGM-30 F & G, Stage 1 Phase B, Series III, TP-H1011	Jun 76
350	Qualification of a New MAPO Source and ERL-510 Curing Agent for Minuteman, Stage 1, UF-2121 Liner	Sep 76
351	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Sep 76
354	Minuteman Stage 1 Motor Reliability Improvement Program Surveillance	Sep 76
358	Propellant Surveillance Report LGM-30 Dissected Motors, Phase VIII, TP-H1011	Oct 76
360	Propellant Surveillance Report LGM-30 F & G, Stage 1 Phase E, Series III, TP-H1011	Nov 76
367	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Apr 77
370	Propellant Surveillance Report LGM-30 F & G, Stage 1, Phase E, Series II, TP-H1011	May 77
377	Qualification of a New MAPO Source and ERL-510 Curing Agent for Minuteman Stage 1, UF-2121 Liner	Oct 77
379	Final RIP Report, Minuteman Stage 1 Motor Reliability Improvement Program Surveillance	Oct 77
385	Propellant Surveillance Report LGM-30 A, B, F, & G, Stage 1, TP-H1043	Dec 77
388	Propellant Surveillance Report LGM-30 A & B, Stage 1, TP-H1011	Jan 78
390	Propellant Surveillance Report LGM-30 F & G Stage 1, Phase E, Series IV, TP-H1011	Feb 78
392	Propellant Surveillance Report LGM-30 Dissected Motors, Phase IX, TP-H1011	Mar 78

GLOSSARY OF TERMS AND ABBREVIATIONS

Aging Trend	A change in properties or performance resulting from aging of material or component
CSA	Cross Sectional Area
DB	Dogbone
Degradation	Gradual deterioration of properties or performance
E	Modulus (psi), defined as stress divided by strain along the initial linear portion of the curve.
EB	End Bonded
EGL	Effective Gage Length
em	Strain at maximum stress
er	Strain at rupture
"F" ratio	The ratio of the variance accounted for by the regression function to the random unexplained variance. The regression function having the most significant "F" ratio is used for plotting data. The ratio is also used in detecting significant changes in random variation between succeeding time points
JANNAF	Joint Army, Navy, NASA, Air Force
MANCP	Propellant Lab Section at Ogden Air Logistics Center
Ogden ALC	Ogden Air Logistics Center, Air Force Logistics Command
r or R	The Correlation Coefficient is a measure of the degree of closeness of the linear relationship between two variables
Regression Equation	The general form of the regression equation is $Y = a + bx$
Regression Line	Line representing mean test values with respect to time
s_b	Standard error of estimate of the regression coefficient

GLOSSARY OF TERMS AND ABBREVIATIONS (cont)

S_e or $S_{Y.X}$	Standard deviation of the data about the regression line
S_m	Maximum Stress
S_r	Stress at rupture
Standard Deviation (S_y)	Square root of variance
Strain Rate	Crosshead speed divided by the EGL
"t" test	A statistical test used to detect significant differences between a measured parameter and an expected value of the parameter (determines if regression slope differs from zero at the 95% confidence level)
Variance	The sum of squares of deviations of the test results from the mean of the series after division by one less than the total number of test results
3 Sigma Band	The area between the upper and lower 3 sigma limit. It can be expected that 99.73% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed.
90-90 Band	It can be stated with 90% confidence that 90% of the inventory represented by the test samples would fall within this range assuming that the population is normally distributed

INTRODUCTION

A. PURPOSE:

Quality assurance tests have been conducted for fifteen years on First Stage LGM-30A and B Minuteman Motor Propellant blocks to evaluate the effects of aging on TP-H1011 propellant.

Statistical analysis of the tests performed, as directed by Engineering, should provide early warning if serious degradation trends occur. Annual evaluation of the propellant provide data that can be directly input into engineering reliability and service life predictions. Testing was performed in accordance with MMWRM Directive GTD-1C and GTD-1C Amendments 1 and 2.

B. BACKGROUND:

Testing was first accomplished at MANCP on LGM-30A TP-H1011 propellant blocks in 1963 and was designated Zero-Time Testing (MAGCP Report Nrs 29B, 29C and 29F). Subsequent testing was accomplished at approximately 24 month intervals (MAGCP Report Nrs 29G, 29H - Phase I; 76 - Phase II; 181 - Phase III).

LGM-30B Zero-Time testing was accomplished in 1964 with subsequent testing at intervals of 24 months (MAGCP Report Nrs 32A-Zero-Time; 32C, 49, 53, 55, 58, 61, 66 - Phase I: 118, 126, 130-Phase II; 195, 268 - Phase III).

Reports prior to MAGCP Report Nr 223(72) contained raw data using sigma relation to compare to Zero-Time variance. MANCP Report Nr 239(72) published in April of 1972 contained all the

data on LGM-30A, B, F and G in the G085 System at that time.

Report Nrs 258(72), 268(73) reported LGM-30A and B data in statistical analysis by itself. This report is the ninth time that LGM-30A and B data have been reported in this manner.

Zero-Time testing was started as soon as possible after receipt of the propellant by MANCP. Data from these tests were used to establish a base line for each test to which each subsequent test data (ATP - Accelerated Test Plan) were compared in the reports listed above.

The LGM-30A and B propellant test matrix (Table 1) were used to determine the number of specimens to be taken from each propellant loaf and the specific test or tests to which these specimens were subjected. Low rate tensile and hardness specimens were taken from all LGM-30A and B blocks. Specimens for other physical and combustion tests were taken from every seventh block.

Some tests were not conducted at the earlier test periods (0-6 years) and, therefore, data are not available for inclusion in the regressions.

Table 1
Test Program

The test matrix is taken from GTD-1C, Amendment 2, and the tests, conditions, number of specimens and test methods are listed below.

<u>Test</u>	<u>Conditions</u>	<u>Description</u>	<u>Per Cond</u>
Hardness	10 Sec	Dogbone Ends	3
Low Rate Tensile	2.0 in/min	1/2" JANNAF Dogbone	3
High Rate Tensile	1750 in/min	3/4" Dogbone	3
High Rate Triaxial Tensile	600 psi, 1750 in/min	3/4" GL Rail End Bonded	1
Low Rate Biaxial Tensile	0.2 in/min	3/4" GL Rail End Bond	1
Stress Relaxation	3% & 5%	1/2" x 1/2" x 4" EB	3
Dynamic Response	70 gm ct wt	3.3" dia x .33" disc	1
Sol Gel		1/2" x 1/2"	8
VLR	2×10^{-3} in/min	1/2" JANNAF Dogbone	3
Ignitability	168 cal/cm ² sec	.050" wafer	3
TCLE		.200" wafer	3
Pressure Time	500 psi	1/2" x 3/8" x 1"	3
Burning Rate	1000 psi	.156" x .156" x 5" Strand	3
DTA	12°C Rise/min	.040" wafer	3
DSC		.040" wafer	3
Poisson's Ratio	77°F \pm 2° 15% Strain	.50" x .50" x 4"	6
Tear Energy	70°F \pm 2°	0.1" x 1.18" x 3"	6
Failure Envelope		JANNAF Dogbone	3

STATISTICAL APPROACH

In order to determine aging trends for shelf/service life predictions, as directed by Service Engineering, First Stage LGM-30A and B Minuteman Motor propellant (TP-H1011) blocks have been under-going testing since 1963, statistically analyzed and reported on a regular test cycle by this laboratory.

The primary reason for performing statistical analysis on test data is for the detection of propellant changes due to aging that would affect motor reliability. Regression analysis was the method used to examine data and to aid in drawing conclusions about dependency relationships that may exist i.e., relationship between age versus test results.

The linear model ($Y = a + bX$) was found to be the best fit model for the regressions within this report. The only exception was for pressure time at maximum pressure where the reciprocal of X ($Y = a + b (1/X)$) proved to be the best fit model.

Individual data points from different time periods were used to establish a least squares trend line for the data. The variance about the regression line, obtained using individual values of the dependent variable, was used to compute a tolerance interval such that at the 90% confidence level 90% of the sample distribution falls within this interval. This tolerance interval was extrapolated to a maximum of 24 months into the future from age of the oldest motor tested. The 't' values and the significance of this statistic, which are reported for each regression model, give an indication of the 'statistical significance' of the slope of the trend line as compared to a line of zero slope.

Data were plotted by computer. The 'y' axis is computed so that the values at one inch intervals are peculiar to the data spread of the parameter tested. Plotted data points represent means at the particular ages at which testing occurred. The number of specimens at each age point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots. Variance at each test age can be determined by consulting the G085 data storage system.

In a few cases, a small change has become apparent in data variance and regression trend lines. However, the changes are gradual and no operational problems are expected at this time.

A post cure effect (propellant stabilizing after the first year or two) has been observed on some of the early test data (pressure time, low rate biaxial tensile, high rate tensile, and high rate triaxial tensile) which tended to bias and skew the projected trend lines. To overcome this factor, two methods of analysis were performed: First, where possible, non-linear models were used that would best fit the total data (pressure time [max pressure]); Second, where non-linear models were not the best fit, this early data was eliminated (low rate biaxial, high rate tensile, and high rate triaxial). By compensating for this post cure biasing, a more accurate aging trend line for service life prediction is provided.

TEST RESULTS

A. TENSILE:

Regressions for very low rate tensile data show a statistically significant decrease in the strain parameters. The maximum stress regression shows a statistically significant increase with the stress at rupture showing no significant change. The regression for modulus shows a statistically significant increase (Figures 1 thru 5).

Low rate tensile for strains show a statistically significant decrease. The maximum stress regression shows a statistically significant increase and the stress at rupture shows a statistically significant decrease. A statistically significant increase is shown for modulus (Figures 6 thru 10).

For low rate biaxial tensile testing, strain at maximum stress does not show a significant trend with strain at rupture showing a statistically significant decrease. The maximum stress regression shows a statistically significant increase and stress at rupture shows no significant change. The regression for modulus shows a statistically significant increase (Figures 11 thru 15).

The high rate tensile strain at maximum stress shows a statistically significant increase. Strain at rupture, maximum stress and stress at rupture show a statistically significant decrease. Modulus shows a statistically significant increase (Figures 16 thru 20).

For triaxial tensile testing, strain at maximum and strain at rupture show a statistically significant increase. Maximum stress and stress at rupture do not show a significant change. Modulus shows a statistically significant decrease (Figures 21 thru 25).

For all of the tensile testing, where changes are shown, the trends are gradual and no operational problems are expected for at least two years beyond the oldest data point.

B. STRESS RELAXATION:

Modulus at both 3% and 5% strain shows a statistically significant increase for all time periods (Figures 26 thru 33). However, the slope of the trend lines are gradual and no operational problems with the propellant are expected.

C. HARDNESS:

There is a statistically significant increase in hardness data (Figure 34). The increase in hardness correlates with the tensile testing data where, in general, the strains show a decrease and the stresses and modulus show an increase.

D. DYNAMIC RESPONSE:

The storage shear modulus at 200 and 400HZ shows a statistically significant decrease while the loss tangent at 200 and 400HZ shows a statistically significant increase (Figures 35 thru 38).

E. CONSTANT STRAIN:

Strain at rupture for constant strain does not show a significant change (Figure 39).

F. TCLE (THERMAL COEFICIENT OF LINEAR EXPANSION):

The thermal coefficient of linear expansion below and above the glass transition point shows a statistically significant increase (Figures 40 & 41).

G. SOL GEL:

The cross link density shows a statistically significant increase with a statistically significant decrease for percent extractables and weight swell ratio (Figures 42 thru 44).

The increasing cross link density trend correlates well with the other physical properties. The tensile testing shows an increase in maximum stress and modulus with strains decreasing. In addition, hardness is increasing as would be expected as the polymer molecules continue to cross link. The stress relaxation, dynamic response and constant strain also correlates with cross link density.

H. DTA (DIFFERENTIAL THERMAL ANALYSIS):

For the DTA regressions the endotherm and first and second exotherms show a statistically significant decrease. The third exotherm shows a statistically significant increase. There is no significant change for ignition temperature (Figures 45 thru 49).

I. PRESSURE TIME:

Maximum pressure shows a statistically significant decrease and the time to maximum pressure shows a statistically significant increase (Figures 50 and 51).

J. BURNING RATE:

The burning rate shows a statistically significant decrease (Figure 52).

This correlates with the increasing time to maximum pressure.

K. IGNITABILITY:

Ignitability shows no significant change (Figure 53).

L. DSC (DIFFERENTIAL SCANNING CALORIMETER):

The endotherm and first exotherm show no significant change (Figures 54 and 55). The second exotherm shows a statistically significant increase (Figure 56).

In general, where statistically significant changes are shown for the thermal analysis the trends are gradual. From this it is concluded that the propellant will remain thermally stable for at least two years beyond the last data point.

CONCLUSIONS

This report includes LGM-30 A and B bulk propellant test results presently in the GO85 System and covers the past fifteen years of testing.

The test results show that under present storage conditions the physical/mechanical and combustion properties of the propellant are remaining relatively stable with age. This is indicated by the regression plots where the slope of the trend line is relatively flat or close to a line of zero slope and have not changed appreciably from the last test period.

From the statistical analyses, all tests conducted indicate that motor propellant reliability will not be affected for two years past the last data point on the regression.

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
15.0	18	96.0	29	121.0	24	147.0	3
17.0	21	97.0	34	122.0	21	152.0	6
18.0	6	98.0	41	123.0	19	153.0	24
21.0	6	99.0	33	124.0	20	154.0	12
73.0	9	100.0	34	125.0	18	155.0	21
75.0	18	101.0	42	126.0	51	156.0	5
77.0	18	102.0	34	127.0	26	157.0	4
78.0	3	103.0	49	128.0	42	158.0	3
79.0	9	104.0	20	129.0	28	160.0	6
80.0	30	105.0	39	130.0	18	162.0	6
81.0	18	106.0	28	131.0	25	163.0	6
82.0	9	107.0	28	132.0	24	164.0	3
83.0	18	108.0	44	133.0	20	165.0	3
84.0	18	109.0	22	134.0	39	166.0	3
85.0	6	110.0	26	135.0	45	168.0	6
86.0	40	111.0	24	136.0	18		
87.0	27	112.0	29	137.0	39		
88.0	23	113.0	41	138.0	21		
89.0	13	114.0	15	139.0	12		
90.0	7	115.0	16	140.0	24		
91.0	36	116.0	27	141.0	33		
92.0	7	117.0	12	142.0	21		
93.0	15	118.0	33	143.0	6		
94.0	22	119.0	27	145.0	3		
95.0	22	120.0	36	146.0	3		

STAGE 1, WING 1E2 VERY LOW RATE CHS=0.002 IN/MIN STRAIN AT MAX STRESS (EM)

This sample size summary is applicable to figures 1 thru 5

$F = +3.2118701E+01$ $\gamma = ((+2.3654353E-01) + (-2.5061728E-04)) * X$
 $R = -1.2923374E-01$ $F = \text{SIGNIFICANT}$
 $\zeta = +5.6673363E+00$ $R = \text{SIGNIFICANT}$
 $N = 1893$ $\zeta = \text{SIGNIFICANT}$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$ $\text{DEGREES OF FREEDOM} = 1891$
 $\text{TEST CONDITIONS} = 77 \text{ DEG/F AMB-RH}$

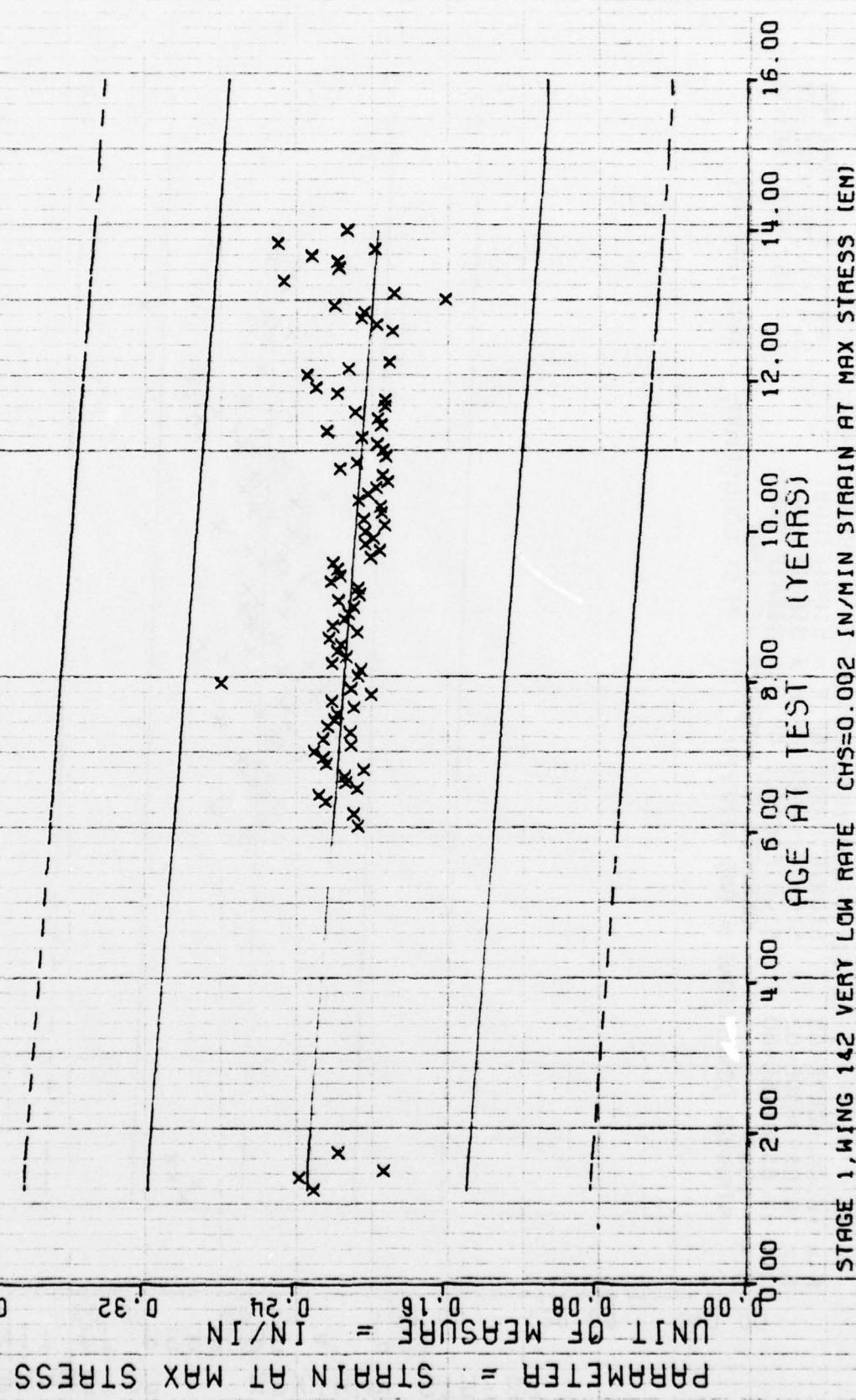


Figure 1

$\gamma = ((+7.9990151E+01) + (+2.0583621E-02) * X)$
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANCE OF } t = \text{SIGNIFICANT}$
 $l = \text{DEGREES OF FREEDOM} = 1891$
 $N = 1893$
 STORAGE CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = PSI
 PARAMETER = MAXIMUM STRESS

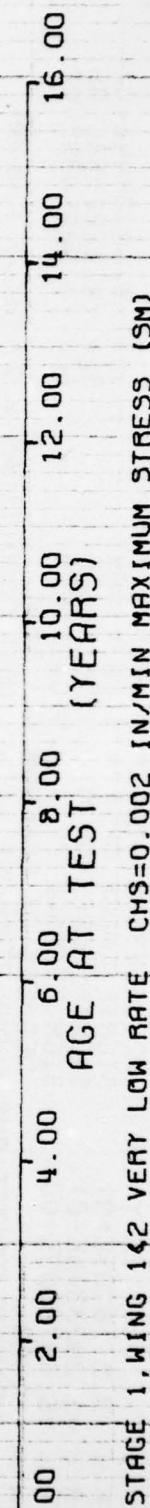


Figure 2

$F = 1.4928007E+02$
 $R = -2.7049299E-01$
 $t = +1.2218022E+01$
 $N = 11893$
 $F = ((+2.4673470E-01) + (-2.3748784E-04) * X)$
 $R = \text{SIGNIFICANCE OF F} = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANCE OF R} = \text{SIGNIFICANT}$
 $N = \text{SIGNIFICANCE OF t} = \text{SIGNIFICANT}$
 $\text{DEGREES OF FREEDOM} = 1891$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\text{PARAMETER} = \text{STRAIN AT RUPTURE}$
 $\text{UNIT OF MEASURE} = \text{IN/IN}$
 $0.00 \quad 0.12 \quad 0.16 \quad 0.20 \quad 0.24 \quad 0.28 \quad 0.32$

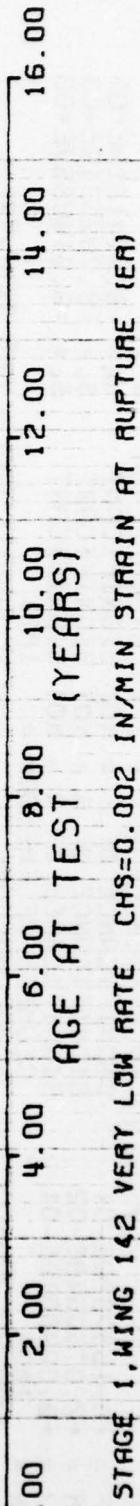


Figure 3

$\gamma = ((+7.6413440E+01) + (+7.9394822E-03) 1 \times X)$
 $F = +8.1537322E-01$ SIGNIFICANCE OF F = NOT SIGNIFICANT
 $R = +2.0760549E-02$ SIGNIFICANCE OF R = NOT SIGNIFICANT
 $\alpha = +9.0298019E-01$ SIGNIFICANCE OF α = NOT SIGNIFICANT
 $N = 1893$ DEGREES OF FREEDOM = 1891
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

40.00 60.00 80.00 100.00 120.00 140.00
 UNIT OF MEASURE = PSI

PARAMETER = STRESS AT RUPTURE

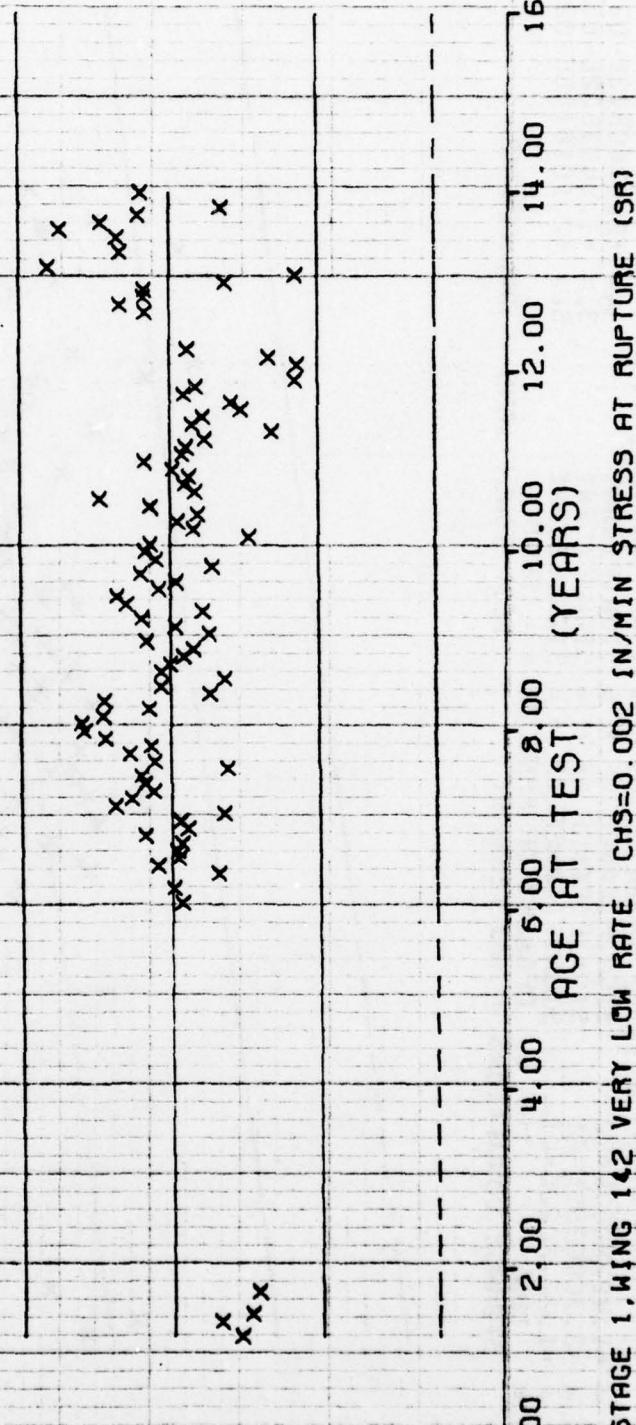


Figure 4

$\gamma = ((+5.2789037E+02) + (+5.0651427E-01) * X)$
 $F = 2.2521082E+01$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = +1.0851543E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $L = +4.7456382E+00$ SIGNIFICANCE OF L = SIGNIFICANT
 $N = 1892$ DEGREES OF FREEDOM = 1890
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 77 DEG/F AMB-RH

UNIT OF MEASURE = PSI $\times 10^3$
 PARAMETER = MODULUS

STAGE 1, WING 142 VERY LOW RATE CHS=0.0002 IN/MIN MODULUS (E)

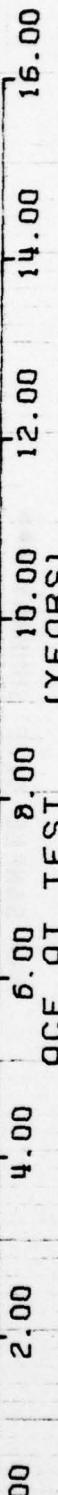


Figure 5

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
72.0	142	97.0	143	122.0	75	147.0	39
73.0	99	98.0	235	123.0	174	148.0	3
74.0	133	99.0	215	124.0	160	150.0	3
75.0	252	100.0	210	125.0	198	151.0	6
76.0	147	101.0	186	126.0	170	152.0	36
77.0	157	102.0	174	127.0	183	153.0	182
78.0	153	103.0	177	128.0	156	154.0	110
79.0	134	104.0	151	129.0	186	155.0	108
80.0	194	105.0	217	130.0	160	156.0	50
81.0	153	106.0	274	131.0	191	157.0	72
82.0	260	107.0	153	132.0	221	158.0	9
83.0	166	108.0	254	133.0	158	159.0	6
84.0	183	109.0	182	134.0	215	160.0	24
85.0	307	110.0	165	135.0	281	161.0	15
86.0	144	111.0	171	136.0	264	162.0	12
87.0	467	112.0	322	137.0	198	163.0	39
88.0	682	113.0	155	138.0	135	164.0	27
89.0	783	114.0	213	139.0	168	165.0	24
90.0	506	115.0	193	140.0	76	166.0	30
91.0	558	116.0	204	141.0	196	167.0	21
92.0	527	117.0	220	142.0	97	168.0	21
93.0	297	118.0	228	143.0	33	169.0	5
94.0	365	119.0	162	144.0	45		
95.0	311	120.0	188	145.0	27		
96.0	212	121.0	238	146.0	3		

STAGE 1, WING 162 LOW RATE TENSILE CHS=2.0 IN/MIN STRAIN AT MAX STRESS (EMI)

This sample size summary is applicable to figures 6 thru 9.

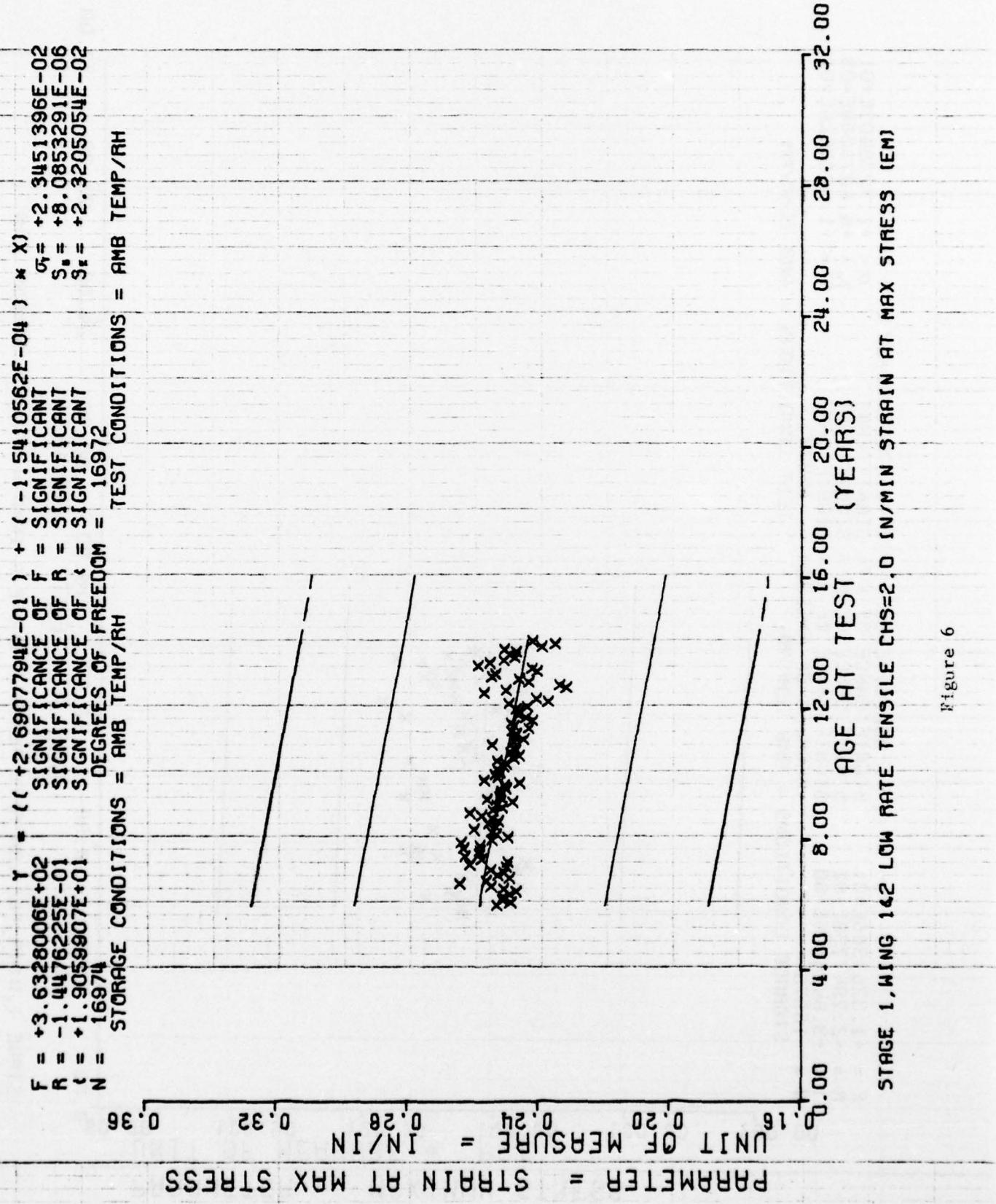


Figure 6

$\gamma = ((+1.3621677E+02) + (+1.6781984E-02) * X) * \sigma$
 $\sigma = +1.3228403E+01$
 $S_F = +1.3223626E+01$
 $S_R = +4.6073089E-03$
 $S_t = +1.3223626E+01$
 $F = +1.3267586E+01$
 $R = +2.7947758E-02$
 $\alpha = +3.6424698E+00$
 $N = 16975$
 $\text{DEGREES OF FREEDOM} = 16973$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

TEST CONDITIONS = AMB TEMP/RH

PARAMETER = MAXIMUM STRESS

UNIT OF MEASURE = PSI

80.00 100.00 120.00 140.00 160.00 180.00

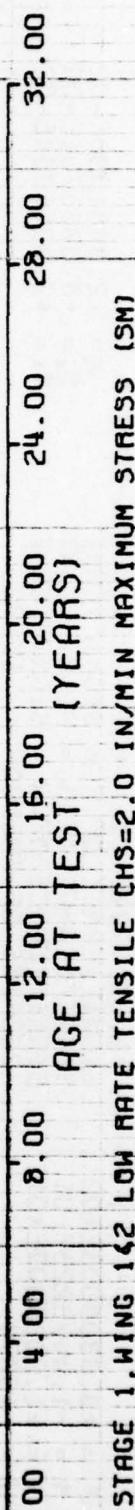


Figure 7

$Y = ((+3.3414919E-01) + (-2.6890430E-04) * X) * S$
 $F =$ SIGNIFICANCE OF F = 8.9313778E+02
 $R =$ SIGNIFICANCE OF R = 2.2359189E-01
 $t =$ SIGNIFICANCE OF t = 9.9885410E+01
 $\zeta =$ DEGREES OF FREEDOM = 16972
 $N =$ STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRAIN AT RUPTURE

UNIT OF MEASURE = IN/IN

0.40

0.36

0.32

0.28

0.24

0.20

0.00

STAGE 1, WING 142 LOW RATE TENSILE CHS=2.0 IN/MIN STRAIN AT RUPTURE (ER)

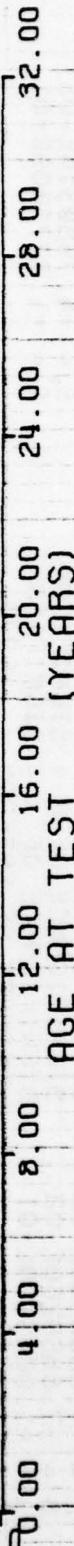


Figure 8

$\gamma = 1.3314763E+02$ $(-4.8196145E-02)$ $(-4.8196145E-02)$ $(-4.8196145E-02)$
 $F =$ SIGNIFICANCE OF $F =$ SIGNIFICANT
 $R =$ SIGNIFICANCE OF $R =$ SIGNIFICANT
 $t =$ SIGNIFICANCE OF $t =$ SIGNIFICANT
 $l =$ DEGREES OF FREEDOM = 16973
 $N =$ 16975 STORAGE CONDITIONS = AMB TEMP/RH

PARAMETER = STRESS AT RUPTURE

UNIT OF MEASURE = PSI

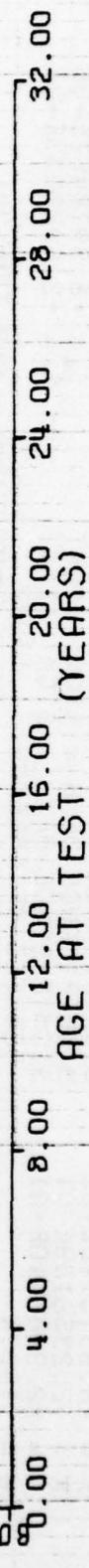


Figure 9

*** SAMPLE SIZE SUMMARY ***

AGE M J N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES	AGE M O N T H S	NR SAMPLES
1.0	1	30.0	391	55.0	24	81.0	153	106	272	138	135								
6.0	4	31.0	307	56.0	18	62.0	260	107	153	139	168								
7.0	33	32.0	215	57.0	30	83.0	166	108	253	140	76								
8.0	53	33.0	114	58.0	24	84.0	183	109	179	141	196								
9.0	49	34.0	91	59.0	24	85.0	307	110	165	111	142	97							
10.0	89	35.0	57	60.0	42	86.0	144	111	171	143	33								
11.0	43	36.0	63	62.0	24	87.0	467	112	322	144	45								
12.0	46	37.0	56	63.0	51	88.0	682	113	155	145	27								
13.0	44	38.0	75	64.0	61	89.0	782	114	213	146	3								
14.0	65	39.0	68	65.0	42	90.0	506	115	193	147	39								
15.0	43	40.0	54	66.0	96	91.0	558	116	204	148	3								
16.0	70	41.0	84	67.0	39	92.0	527	117	220	150	3								
17.0	42	42.0	216	68.0	69	93.0	297	118	228	151	6								
- 18.0 -	75	43.0	241	69.0	97	94.0	366	119	162	152	36								
19.0	127	44.0	158	70.0	89	95.0	311	120	238	153	182								
20.0	65	45.0	194	71.0	165	96.0	212	121	188	154	110								
21.0	85	46.0	136	72.0	142	97.0	143	122	122	75	108								
22.0	51	47.0	53	73.0	99	98.0	235	123	174	156	50								
23.0	75	48.0	36	74.0	133	99.0	215	124	160	157	72								
24.0	338	49.0	27	75.0	252	100.0	210	125	198	158	9								
25.0	210	50.0	17	76.0	147	161.0	186	126	170	159	6								
26.0	251	51.0	16	77.0	157	102.0	174	127	183	160	24								
27.0	391	52.0	21	78.0	154	103.0	177	128	156	161	15								
28.0	354	53.0	6	79.0	134	104.0	151	129	186	162	12								
29.0	517	54.0	3	80.0	194	105.0	217	130	160	163	39								
								131	191	164	27								
								132	221	165	24								
								133	158	166	30								
								134	215	167	21								
								135	281	168	20								
								136	264	169	5								

$Y = ((+9.0383097E+02) + (+1.0662483E+00) * X)$
 $F =$ SIGNIFICANCE OF F = SIGNIFICANT
 $R =$ SIGNIFICANCE OF R = SIGNIFICANT
 $S =$ SIGNIFICANCE OF S = SIGNIFICANT
 $t =$ DEGREES OF FREEDOM = 23703
 $N =$ STORAGE CONDITIONS = AMB TEMP/RH

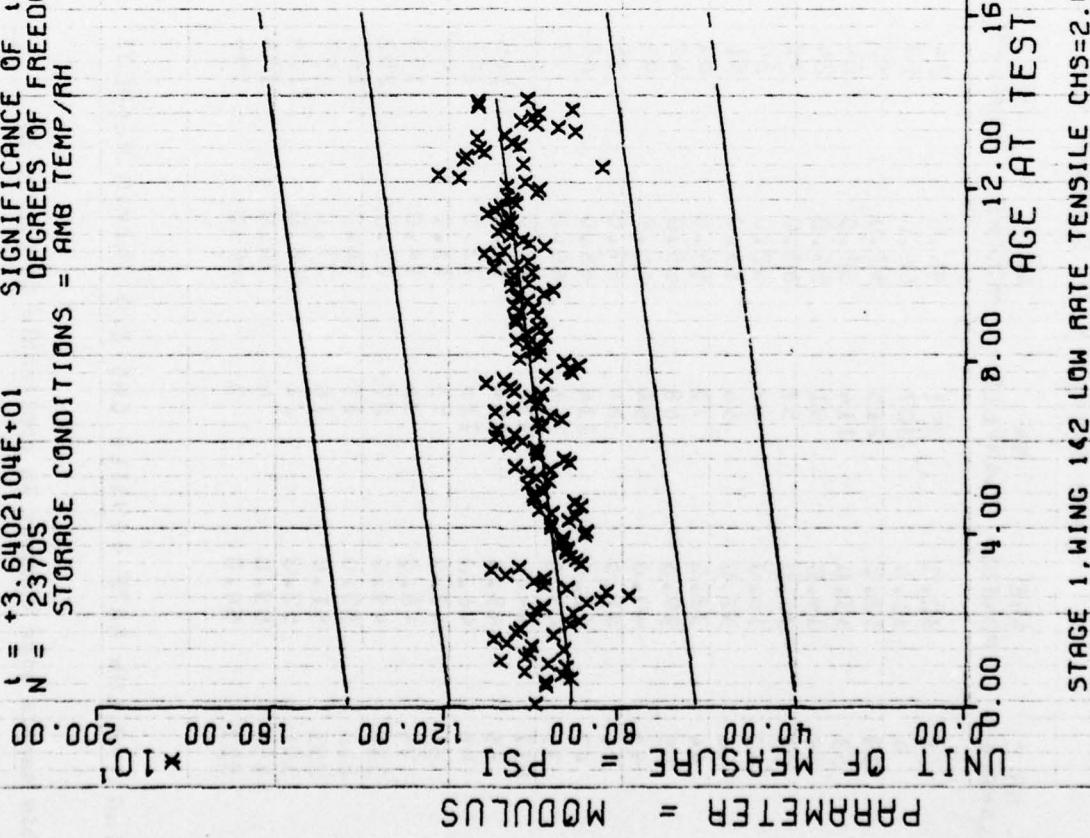


Figure 10

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
72.0	3	97.0	9	122.0	16	155.0	3
73.0	9	98.0	1	123.0	6	156.0	2
74.0	4	99.0	3	124.0	13	157.0	4
75.0	5	100.0	5	125.0	9	161.0	2
76.0	9	101.0	2	126.0	11	163.0	1
77.0	12	102.0	4	127.0	6	164.0	3
78.0	7	103.0	4	128.0	8	165.0	1
79.0	16	104.0	9	129.0	7	166.0	3
80.0	5	105.0	9	130.0	7	169.0	1
81.0	10	106.0	9	131.0	4		
82.0	15	107.0	10	132.0	4		
83.0	13	108.0	11	133.0	4		
84.0	11	109.0	12	134.0	6		
85.0	11	110.0	8	135.0	13		
86.0	18	111.0	12	136.0	16		
87.0	11	112.0	10	137.0	10		
88.0	14	113.0	8	138.0	8		
89.0	26	114.0	6	139.0	1		
90.0	34	115.0	11	140.0	2		
91.0	23	116.0	5	141.0	9		
92.0	37	117.0	8	142.0	5		
93.0	30	118.0	5	148.0	1		
94.0	20	119.0	4	152.0	1		
95.0	11	120.0	12	153.0	5		
96.0	10	121.0	11	154.0	3		

STAGE 1, WING 162 LOW RATE BIAXIAL CHS=0.2 IN/MIN STRAIN AT MAX STRESS (EM)

- This sample size summary is applicable to figures 11 thru 15.

$\gamma = (1.1636035E-01 + (-4.4977947E-05) \times X)$
 $F = +2.3371213E+00$ SIGNIFICANCE OF F = NOT SIGNIFICANT
 $R = -5.5921956E-02$ SIGNIFICANCE OF R = NOT SIGNIFICANT
 $\zeta = +1.5287646E+00$ SIGNIFICANCE OF ζ = NOT SIGNIFICANT
 $N = 747$ DEGREES OF FREEDOM = 745
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = IN/IN
 PARAMETER = STRAIN AT MAX STRESS

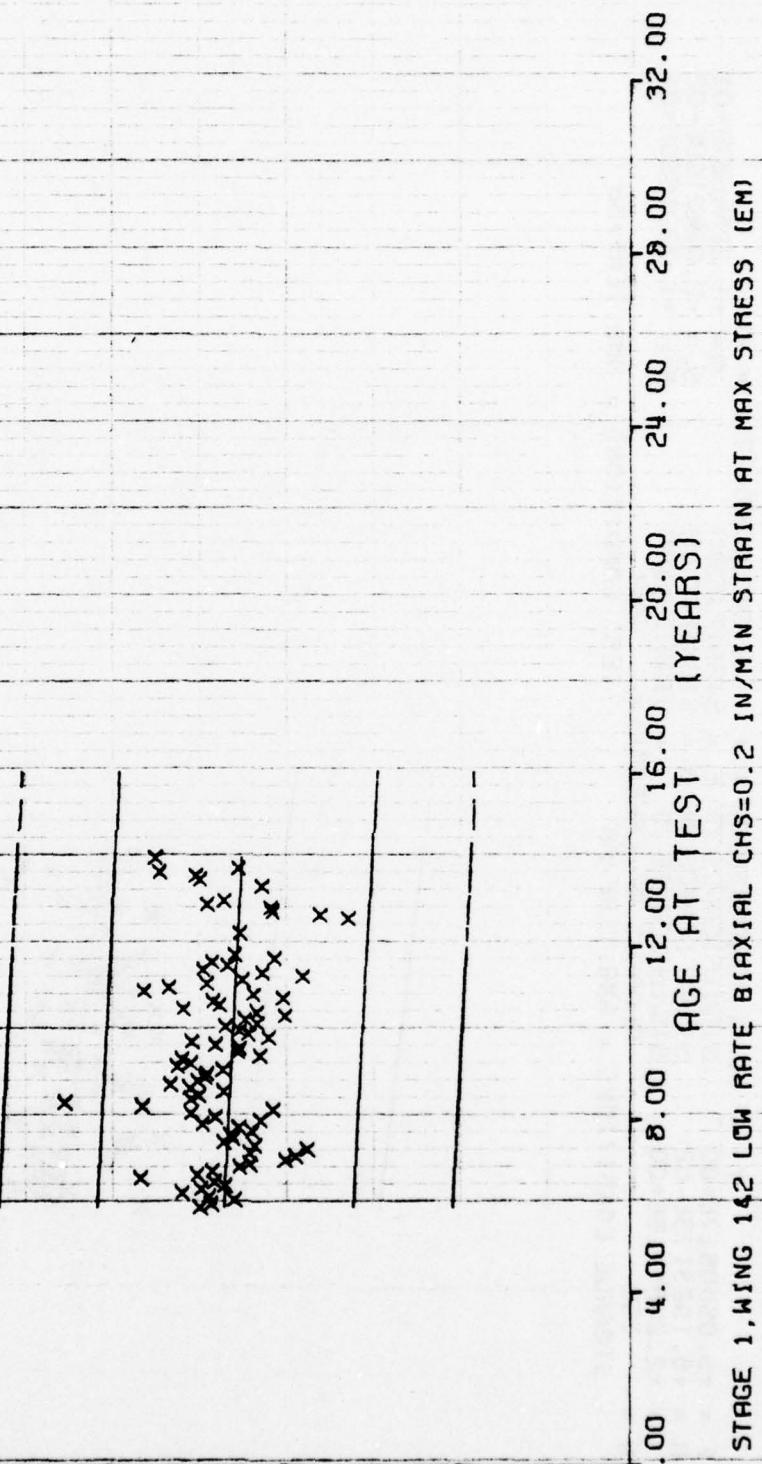


Figure 11

$\gamma = (1.1799033E+00 + (4.6405976E-02) \times \sigma_1) \times \sigma_1$
 $F = \text{SIGNIFICANCE OF } \gamma$
 $R = \text{SIGNIFICANCE OF } \sigma_1$
 $t = \text{SIGNIFICANCE OF } \sigma_1$
 $t = \text{DEGREES OF FREEDOM} = 748$
 $N = 750$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

UNIT OF MEASURE = PSI
 0.00 4.00 8.00 12.00 16.00 20.00 24.00 28.00 32.00

PARAMETER = MAXIMUM STRESS

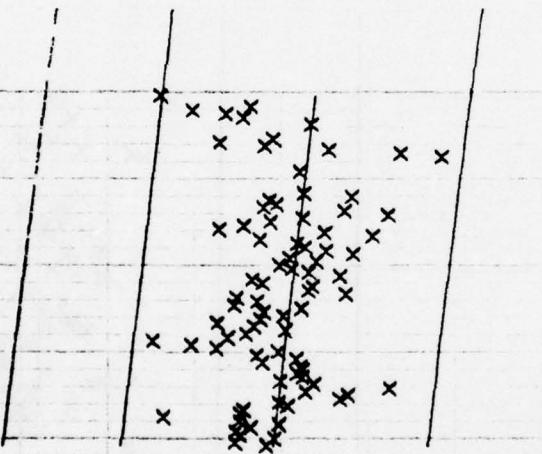
0.00 4.00 8.00 12.00 16.00 20.00 24.00 28.00 32.00

STAGE 1, WING 142 LOW RATE BIAXIAL CHS=0.2 IN/MIN MAXIMUM STRESS (SM)

Figure 12

$\gamma = ((+2.4814092E-01) + (-1.0012801E-04) \times x)$
 $F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANT}$
 $R^2 = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANT}$
 $N = 747$
 $\text{DEGREES OF FREEDOM} = 745$
 $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$
 $F = +8.0453136E+00$
 $R = -1.0336200E-01$
 $R^2 = +2.8364262E+00$
 $t = +2.747$
 $N = \text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\text{PARAMETER} = \text{STRAIN AT RUPTURE}$
 $\text{UNIT OF MEASURE} = \text{IN/IN}$
 $0.00 \quad 0.20 \quad 0.24 \quad 0.28 \quad 0.32 \quad 0.36$



$\text{STAGE 1, WING 142 LOW RATE BIAXIAL } \text{CHS}=0.2 \text{ IN/MIN STRAIN AT RUPTURE (ER)}$
 $0.00 \quad 4.00 \quad 8.00 \quad 12.00 \quad 16.00 \quad 20.00 \quad 24.00 \quad 28.00 \quad 32.00$

Figure 13

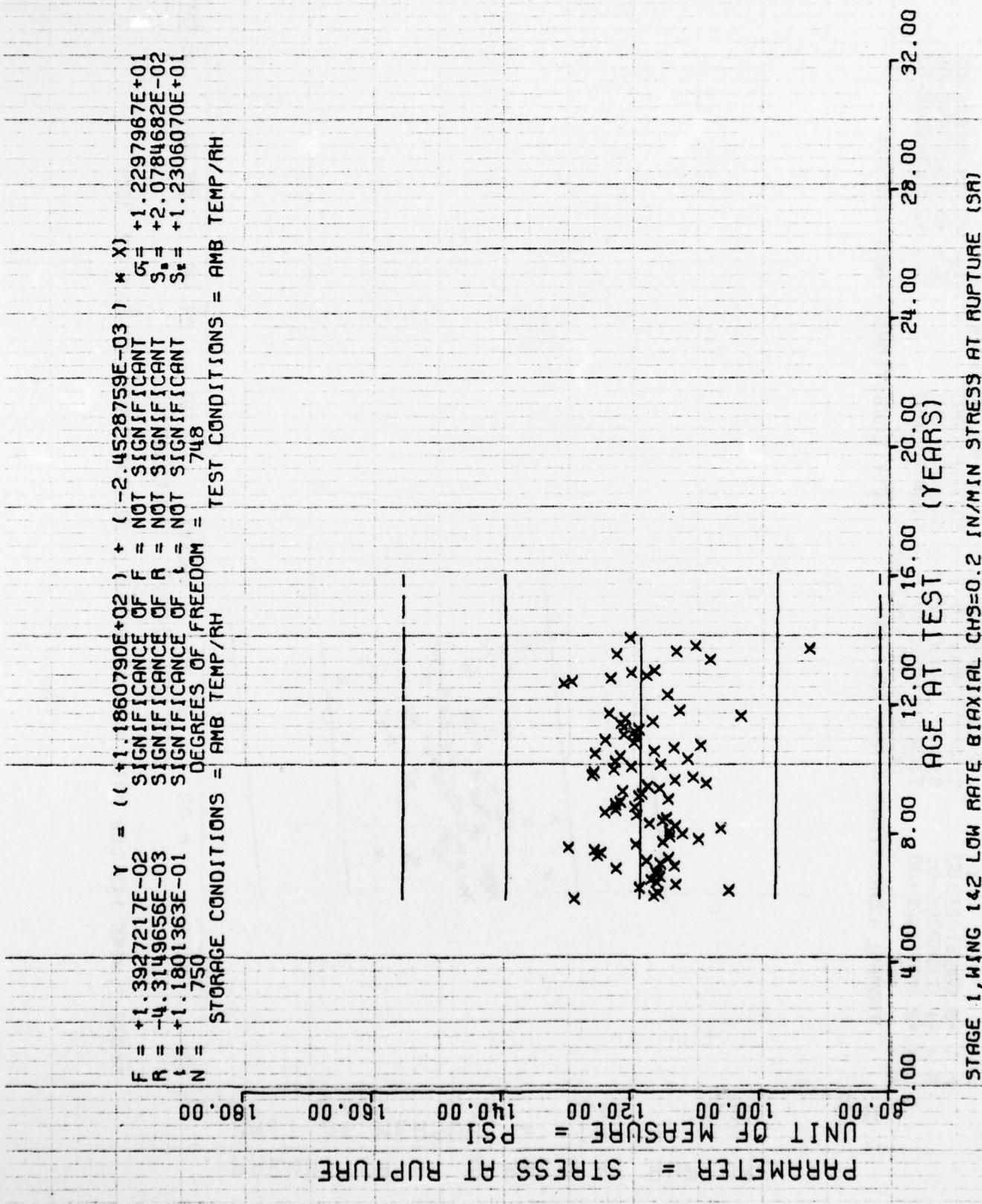


Figure 14

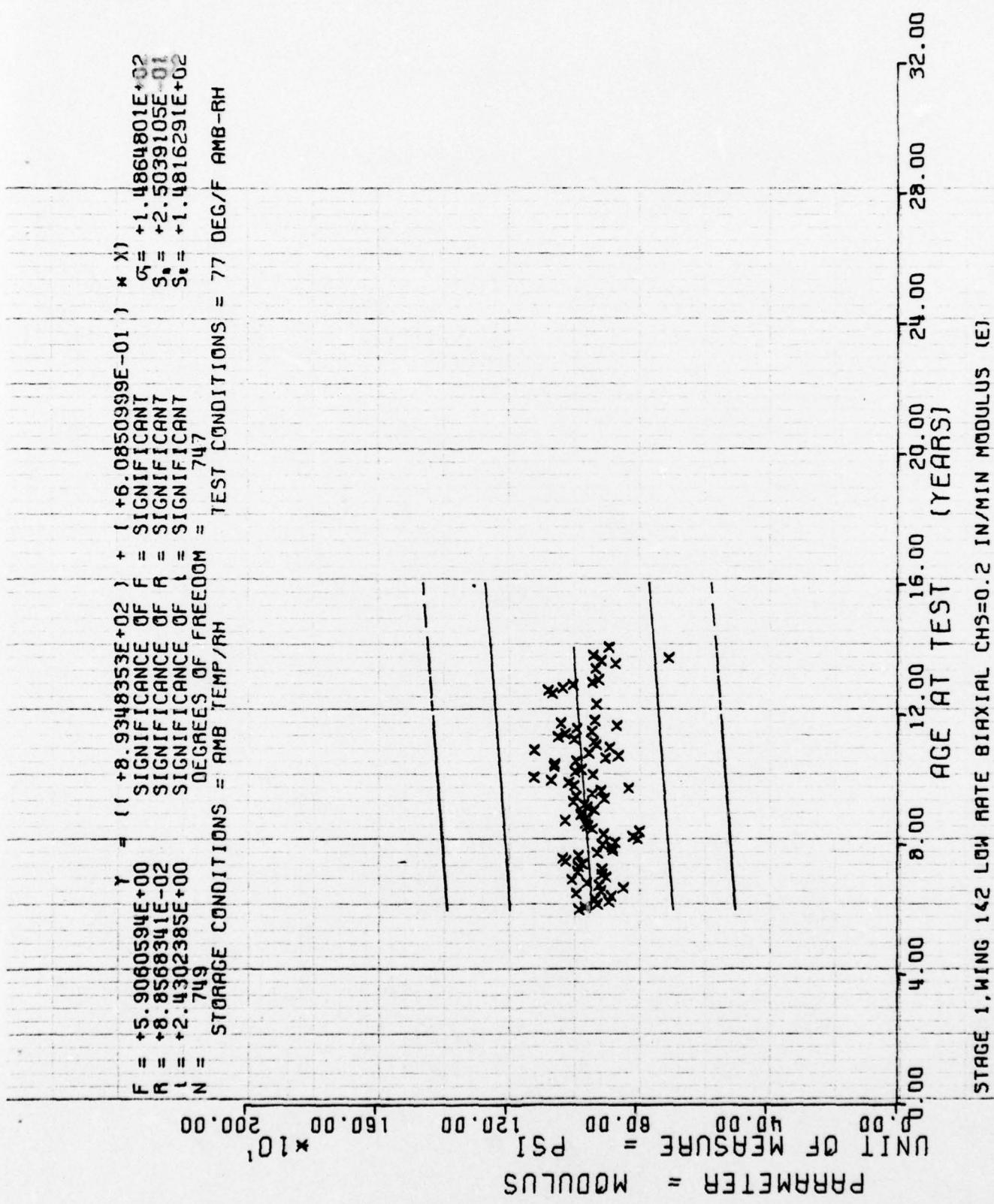


Figure 15

*** SAMPLE SIZE SUMMARY ***					
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
72.0	21	97.0	26	122.0	36
73.0	24	98.0	24	123.0	30
74.0	12	99.0	18	124.0	9
75.0	18	100.0	15	125.0	30
76.0	9	101.0	18	126.0	21
77.0	45	102.0	18	127.0	38
78.0	18	103.0	20	128.0	12
79.0	36	104.0	42	129.0	18
80.0	27	105.0	27	130.0	15
81.0	21	106.0	31	131.0	18
82.0	63	107.0	18	132.0	36
83.0	24	108.0	21	133.0	29
84.0	39	109.0	23	134.0	20
85.0	22	110.0	30	135.0	45
86.0	20	111.0	52	136.0	37
87.0	69	112.0	20	137.0	33
88.0	101	113.0	48	138.0	28
89.0	75	114.0	26	139.0	18
90.0	85	115.0	18	140.0	27
91.0	66	116.0	35	141.0	11
92.0	96	117.0	25	142.0	12
93.0	49	118.0	16	143.0	37
94.0	61	119.0	43	144.0	23
95.0	42	120.0	42	145.0	6
96.0	30	121.0	21	148.0	6

STAGE 1, WING 162 HIGH RATE

CHS=1750 IN/MIN STRAIN AT MAX STRESS (EMI)

This sample size summary is applicable to figures 16 thru 20.

$\gamma = ((+2.1700743E-01) + (+1.3884895E-04) \times X)$
 $F = \text{SIGNIFICANCE OF }$
 $R = \text{SIGNIFICANCE OF }$
 $t = \text{SIGNIFICANCE OF }$
 $N = \text{DEGREES OF FREEDOM} = 2480$
 $\text{STORAGE CONDITIONS = AMB TEMP/RH}$

PARAMETER = STRAIN AT MAX STRESS

UNIT OF MEASURE = IN/JIN

0.00 0.14 0.22 0.30 0.38 0.46

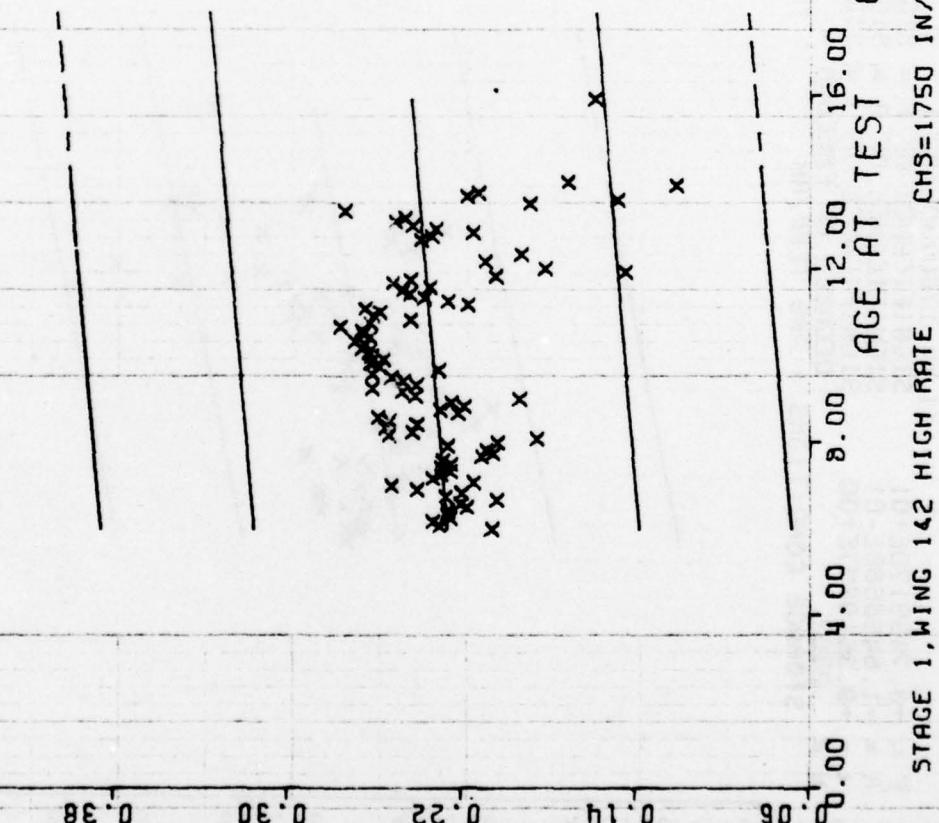


Figure 16

$F = +8.7459170E+01$ $\gamma = ((+4.1760730E+02) + (-3.4910254E-01) \times x)$
 $R = -1.8456566E-01$ $F = \text{SIGNIFICANT}$
 $t = +9.3519607E+00$ $R = \text{SIGNIFICANT}$
 $N = 2482$ $t = \text{SIGNIFICANT}$
 $\text{DEGREES OF FREEDOM} = 2480$ $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

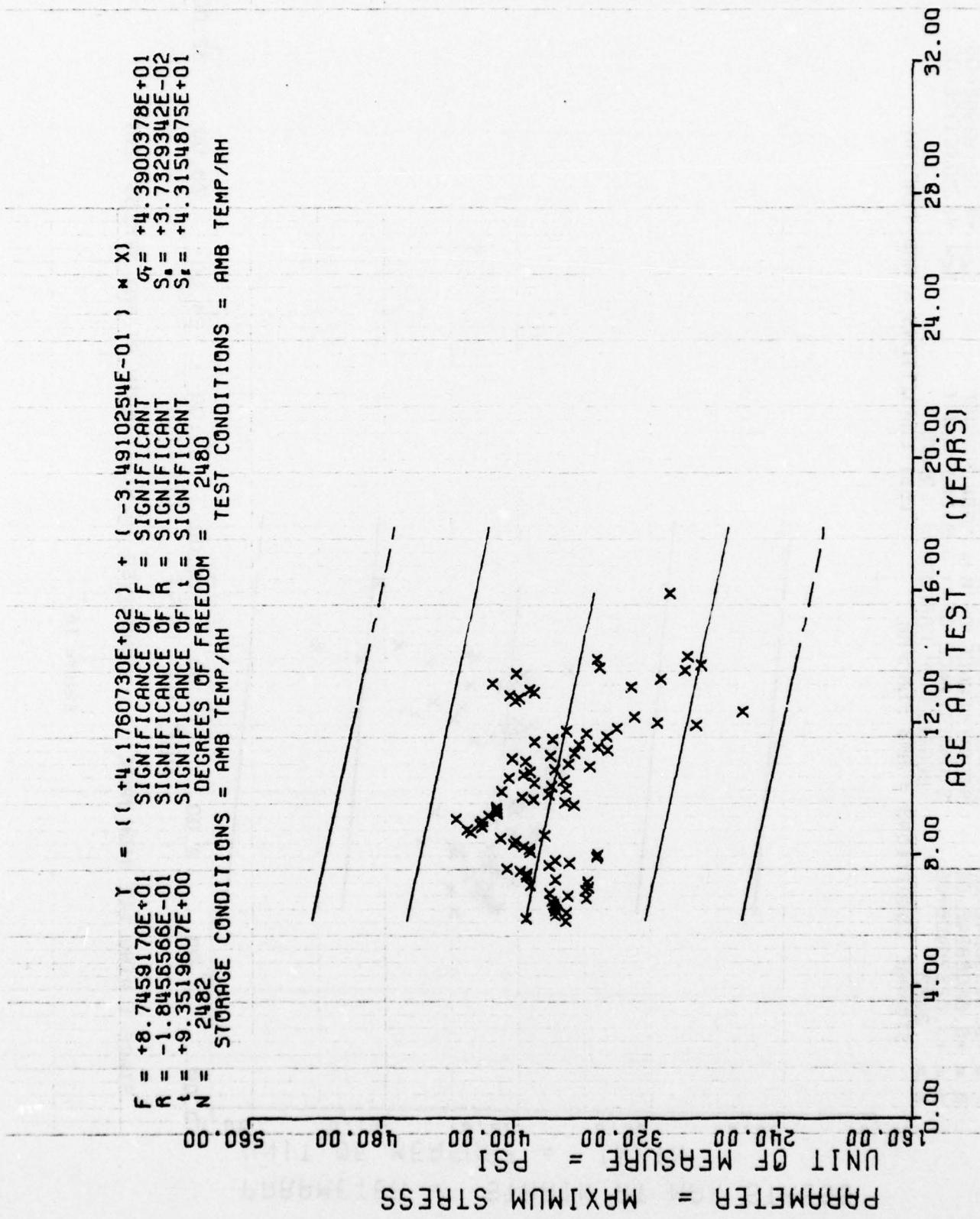


Figure 17

$Y = ((+4.1483149E-01) + (-7.4121932E-04) * X) * S$
 $F =$ SIGNIFICANT
 $R =$ SIGNIFICANT
 $S =$ SIGNIFICANT
 $S_t =$ SIGNIFICANT
 $F =$ SIGNIFICANT
 $R =$ SIGNIFICANT
 $S =$ SIGNIFICANT
 $S_t =$ SIGNIFICANT
 $D = 2180$
 $N = 2482$
 $STORAGE CONDITIONS = AMB TEMP/RH$
 $DEGREES OF FREEDOM = 2180$
 $TEST CONDITIONS = AMB TEMP/RH$

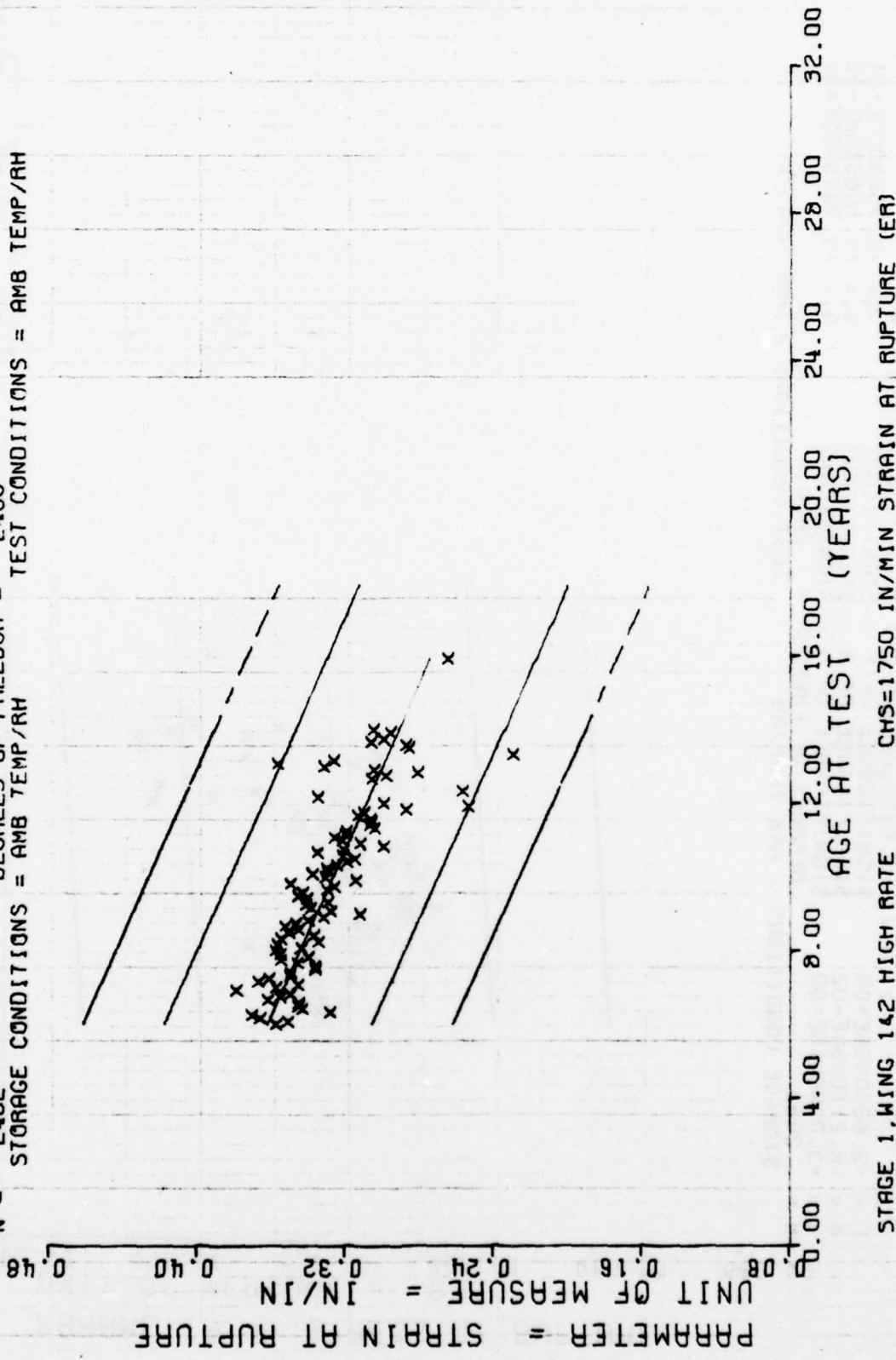


Figure 18

$F = +9.6040556E+00$
 $R = -6.2110094E-02$
 $S = +3.0990410E+00$
 $N = 2482$
 SIGNIFICANCE OF $F =$ SIGNIFICANT
 SIGNIFICANCE OF $R =$ SIGNIFICANT
 SIGNIFICANCE OF $S =$ SIGNIFICANT
 DEGREES OF FREEDOM = 2480
 TEST CONDITIONS = AMB TEMP/RH
 STORAGE CONDITIONS = AMB TEMP/RH

PARAMETER = STRESS AT RUPTURE

UNIT OF MEASURE = PSI

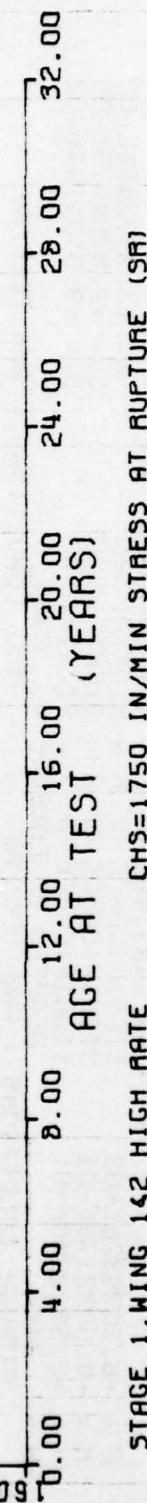
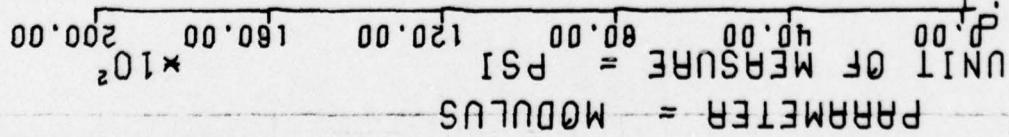


Figure 19

$\gamma = ((+5.0843350E+03) + (+9.2869871E+00) \times x)$
 $F = +6.0260038E+01$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = +1.5401946E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $t = +7.7627339E+00$ SIGNIFICANCE OF t = SIGNIFICANT
 $N = 2482$ DEGREES OF FREEDOM = 2480
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH



STAGE 1, WING 142 HIGH RATE CHS=1750 IN/MIN MODULUS (E)

Figure 20

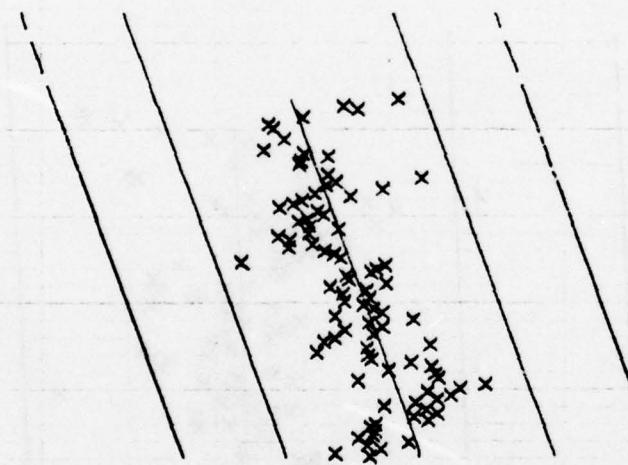
*** SAMPLE SIZE SUMMARY ***					
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
72.0	9	97.0	8	122.0	15
73.0	4	98.0	18	123.0	6
74.0	7	99.0	10	124.0	9
75.0	17	100.0	10	125.0	16
76.0	5	101.0	8	126.0	3
77.0	9	102.0	10	127.0	7
78.0	10	103.0	5	128.0	10
79.0	5	104.0	7	129.0	8
80.0	11	105.0	7	130.0	7
81.0	4	106.0	12	131.0	3
82.0	13	107.0	13	132.0	8
83.0	7	108.0	14	133.0	12
84.0	4	109.0	8	134.0	12
85.0	15	110.0	12	135.0	2
86.0	5	111.0	7	136.0	8
87.0	14	112.0	14	137.0	6
88.0	21	113.0	7	138.0	3
89.0	25	114.0	5	139.0	5
90.0	28	115.0	6	140.0	3
91.0	13	116.0	9	141.0	3
92.0	5	117.0	8	142.0	2
93.0	6	118.0	9	143.0	5
94.0	8	119.0	11	144.0	10
95.0	9	120.0	11	145.0	1
96.0	8	121.0	9	146.0	3

STAGE 1, WING 162 H.R. TRIAXIAL CHS=1750, PSI=600. STRAIN AT MAX STRESS (EMI)

This sample size summary is applicable to figures 21 thru 25.

$\gamma = ((+1.8022176E-01) + (+6.0284207E-04) \times X)$
 $F = +1.1576875E+02$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = +3.7063077E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $t = +1.0759589E+01$ SIGNIFICANCE OF t = SIGNIFICANT
 $N = 729$ DEGREES OF FREEDOM = 727
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

PARAMETER = STRAIN AT MAX STRESS
 UNIT OF MEASURE = IN/IN
 0.00 0.16 0.24 0.32 0.40 0.48



STAGE 1, WING 142 H.R. TRIAXIAL CHS=1750, PSI=600, STRAIN AT MAX STRESS (EM)

AGE AT TEST (YEARS)	PARAMETER = STRAIN AT MAX STRESS (EM)
4.00	0.00
4.00	0.04
4.00	0.08
4.00	0.12
4.00	0.16
4.00	0.20
4.00	0.24
4.00	0.28
4.00	0.32
4.00	0.36
4.00	0.40
4.00	0.44
8.00	0.00
8.00	0.04
8.00	0.08
8.00	0.12
8.00	0.16
8.00	0.20
8.00	0.24
8.00	0.28
8.00	0.32
8.00	0.36
12.00	0.00
12.00	0.04
12.00	0.08
12.00	0.12
12.00	0.16
12.00	0.20
12.00	0.24
12.00	0.28
12.00	0.32
16.00	0.00
16.00	0.04
16.00	0.08
16.00	0.12
16.00	0.16
16.00	0.20
16.00	0.24
16.00	0.28
16.00	0.32
20.00	0.00
20.00	0.04
20.00	0.08
20.00	0.12
20.00	0.16
20.00	0.20
20.00	0.24
20.00	0.28
20.00	0.32
24.00	0.00
24.00	0.04
24.00	0.08
24.00	0.12
24.00	0.16
24.00	0.20
24.00	0.24
24.00	0.28
24.00	0.32
28.00	0.00
28.00	0.04
28.00	0.08
28.00	0.12
28.00	0.16
28.00	0.20
28.00	0.24
28.00	0.28
28.00	0.32
32.00	0.00
32.00	0.04
32.00	0.08
32.00	0.12
32.00	0.16
32.00	0.20
32.00	0.24
32.00	0.28
32.00	0.32

Figure 21

$F = +1.1793352E+00$ $R = -4.0243843E-02$ $N = +1.0859720E+00$
 $F = +6.5495523E+02$ $R = +1.0859720E+00$ $N = 729$
 $\gamma = ((+6.5495523E+02) + (-9.9543622E-02) * X)$
 $F = \text{SIGNIFICANT}$ $R = \text{NOT SIGNIFICANT}$ $N = +5.9635232E+01$
 $F = \text{SIGNIFICANT}$ $R = \text{NOT SIGNIFICANT}$ $N = +9.1663156E-02$
 $F = \text{SIGNIFICANT}$ $R = \text{NOT SIGNIFICANT}$ $N = +5.9627888E+01$
 $\text{DEGREES OF FREEDOM} = 727$ $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$

UNIT OF MEASURE = PSI
 PARAMETER = MAXIMUM STRESS

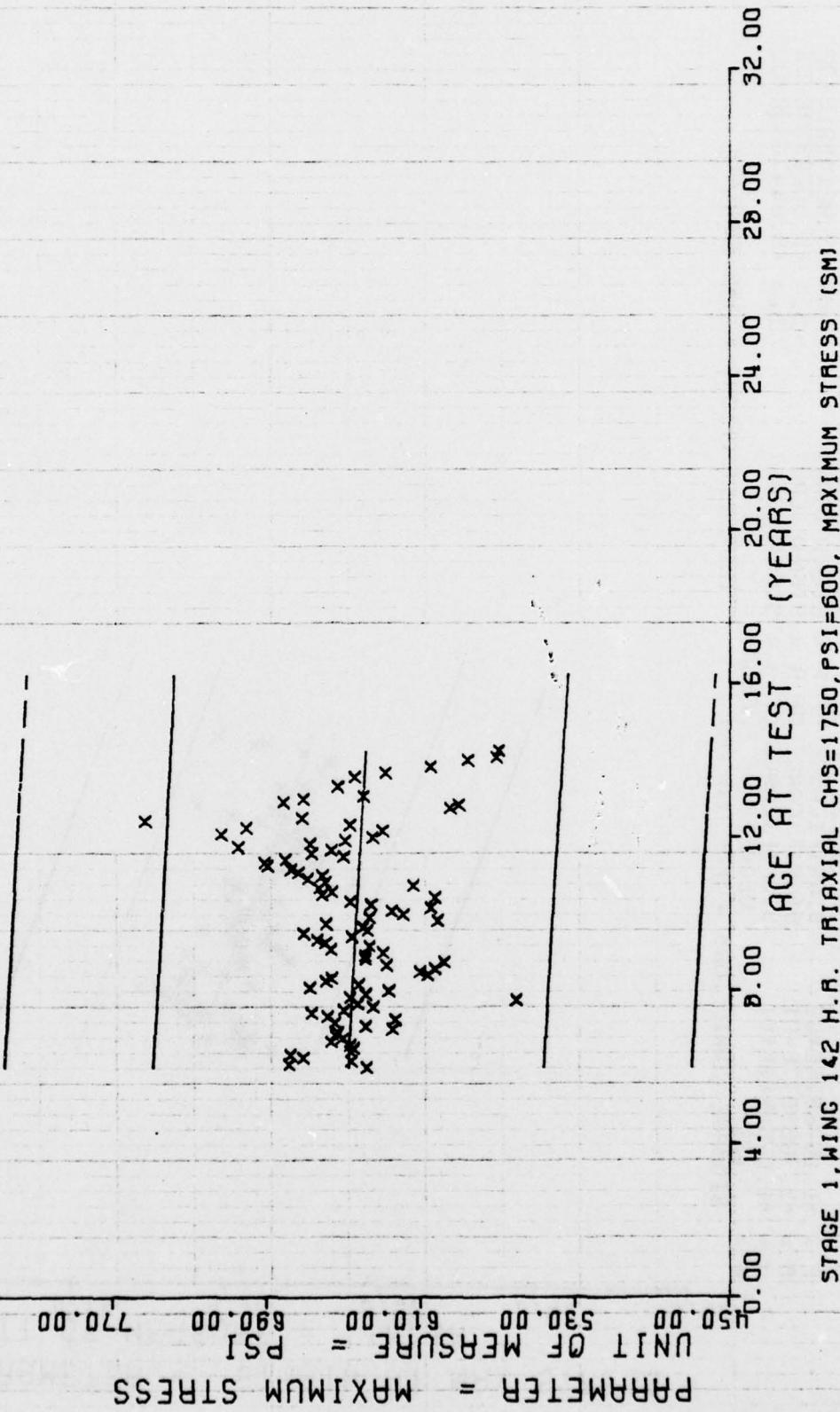


Figure 22

$\gamma = ((+2.6599416E-01) + (+1.6344824E-04) * x) \quad \sigma_t = +3.7371768E-02$
 $F = +8.1741248E+00 \quad F = \text{SIGNIFICANT}$
 $R = +1.0544488E-01 \quad R = \text{SIGNIFICANT}$
 $t = +2.8590426E+00 \quad t = \text{SIGNIFICANT}$
 $N = 729 \quad \text{DEGREES OF FREEDOM} = 727$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\text{UNIT OF MEASURE} = \text{IN/IN}$
 $0.00 \quad 0.21 \quad 0.26 \quad 0.31 \quad 0.36 \quad 0.41$
 $\text{PARAMETER} = \text{STRAIN AT RUPTURE}$

$0.00 \quad 4.00 \quad 8.00 \quad 12.00 \quad 16.00 \quad 20.00 \quad 24.00 \quad 28.00 \quad 32.00$

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STAGE 1, WING 142 H.R. TRIAXIAL CHS=1750, PSI=600, STRAIN AT RUPTURE (ER)

Figure 23

$\gamma = ((+6.2963671E+02) + (-1.0723823E-02) \times x)$
 $F = +1.4122207E-02$ SIGNIFICANCE OF $F = \text{NOT SIGNIFICANT}$ $\sigma_t = +5.8662248E+01$
 $R = -4.4073737E-03$ SIGNIFICANCE OF $R = \text{NOT SIGNIFICANT}$ $S_a = +9.0239847E-02$
 $t = +1.1883689E-01$ SIGNIFICANCE OF $t = \text{NOT SIGNIFICANT}$ $S_t = +5.8702010E+01$
 $N = 729$ DEGREES OF FREEDOM = 727 TEST CONDITIONS = 77 DEG/F AMB-RH

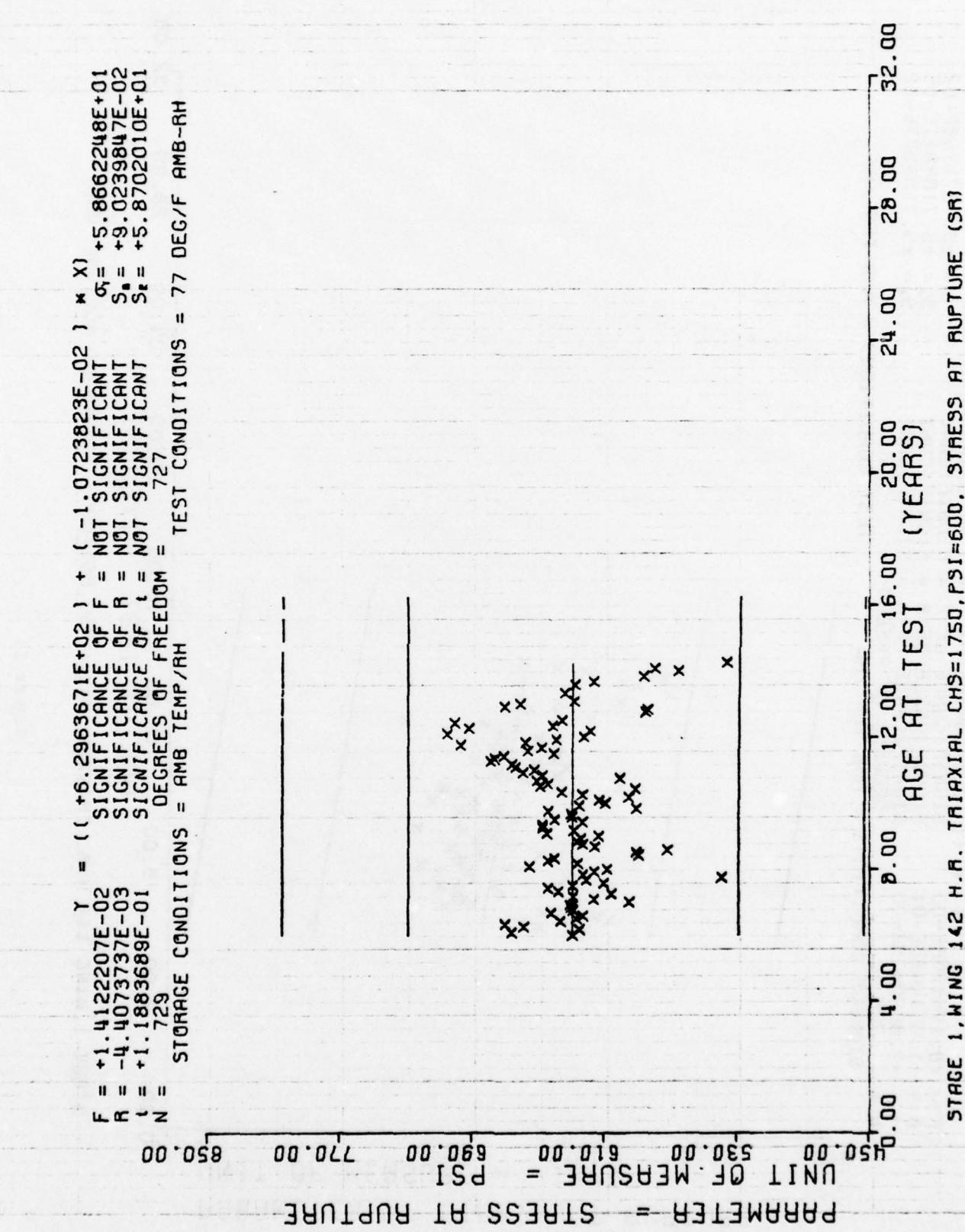


Figure 24

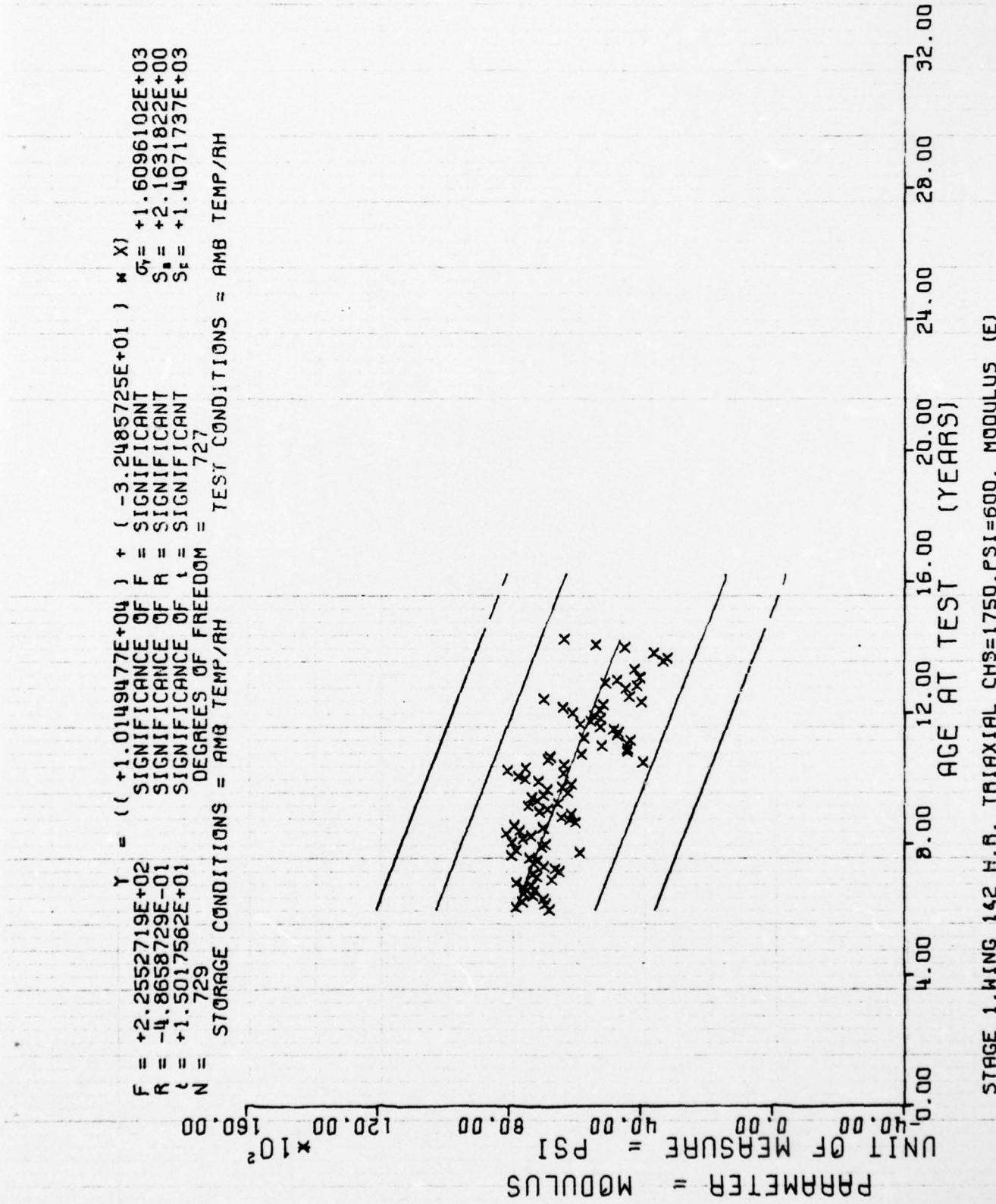


Figure 25

*** SAMPLE SIZE SUMMARY ***

AGE MONTHS	NR SAMPLES											
7.0	4	32.0	39	64.0	9	69.0	66	114	33	139	27	
8.0	3	33.0	38	65.0	12	50.0	72	115	36	140	18	
9.0	5	34.0	30	66.0	12	51.0	113	116	20	141	33	
10.0	4	35.0	29	67.0	24	92.0	63	117	35	142	15	
11.0	5	36.0	28	68.0	6	53.0	93	118	21	144	3	
12.0	7	37.0	23	69.0	17	54.0	60	119	36	145	3	
13.0	6	38.0	20	70.0	27	55.0	39	120	42	147	6	
14.0	10	39.0	15	71.0	24	56.0	30	121	15	152	3	
15.0	12	40.0	28	72.0	24	57.0	23	122	9	153	24	
16.0	9	41.0	38	73.0	21	58.0	30	123	12	154	15	
17.0	6	42.0	84	74.0	21	99.0	15	124	27	155	15	
18.0	8	43.0	63	75.0	21	100.0	48	125	21	156	12	
19.0	7	44.0	50	76.0	33	101.0	36	126	44	157	6	
20.0	11	45.0	39	77.0	15	102.0	27	127	51	158	3	
-	12	46.0	48	78.0	27	103.0	43	128	21	159	54	
22.0	9	47.0	36	79.0	27	104.0	34	129	33	160	9	
23.0	8	48.0	15	80.0	32	105.0	27	130	18	162	6	
-	24.0	8	49.0	17	81.0	18	106.0	33	131	27	163	3
25.0	18	50.0	5	82.0	36	107.0	24	132	28	164	3	
26.0	13	51.0	4	83.0	27	108.0	33	133	18	165	6	
27.0	17	52.0	3	84.0	27	109.0	27	134	33	166	6	
28.0	17	53.0	5	85.0	33	110.0	21	135	36	168	3	
29.0	29	54.0	3	86.0	33	111.0	18	136	39			
30.0	29	62.0	3	87.0	69	112.0	36	137	33			
31.0	50	63.0	3	88.0	69	113.0	21	138	15			

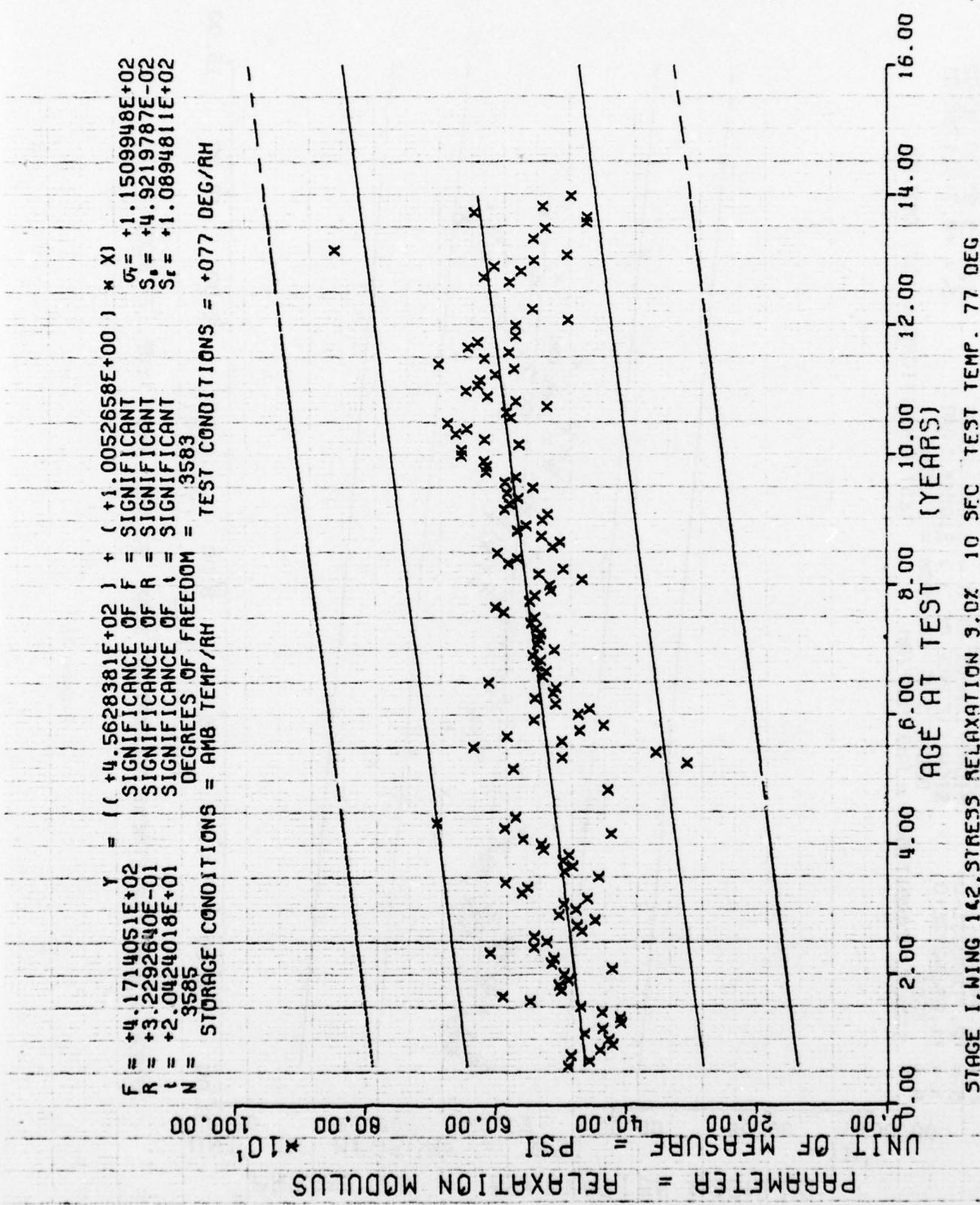
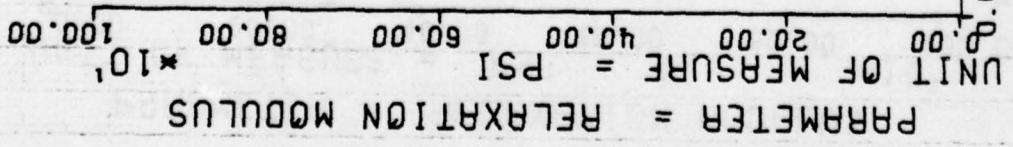


Figure 26

$\gamma = ((+3.7599201E+02) + (+6.6554611E-01) * X) \times 10^4$
 $F = +2.8645832E+02$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = +2.7198023E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $t = +1.6925079E+01$ SIGNIFICANCE OF t = SIGNIFICANT
 $N = 3588$ DEGREES OF FREEDOM = 3586
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = +077 DEG/RH



STAGE 1, WING 142, STRESS RELAXATION 3.0% 50 SEC TEST TEMP. 77 DEG

Figure 27

$\gamma = \{ (+3.5092883E+02) + (+6.1695486E-01) \times x \}$
 SIGNIFICANCE OF $F =$ SIGNIFICANT
 SIGNIFICANCE OF $R =$ SIGNIFICANT
 SIGNIFICANCE OF $t =$ SIGNIFICANT
 DEGREES OF FREEDOM = 3586

STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = +077 DEG/RH

0.00 20.00 40.00 60.00 80.00 100.00

PARAMETER = RELAXATION MODULUS

UNIT OF MEASURE = PSI $\times 10^4$

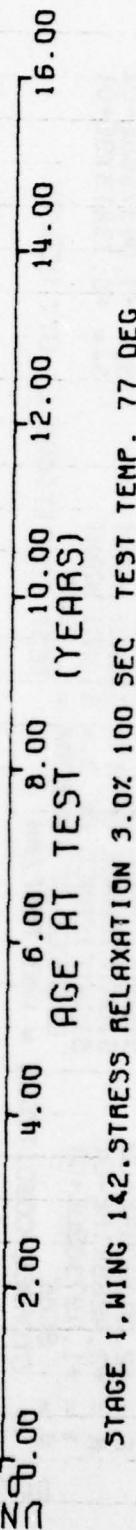


Figure 28

$F = +2.2419344E+02$ $\gamma = ((+2.9072528E+02) + (+4.5574474E-01) * X)$
 $R = +2.4263416E-01$ $F = \text{SIGNIFICANT}$
 $\alpha = +1.4973090E+01$ $R = \text{SIGNIFICANT}$
 $N = 3584$ $\alpha = \text{SIGNIFICANT}$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$ $\text{DEGREES OF FREEDOM} = 3584$ $\text{TEST CONDITIONS} = +077 \text{ DEG/RH}$

PARAMETER = RELAXATION MODULUS

UNIT OF MEASURE = PSI *10⁴

STAGE 1, WING 142, STRESS RELAXATION 3.0X 1000 SEC TEST TEMP. 77 DEG

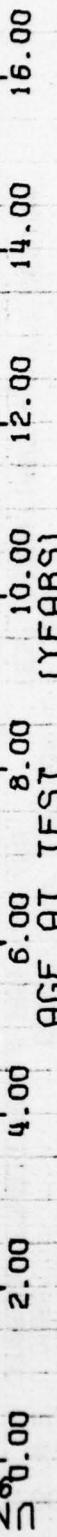


Figure 29

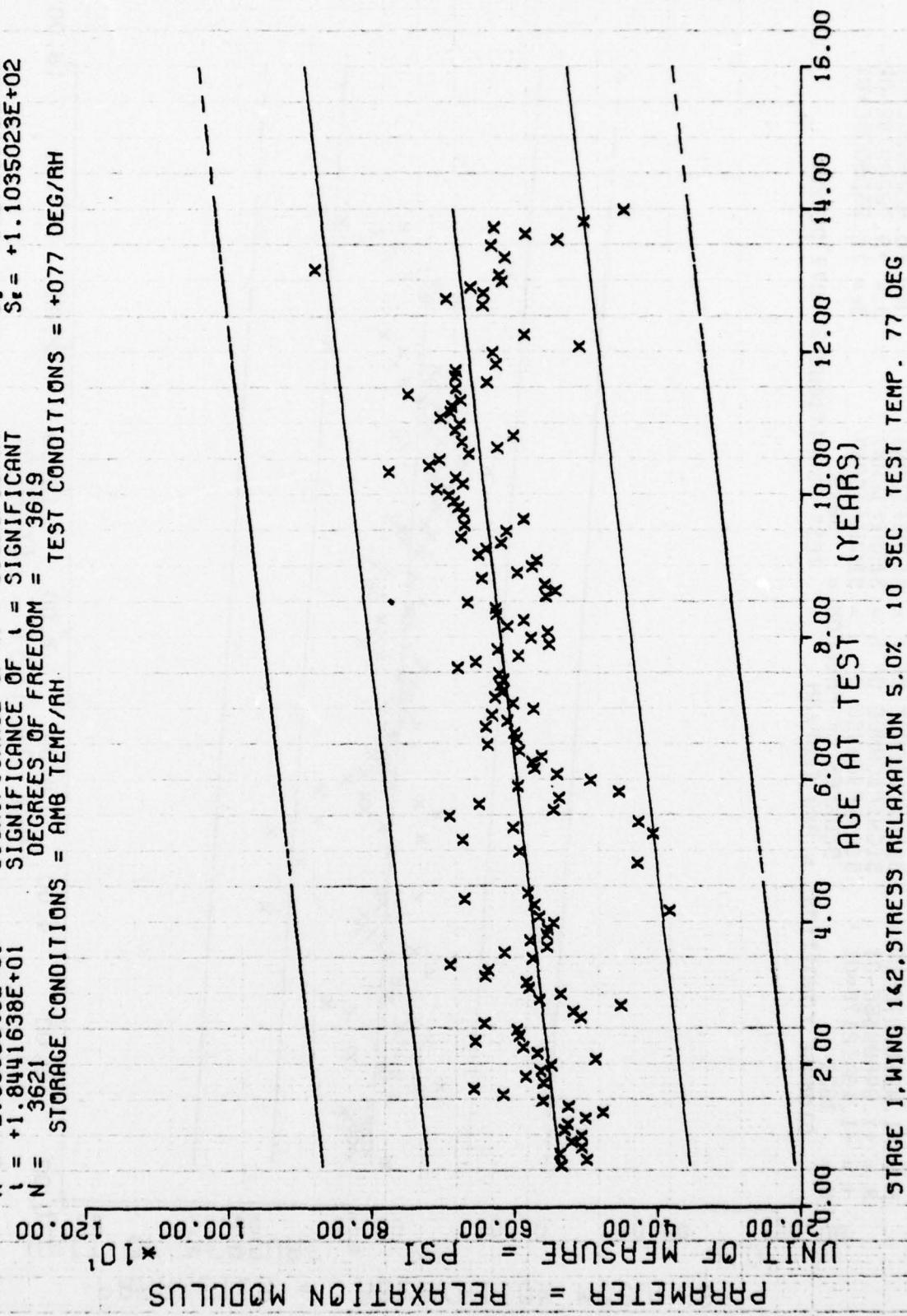
*** SAMPLE SIZE SUMMARY ***

AGE MONTHS	NR SAMPLES										
7.0	4	32.0	40	63.0	3	88.0	66	113	18	138	15
8.0	3	33.0	38	64.0	9	89.0	69	114	33	139	27
9.0	5	34.0	30	65.0	12	90.0	81	115	30	140	18
10.0	4	35.0	30	66.0	12	91.0	99	116	27	141	32
11.0	5	36.0	27	67.0	24	92.0	69	117	39	142	15
12.0	7	37.0	26	68.0	6	93.0	84	118	21	144	3
13.0	6	38.0	20	69.0	18	94.0	60	119	36	145	3
14.0	10	39.0	23	70.0	27	95.0	36	120	42	147	6
15.0	12	40.0	27	71.0	27	96.0	30	121	15	152	3
16.0	9	41.0	38	72.0	26	97.0	27	122	9	153	24
17.0	6	42.0	92	73.0	21	98.0	30	123	12	154	15
18.0	8	43.0	61	74.0	21	99.0	18	124	24	155	15
19.0	7	44.0	44	75.0	27	100.0	39	125	27	156	12
20.0	11	45.0	40	76.0	38	101.0	39	126	33	158	3
21.0	12	46.0	48	77.0	15	102.0	30	127	44	160	9
22.0	9	47.0	36	78.0	27	103.0	36	128	40	162	6
23.0	8	48.0	15	79.0	27	104.0	27	129	21	163	3
24.0	8	49.0	17	80.0	30	105.0	38	130	27	164	3
25.0	18	50.0	5	81.0	20	106.0	36	131	27	165	6
26.0	13	51.0	13	82.0	33	107.0	24	132	33	166	6
27.0	17	52.0	4	83.0	27	108.0	33	133	21	168	3
28.0	17	53.0	5	84.0	27	109.0	27	134	33		
29.0	29	58.0	3	85.0	33	110.0	24	135	33		
30.0	29	60.0	3	86.0	32	111.0	18	136	39		
31.0	51	62.0	3	87.0	69	112.0	36	137	27		

STAGE 1, WING 1-2, STRESS RELAXATION 5.0% 1000 SEC TEST TEMP. 77 DEG

This sample size summary is applicable to figures 30 thru 33.

$\gamma = ((+5.3304225E+02) + (+9.1673488E-01) * X)$
 $F = +3.4009404E+02$ SIGNIFICANCE OF $F = \text{SIGNIFICANT}$
 $R = +2.9309040E-01$ SIGNIFICANCE OF $R = \text{SIGNIFICANT}$
 $t = +1.8441638E+01$ SIGNIFICANCE OF $t = \text{SIGNIFICANT}$
 $N = 3621$ DEGREES OF FREEDOM = 3619
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = +077 DEG/RH



$F = +3.4009404E+02$ $\gamma = ((+5.3304225E+02) + (+9.1673488E-01) * X)$
 $R = +2.9309040E-01$ $F = \text{SIGNIFICANT}$ $\sigma = +1.1540294E+02$
 $R = +1.8441638E+01$ $R = \text{SIGNIFICANT}$ $S_u = +4.9710055E-02$
 $N = 3621$ $\alpha = \text{SIGNIFICANT}$ $S_t = +1.1035023E+02$
 $\text{DEGREES OF FREEDOM} = 3619$ $\text{TEST CONDITIONS} = +077 \text{ DEG/RH}$

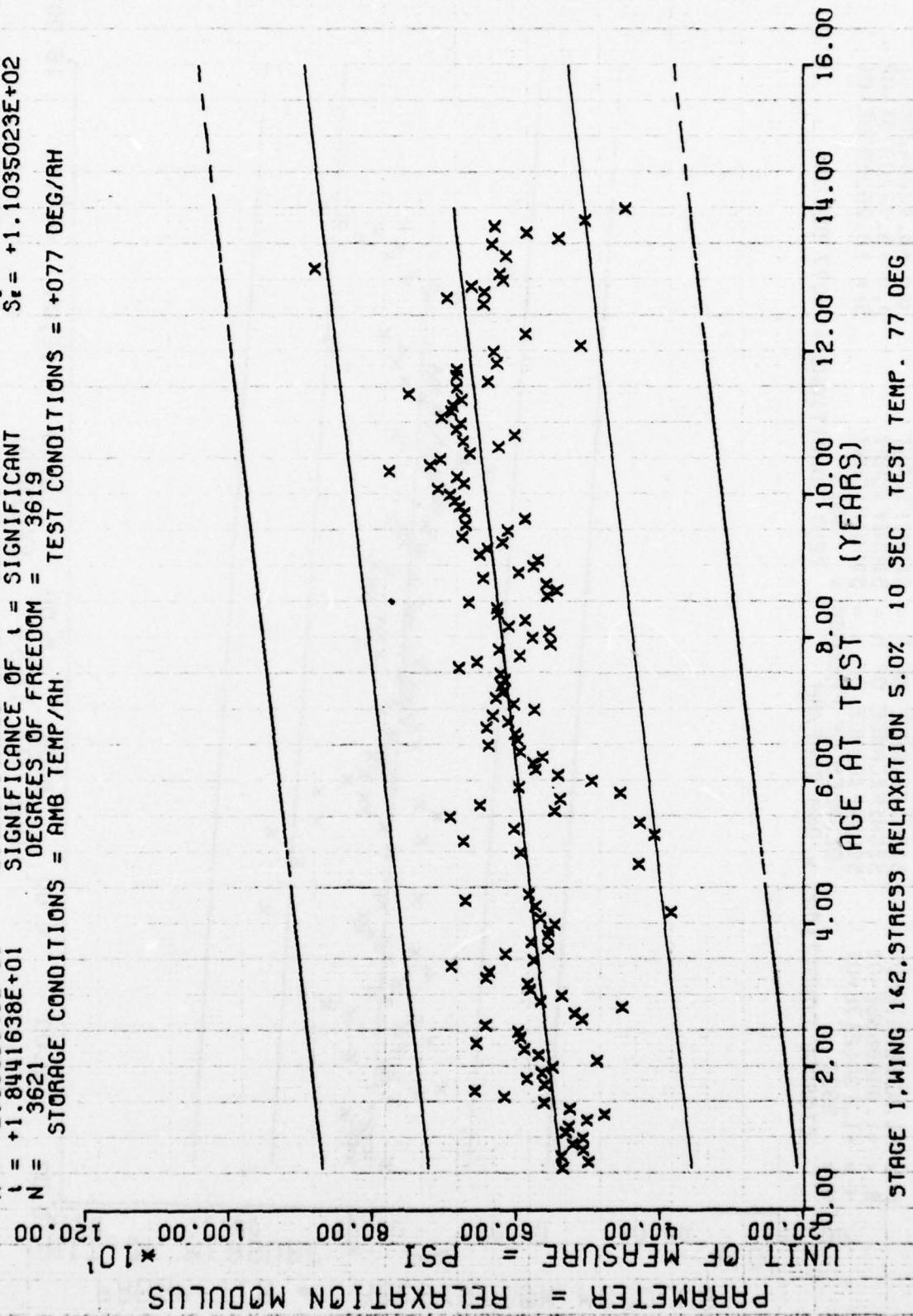


Figure 30

$\gamma = (1.4259267E+02 + 4.2647426E+02) \times 10^{-4}$
 $F = 1.4259267E+02$
 $R = 1.9469855E-01$
 $\epsilon = 1.1941217E+01$
 $N = 3621$
 STORAGE CONDITIONS = AMB TEMP/RH

TEST CONDITIONS = +077 DEG/RH

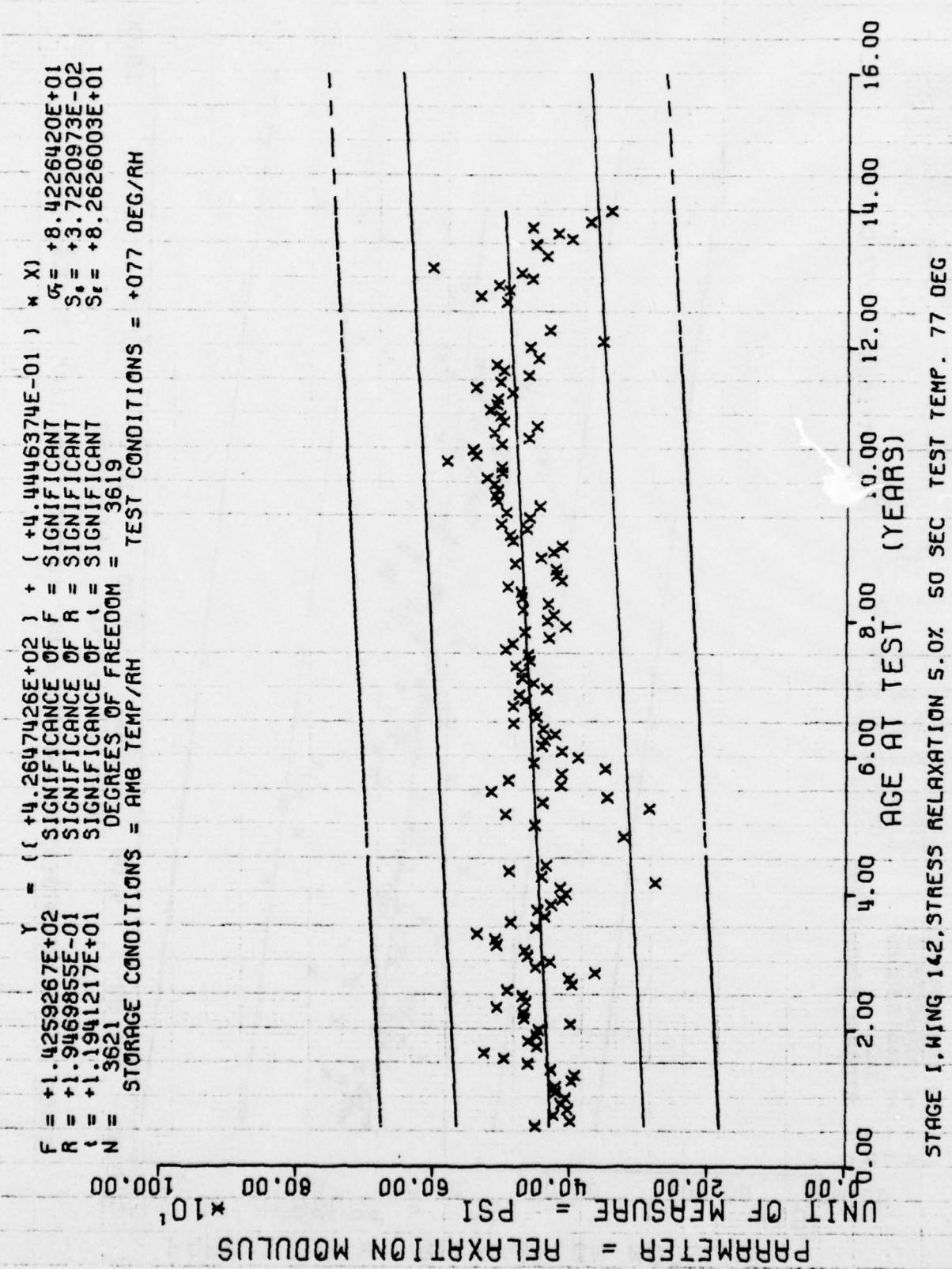


Figure 31

$\gamma = ((+3.9834522E+02) + (+3.9556301E-01) * X)$
 $F = 1.3053316E+02$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = 1.8658289E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $I = 1.1425111E+01$ SIGNIFICANCE OF I = SIGNIFICANT
 $N = 3621$ DEGREES OF FREEDOM = 3619
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = +077 DEG/RH

UNIT OF MEASURE = PSI
 PARAMETER = RELAXATION MODULUS
 0.00 2.00 4.00 6.00 8.00 10.00 12.00 14.00 16.00

STAGE 1, WING 142, STRESS RELAXATION 5.02 100 SEC TEST TEMP. 77 DEG

Figure 32

$\gamma = ((+3.2784520E+02) + (+2.6421476E-01) * X) * 10^3$
 $F = +8.4697263E+01$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = +1.5134532E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $R^2 = +9.2031116E+00$ SIGNIFICANCE OF R^2 = SIGNIFICANT
 $N = 3615$ DEGREES OF FREEDOM = 3613
 $STORAGE CONDITIONS = A M B T E M P / R H$ TEST CONDITIONS = +077 DEG/RH

UNIT OF MEASURE = PSI
 PARAMETER = RELAXATION MODULUS

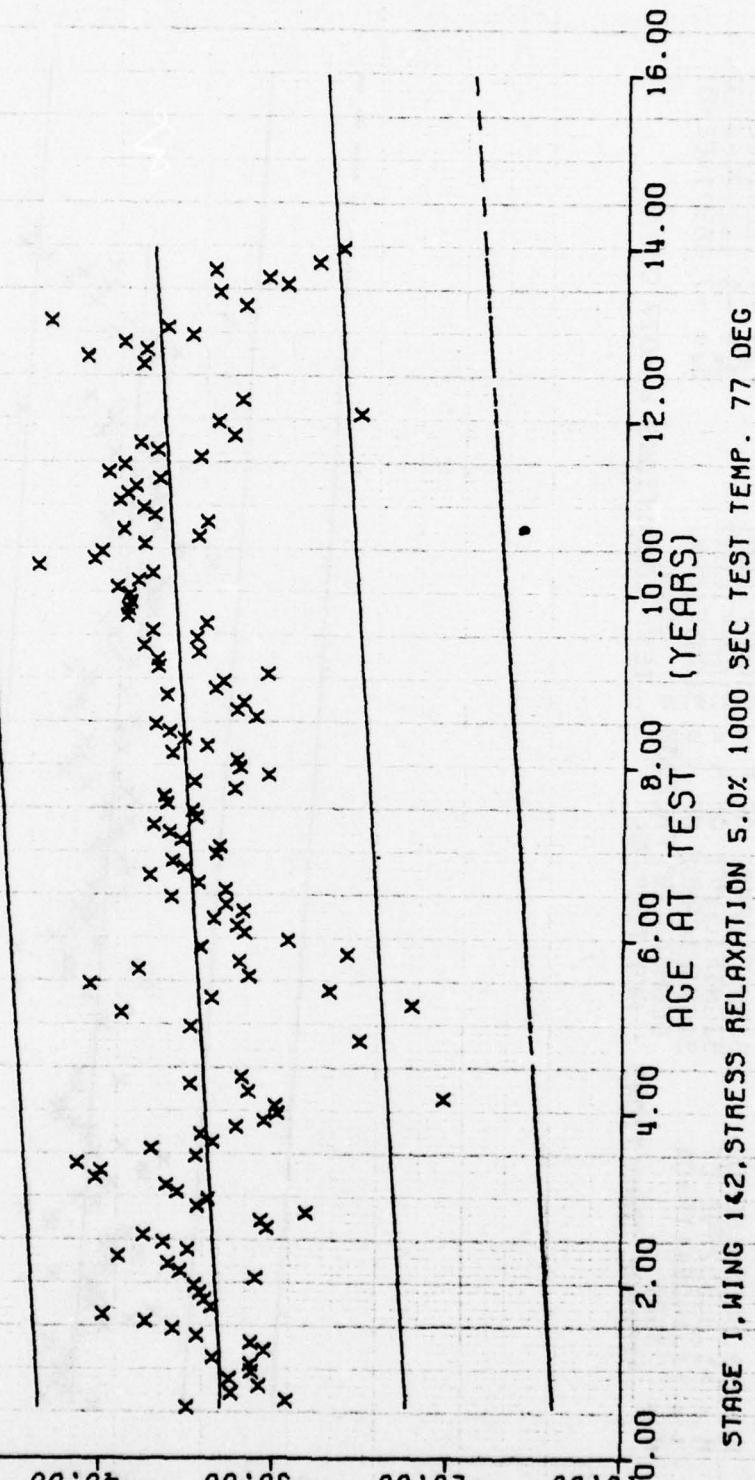


Figure 33

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES										
6.0	3	31.0	150	63.0	18	88.0	525	113	138	27	
7.0	24	32.0	63	64.0	24	89.0	585	114	153	27	
8.0	74	33.0	90	65.0	3	90.0	610	115	87	140	15
9.0	30	34.0	72	66.0	30	91.0	466	116	102	141	6
10.0	28	35.0	74	67.0	6	92.0	528	117	78	142	9
11.0	32	36.0	99	68.0	21	93.0	302	118	143	143	12
12.0	40	37.0	61	69.0	35	94.0	346	119	121	144	12
13.0	75	38.0	77	70.0	22	95.0	191	120	120	145	6
14.0	95	39.0	68	71.0	87	96.0	180	121	15	146	3
15.0	105	40.0	49	72.0	78	97.0	180	122	72	155	3
16.0	25	41.0	99	73.0	63	98.0	183	123	42	163	33
17.0	60	42.0	95	74.0	144	99.0	161	124	27	166	6
18.0	57	43.0	174	75.0	102	100.0	198	125	42		
19.0	72	44.0	223	76.0	105	101.0	159	126	33		
20.0	57	45.0	234	77.0	114	102.0	171	127	33		
21.0	34	46.0	212	78.0	120	103.0	171	128	21		
22.0	73	47.0	234	79.0	117	104.0	195	129	27		
23.0	54	48.0	176	80.0	150	105.0	195	130	30		
24.0	87	49.0	138	81.0	168	106.0	213	131	24		
25.0	141	50.0	87	82.0	207	107.0	117	132	15		
26.0	218	51.0	57	83.0	190	108.0	207	133	9		
27.0	264	52.0	25	84.0	308	109.0	126	134	12		
28.0	257	53.0	26	85.0	180	110.0	156	135	24		
29.0	243	57.0	3	86.0	195	111.0	195	136	15		
30.0	224	60.0	3	87.0	552	112.0	171	137	12		

$F = 7.6491085E+01$
 $R = 6.8787727E-02$
 $t = 8.7459182E+00$
 $N = 16091$
 STORAGE CONDITIONS = AMB TEMP/RH

$\gamma = ((+6.5879664E+01) + (+7.7591479E-03)) * X$
 SIGNIFICANCE OF R = SIGNIFICANT
 SIGNIFICANCE OF R = SIGNIFICANT
 SIGNIFICANCE OF R = SIGNIFICANT
 DEGREES OF FREEDOM = 16089
 TEST CONDITIONS = AMB TEMP/RH

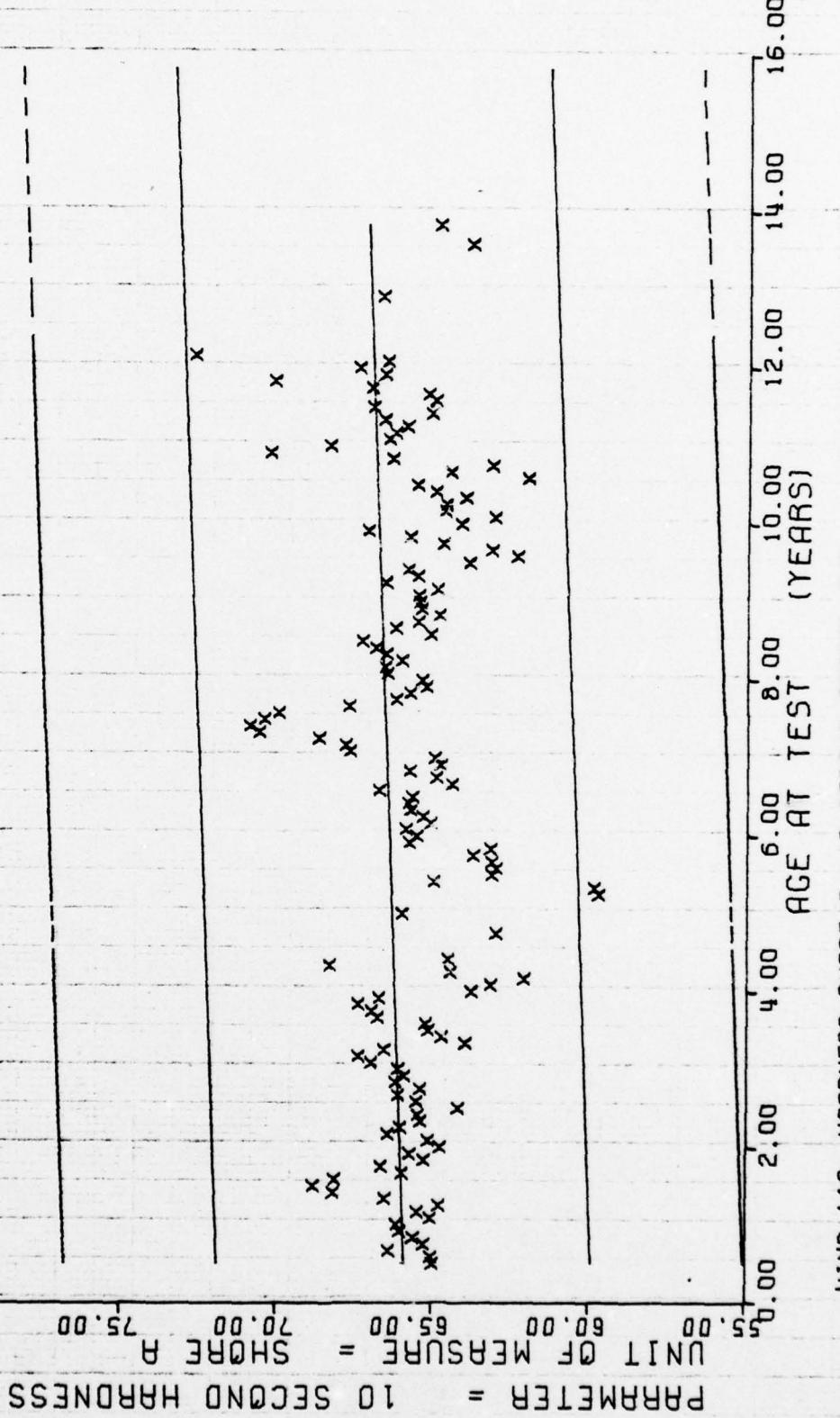


Figure 34

*** SAMPLE SIZE SUMMARY ***					
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
18.0	1	78.0	5	103.0	11
24.0	1	79.0	5	104.0	6
25.0	3	80.0	5	105.0	6
27.0	7	81.0	9	106.0	13
28.0	8	82.0	9	107.0	3
29.0	8	83.0	11	108.0	6
30.0	11	84.0	8	109.0	7
31.0	3	85.0	6	110.0	7
33.0	2	86.0	9	111.0	6
34.0	3	87.0	14	112.0	4
57.0	1	88.0	14	113.0	3
63.0	3	89.0	13	114.0	6
65.0	3	90.0	45	115.0	3
66.0	2	91.0	33	116.0	6
67.0	4	92.0	28	117.0	8
68.0	3	93.0	23	118.0	8
69.0	4	94.0	10	119.0	4
70.0	7	95.0	7	120.0	7
71.0	3	96.0	7	121.0	10
72.0	7	97.0	9	122.0	6
73.0	7	98.0	4	123.0	5
74.0	3	99.0	4	124.0	4
75.0	7	100.0	9	125.0	4
76.0	9	101.0	7	126.0	5
77.0	9	102.0	7	127.0	3

WING 162 SI TP-H1011 DYNAMIC RESPONSE, CENTER-WT 70 GM, STOR SHEAR AT 200 Hz

This sample size summary is applicable to figures 35 thru 38.

$\gamma = ((+3.2793182E+03) + (-4.6827322E+00) \times x)$
 $F = +5.6448158E+01$ SIGNIFICANCE OF $F =$ SIGNIFICANT
 $R = -2.8307425E-01$ SIGNIFICANCE OF $R =$ SIGNIFICANT
 $t = +7.5131989E+00$ SIGNIFICANCE OF $t =$ SIGNIFICANT
 $N = 650$ DEGREES OF FREEDOM = 648

TEST CONDITIONS = AMB TEMP/RH
STORAGE CONDITIONS = AMB TEMP/RH

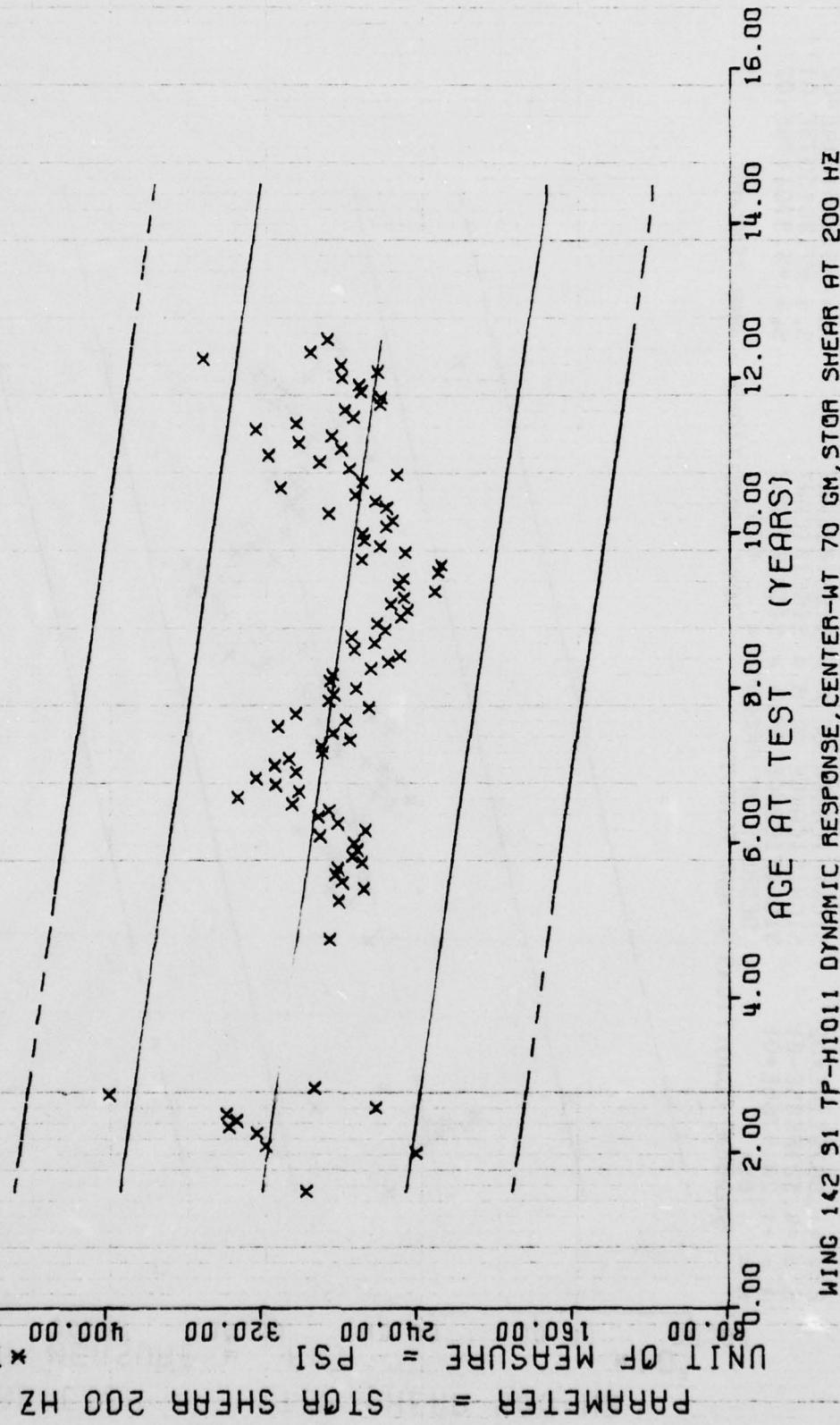
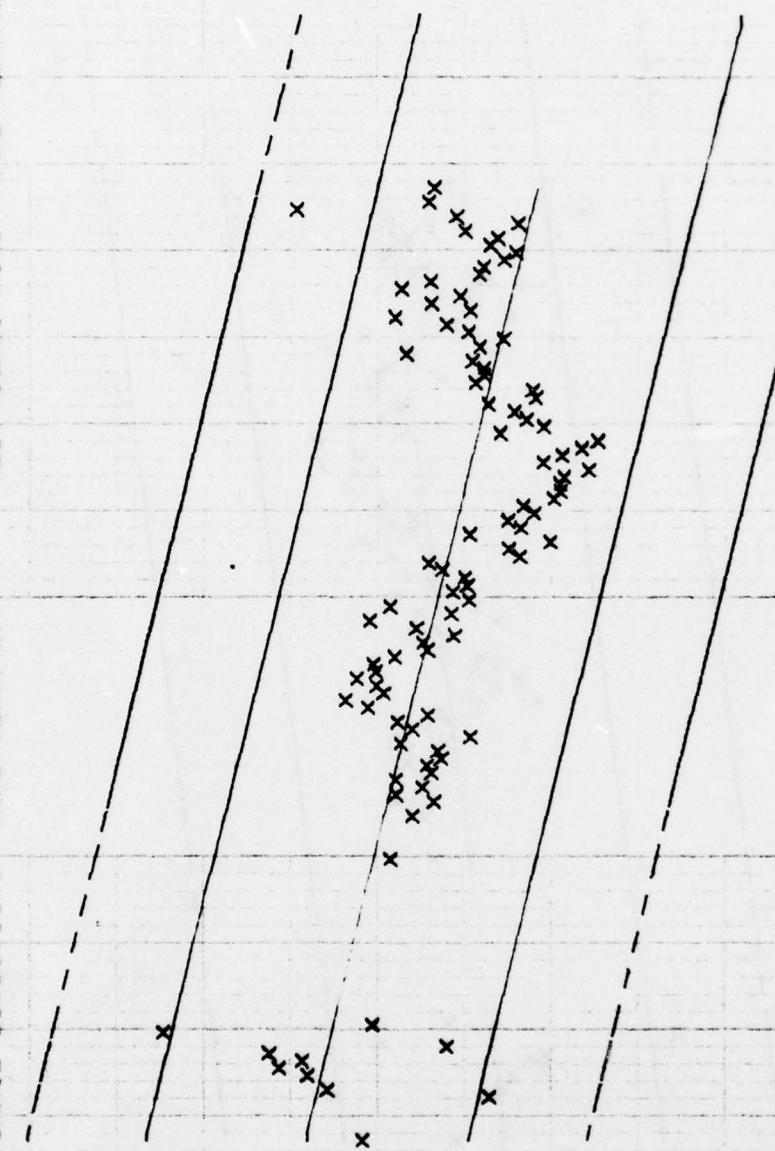


Figure 35

$F = +1.6679114E+02$
 $R = -4.5216505E-01$
 $t = +1.2914764E+01$
 $N = 651$
 $Y = ((+4.5614651E+03) + (-1.0152951E+01) * X)$
 $F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANT}$
 $\text{DEGREES OF FREEDOM} = 649$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$
 $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$

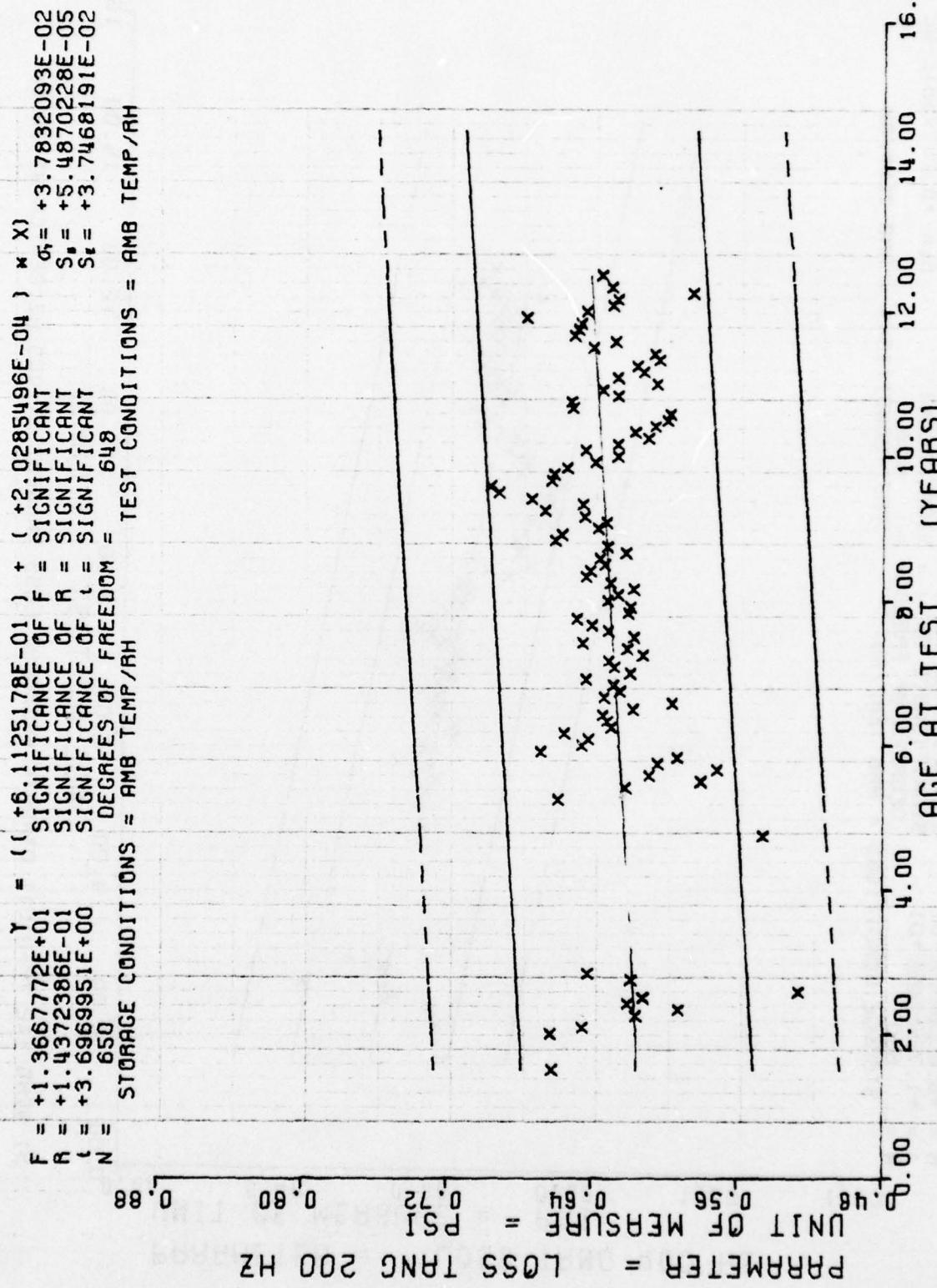
$\text{PARAMETER} = \text{STOR SHEAR 400 Hz}$
 $\text{UNIT OF MEASURE} = \text{PSI}$
 $* 10^2$



$\text{WING 142 S1 TP-H1011 DYNAMIC RESPONSE, CENTER-WT 70 GM, STOR SHEAR AT 400 Hz}$
 $0.00 \quad 2.00 \quad 4.00 \quad 6.00 \quad 8.00 \quad 10.00 \quad 12.00 \quad 14.00 \quad 16.00$
 $\text{AGE AT TEST (YEARS)}$

Figure 36

$\gamma = ((+6.1125178E-01) + (+2.0285496E-01) \times X)$
 $F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANT}$
 $l = \text{SIGNIFICANT}$
 $D = \text{DEGREES OF FREEDOM} = 648$
 $N = \text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$



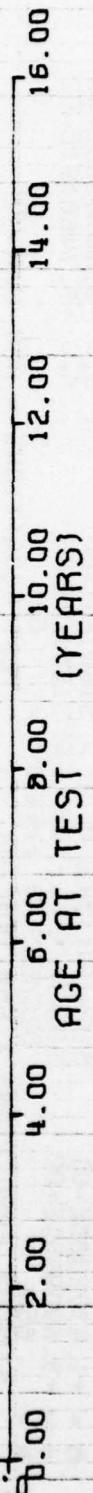
S1 WING 142 TP-H1011 DYNAMIC RESPONSE LOSS TANGENT AT 200 Hz, CENTER-WT 70 GM

Figure 37

$F = +2.9741839E+02$ $R = +5.6088249E-01$ $L = +1.7245822E+01$ $N = 650$
 $\gamma = ((+5.1764406E-01) + (+1.5898774E-03) * X)$
 $\sigma_f = +7.5950513E-02$
 $S_u = +9.2189133E-05$
 $S_l = +6.2927581E-02$
 $\text{DEGREES OF FREEDOM} = 648$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$ $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$

TEST CONDITIONS = AMB TEMP/RH

PARAMETER = LOSS TANG 400 Hz
UNIT OF MEASURE = PSI
0.40 0.60 0.80 1.00 1.20



SI WING 142 TP-H101 DYNAMIC RESPONSE LOSS TANGENT AT 400 Hz. CENTER-WT 70 gm

Figure 38

*** SAMPLE SIZE SUMMARY ***

AGE MONTHS	NR SAMPLES								
2.0	1	31.0	24	56.0	1	88.0	15	113	30
7.0	1	32.0	18	64.0	2	89.0	24	114	15
8.0	3	33.0	14	65.0	1	90.0	11	115	18
9.0	3	34.0	9	66.0	3	91.0	9	116	21
10.0	5	35.0	9	67.0	3	92.0	47	117	15
11.0	3	36.0	6	68.0	8	93.0	25	118	18
12.0	8	37.0	9	69.0	2	94.0	39	119	33
13.0	5	38.0	3	70.0	6	95.0	50	120	21
14.0	9	39.0	9	71.0	4	96.0	66	121	18
15.0	15	40.0	5	72.0	6	97.0	37	122	12
16.0	13	41.0	8	73.0	1	98.0	34	123	6
17.0	4	42.0	6	74.0	6	99.0	45	124	6
18.0	14	43.0	12	75.0	4	100.0	39	125	3
19.0	6	44.0	11	76.0	5	101.0	25		
20.0	10	45.0	27	77.0	5	102.0	19		
21.0	10	46.0	28	78.0	4	103.0	48		
22.0	12	47.0	39	79.0	4	104.0	12		
23.0	11	48.0	26	80.0	7	105.0	33		
24.0	8	49.0	21	81.0	6	106.0	21		
25.0	16	50.0	18	82.0	7	107.0	22		
26.0	24	51.0	14	83.0	10	108.0	12		
27.0	28	52.0	7	84.0	2	109.0	21		
28.0	21	53.0	2	85.0	12	110.0	15		
29.0	28	54.0	3	86.0	11	111.0	9		
30.0	27	55.0	3	87.0	5	112.0	18		

STAGE 1 WING 1-2 TP/H1011 CONSTANT STRAIN

This sample size summary is applicable to figure 39

$F = +4.4723813E-01$ $S_{\sigma} = +3.0299829E+00$
 $R = -1.6501510E-02$ $S_{\sigma} = +2.1568808E-03$
 $\lambda = +6.6875865E-01$ $S_{\sigma} = +3.0304927E+00$
 $N = 1644$ $STORAGE\ CONDITIONS = AMB\ TEMP/RH$
 $DEGREES\ OF\ FREEDOM = 1642$
 $TEST\ CONDITIONS = AMB\ TEMP/RH$

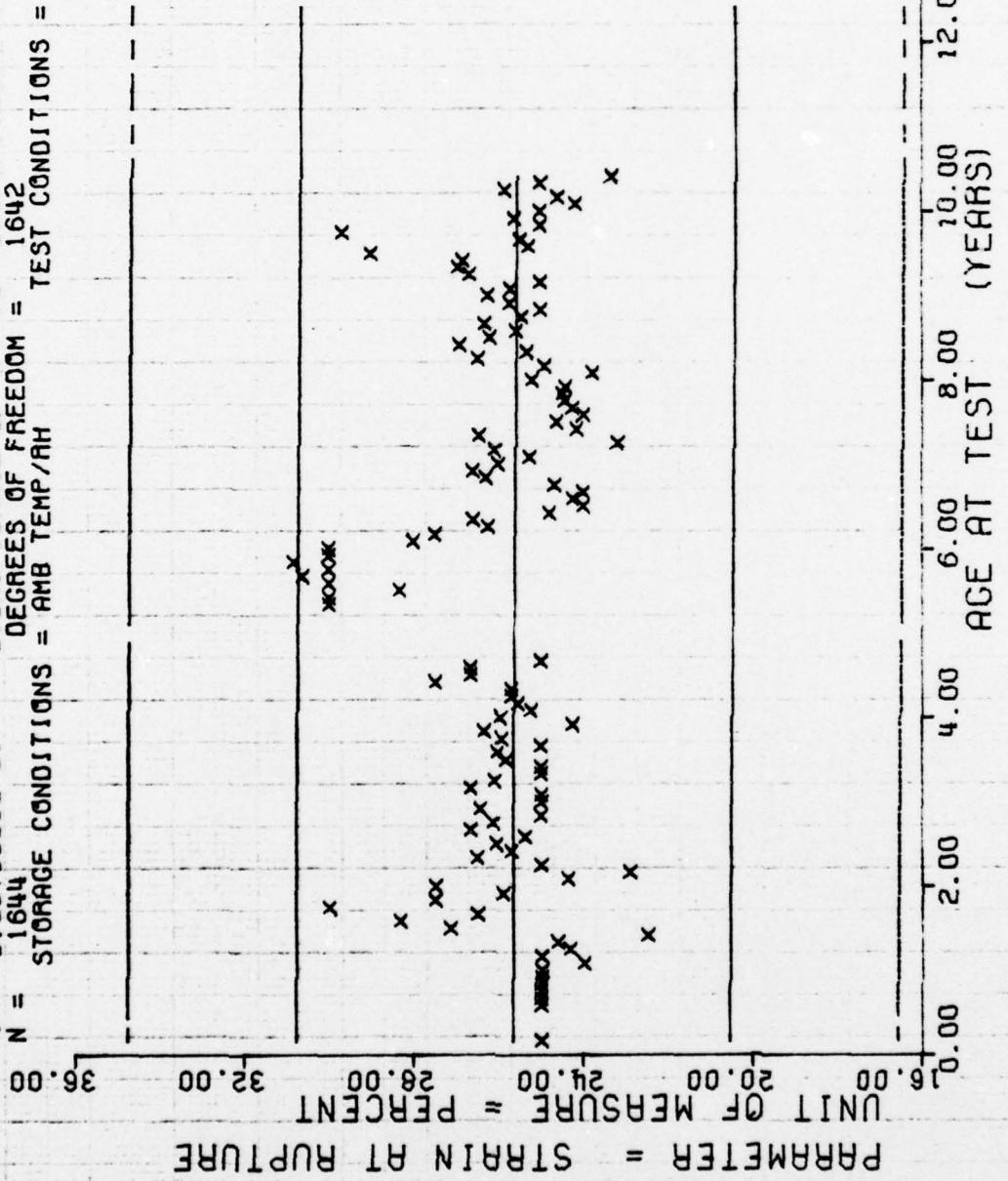


Figure 39

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
88.0	3	117.0	19	142.0	24
91.0	3	118.0	24	143.0	33
92.0	3	119.0	43	144.0	18
93.0	9	120.0	29	145.0	15
94.0	9	121.0	20	146.0	3
95.0	3	122.0	15	147.0	6
97.0	3	123.0	13	148.0	3
98.0	3	124.0	18	150.0	3
99.0	9	125.0	9	152.0	16
101.0	9	126.0	21	153.0	12
102.0	6	127.0	18	154.0	8
103.0	21	128.0	24	155.0	10
104.0	9	129.0	30	158.0	3
105.0	9	130.0	30	159.0	3
106.0	9	131.0	24	161.0	6
107.0	24	132.0	9	164.0	6
108.0	12	133.0	12	165.0	3
109.0	24	134.0	24	167.0	6
110.0	3	135.0	12	169.0	6
111.0	9	136.0	24		
112.0	3	137.0	21		
113.0	24	138.0	30		
114.0	9	139.0	41		
115.0	21	140.0	9		
116.0	18	141.0	24		

$F = 2.3933509E+01$
 $R = +1.5440121E-01$
 $L = +4.8921885E+00$
 $N = 982$
 STORAGE CONDITIONS = AMB TEMP/RH

$\gamma = ((+4.7128232E-05) + (+7.1241148E-08) * X)$
 SIGNIFICANCE OF F = SIGNIFICANT
 SIGNIFICANCE OF R = SIGNIFICANT
 SIGNIFICANCE OF L = SIGNIFICANT
 DEGREES OF FREEDOM = 980

TEST CONDITIONS = 5 DEGREES C/MIN

UNIT OF MEASURE = IN/IN/DEG C $\times 10^{-3}$
 PARAMETER = TCLE BELOW T (G)

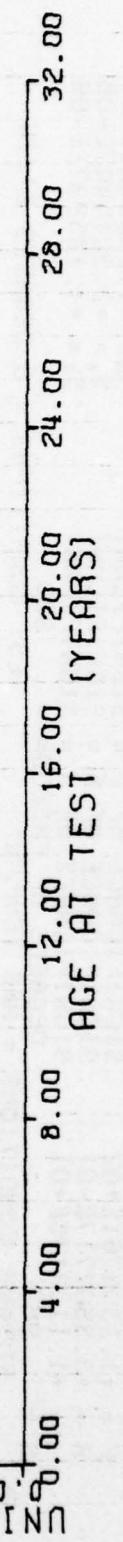
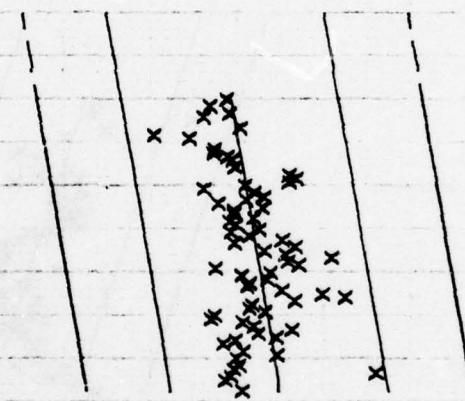
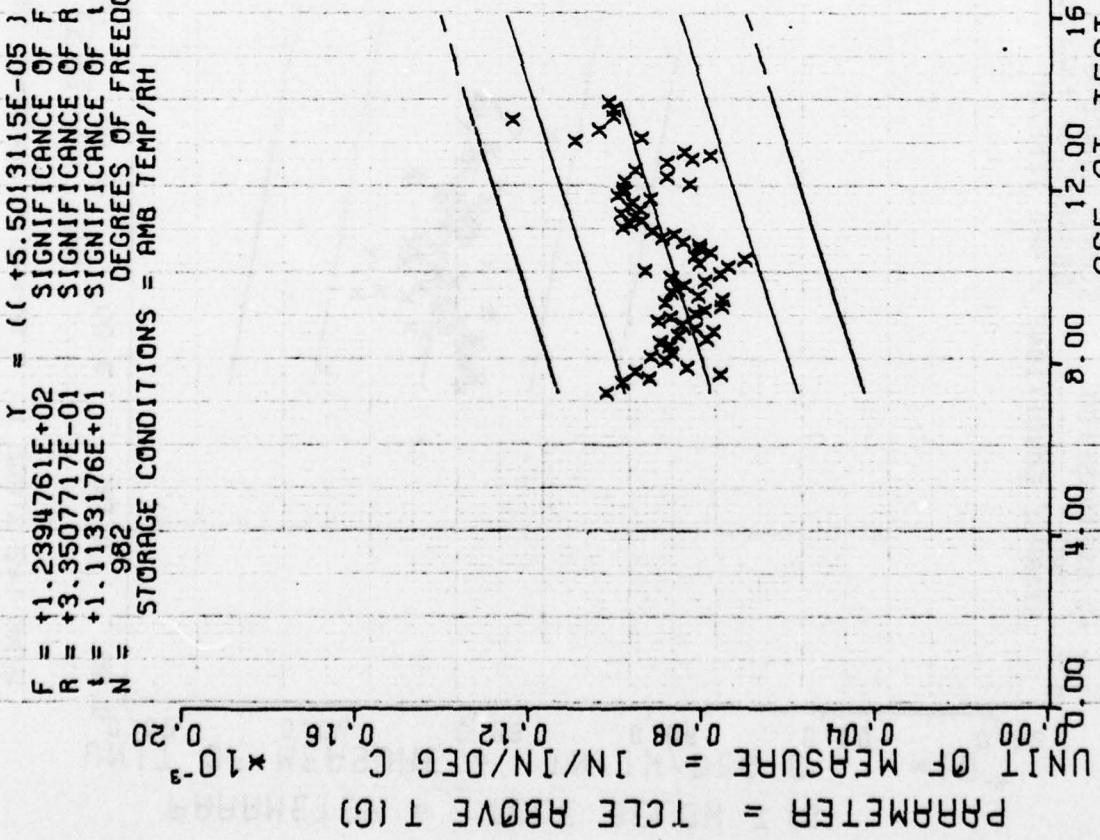


Figure 40

$\gamma = ((+5.5013115E-05) + (+2.5845052E-07) * x)$
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANCE OF } t = \text{SIGNIFICANT}$
 $N = \text{DEGREES OF FREEDOM} = 980$
 $N = \text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\sigma_f = +1.2560021E-05$
 $S_f = +2.3214447E-08$
 $S_t = +1.1839971E-05$
 $\text{TEST CONDITIONS} = 5 \text{ DEGREES C/MIN}$



WING 142 STAGE 1 TP-H1011. THERMAL COEFFICIENT OF LINEAR EXPANSION ABOVE TG

Figure 41

*** SAMPLE SIZE SUMMARY ***

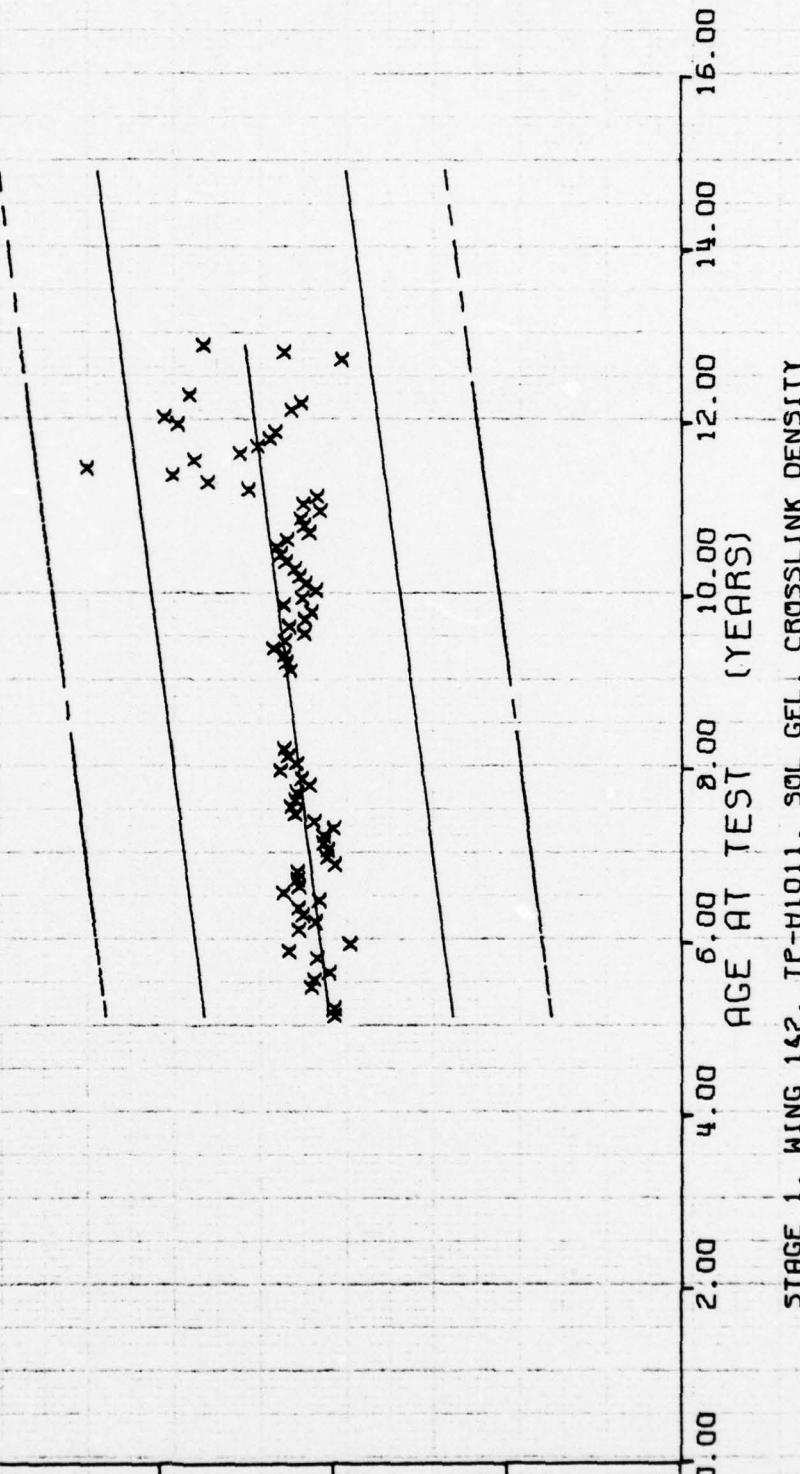
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
62, 0	4	91, 0	72	126, 0	28	155, 0	28
63, 0	8	92, 0	103	127, 0	45		
66, 0	4	93, 0	108	128, 0	24		
67, 0	8	94, 0	130	129, 0	21		
68, 0	12	95, 0	84	130, 0	8		
70, 0	4	96, 0	59	131, 0	20		
71, 0	4	97, 0	19	132, 0	16		
72, 0	3	98, 0	24	133, 0	12		
74, 0	36	99, 0	12	134, 0	4		
75, 0	8	110, 0	28	135, 0	8		
76, 0	16	111, 0	40	136, 0	36		
77, 0	20	112, 0	32	137, 0	32		
78, 0	36	113, 0	15	138, 0	20		
79, 0	20	114, 0	36	139, 0	28		
80, 0	36	115, 0	16	140, 0	40		
81, 0	56	116, 0	24	141, 0	28		
82, 0	40	117, 0	8	142, 0	52		
83, 0	24	118, 0	16	143, 0	40		
84, 0	24	119, 0	16	144, 0	12		
85, 0	24	120, 0	28	145, 0	12		
86, 0	32	121, 0	20	146, 0	12		
87, 0	20	122, 0	56	147, 0	4		
88, 0	4	123, 0	40	148, 0	4		
89, 0	72	124, 0	28	153, 0	12		
90, 0	68	125, 0	48	154, 0	36		

STAGE 1, WING 162, TP-H1011, SOL GEL, CROSSLINK DENSITY

This sample size summary is applicable to figure 42.

$Y = ((+5.6471693E-03) + (+4.1439889E-05) * X)$
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$
 $S = \text{SIGNIFICANCE OF } S = \text{SIGNIFICANT}$
 $S_e = +3.4399686E-03$
 $t = \text{SIGNIFICANT}$
 $t_e = +3.1446639E-06$
 $\text{DEGREES OF FREEDOM} = 2225$
 $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$
 $N = 2227$

$\text{UNIT OF MEASURE} = \text{MILLIEGKV/CC} * 10^{-1}$
 $\text{PARAMETER} = \text{CROSSLINK DENSITY}$



STAGE 1. WING 142. TP-H1011. SOL GEL, CROSSLINK DENSITY

Figure 42

*** SAMPLE SIZE SUMMARY ***

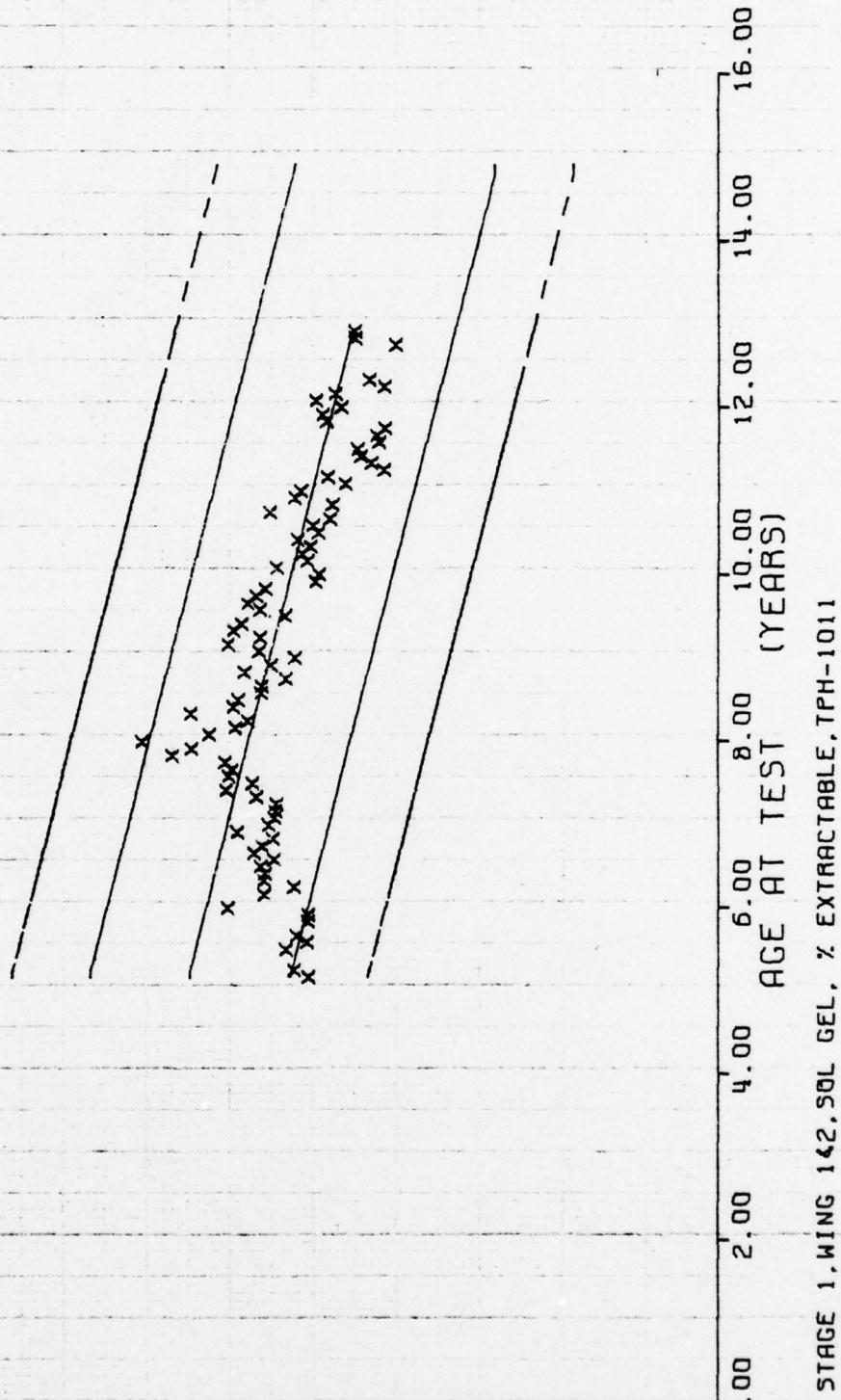
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
62, 0	4	91, 0	72	116, 0	40	141, 0	28
63, 0	6	92, 0	103	117, 0	12	142, 0	52
66, 0	4	93, 0	108	118, 0	32	143, 0	39
67, 0	8	94, 0	130	119, 0	16	144, 0	12
68, 0	12	95, 0	84	120, 0	28	145, 0	12
70, 0	4	96, 0	59	121, 0	20	146, 0	12
71, 0	4	97, 0	19	122, 0	56	147, 0	4
72, 0	3	98, 0	24	123, 0	40	148, 0	4
74, 0	36	99, 0	12	124, 0	28	153, 0	12
75, 0	8	100, 0	16	125, 0	48	154, 0	36
76, 0	16	101, 0	32	126, 0	27	155, 0	28
77, 0	20	102, 0	16	127, 0	44		
78, 0	36	103, 0	8	128, 0	24		
79, 0	20	104, 0	12	129, 0	21		
80, 0	36	105, 0	12	130, 0	8		
81, 0	56	106, 0	20	131, 0	20		
82, 0	40	107, 0	32	132, 0	16		
83, 0	24	108, 0	24	133, 0	12		
84, 0	24	109, 0	32	134, 0	4		
85, 0	24	110, 0	16	135, 0	8		
86, 0	32	111, 0	24	136, 0	36		
87, 0	20	112, 0	20	137, 0	32		
88, 0	4	113, 0	20	138, 0	19		
89, 0	72	114, 0	20	139, 0	28		
90, 0	68	115, 0	16	140, 0	40		

STAGE 1, WING 162, SOL GEL, % EXTRACTABLE, TPH-1011

This sample size summary is applicable to figure 43.

$\gamma = ((+9.6302156E+00) + (-2.1263271E-02) \times X)$
 $F = 1.0816657E+03$
 $R = -5.5658334E-01$
 $I = +3.2888687E+01$
 $N = 2412$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$
 $F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANT}$
 $I = \text{SIGNIFICANT}$
 $\text{DEGREES OF FREEDOM} = 2410$
 $\text{TEST CONDITIONS} = \text{AMB TEMP/RH}$

$\text{PARAMETER} = \% \text{ EXTRACTABLE}$
 $\text{UNIT OF MEASURE} = \%$
 0.00 2.00 4.00 6.00 8.00 10.00 12.00 14.00 16.00



STAGE 1, WING 142, SOL GEL, % EXTRACTABLE, TPH-1011

Figure 43

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
62, 0	4	91, 0	72	116, 0	48	141, 0	28
63, 0	8	92, 0	103	117, 0	12	142, 0	52
66, 0	4	93, 0	108	118, 0	32	143, 0	40
67, 0	8	94, 0	130	119, 0	16	144, 0	12
68, 0	12	95, 0	84	120, 0	28	145, 0	12
70, 0	4	96, 0	59	121, 0	20	146, 0	12
71, 0	4	97, 0	19	122, 0	56	147, 0	4
72, 0	3	98, 0	24	123, 0	40	148, 0	4
74, 0	36	99, 0	12	124, 0	28	153, 0	12
75, 0	8	100, 0	16	125, 0	48	154, 0	36
76, 0	16	101, 0	32	126, 0	27	155, 0	28
77, 0	20	102, 0	16	127, 0	44		
78, 0	36	103, 0	8	128, 0	24		
79, 0	20	104, 0	12	129, 0	21		
80, 0	36	105, 0	12	130, 0	8		
81, 0	56	106, 0	20	131, 0	20		
82, 0	40	107, 0	32	132, 0	16		
83, 0	24	108, 0	24	133, 0	12		
84, 0	24	109, 0	36	134, 0	4		
85, 0	24	110, 0	28	135, 0	8		
86, 0	32	111, 0	40	136, 0	36		
87, 0	20	112, 0	32	137, 0	32		
88, 0	4	113, 0	35	138, 0	19		
89, 0	72	114, 0	40	139, 0	28		
90, 0	63	115, 0	28	140, 0	40		

STAGE 1, WING 162, TP-H1011, SOL GEL, WT, SWELL RATIO

This sample size summary is applicable to Figure 44.

$Y = ((+4.1382588E+00) + (-9.6368579E-04) * X) * S$
 $S =$ SIGNIFICANCE OF $F =$ SIGNIFICANT
 $S =$ SIGNIFICANT OF $R =$ SIGNIFICANT
 $S =$ SIGNIFICANT OF $t =$ SIGNIFICANT
 $S =$ DEGREES OF FREEDOM = 2510
 $S =$ TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = RATE
 PARAMETER = WEIGHT SWELL
 30 3.10 3.98 4.78 5.58 6.38

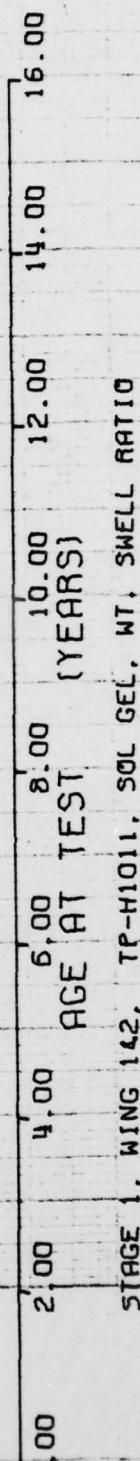


Figure 44

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
9.0	16	102.0	15	127.0	12	155.0	21
12.0	24	103.0	30	128.0	6	156.0	12
13.0	12	104.0	15	129.0	3	157.0	6
14.0	16	105.0	6	130.0	4	158.0	9
15.0	8	106.0	27	132.0	5	162.0	3
38.0	3	107.0	9	133.0	4	163.0	3
81.0	1	108.0	9	134.0	16	165.0	6
82.0	1	109.0	11	135.0	25	166.0	6
83.0	1	110.0	23	136.0	25	167.0	3
84.0	1	111.0	11	137.0	29		
85.0	2	112.0	17	138.0	43		
87.0	1	113.0	34	139.0	38		
88.0	1	114.0	11	140.0	21		
89.0	1	115.0	19	141.0	8		
90.0	5	116.0	18	142.0	8		
91.0	6	117.0	14	143.0	22		
92.0	5	118.0	17	144.0	11		
93.0	4	119.0	24	145.0	12		
95.0	4	120.0	18	146.0	21		
96.0	12	121.0	21	147.0	12		
97.0	16	122.0	6	148.0	17		
98.0	21	123.0	9	149.0	3		
99.0	9	124.0	17	150.0	6		
100.0	17	125.0	11	152.0	6		
101.0	14	126.0	3	154.0	24		

STAGE 1, WING 162, DTA, 12 DEG C RISE/MIN, ENDOTHERM

This sample size summary is applicable to figures 45 and 46.

$\gamma = ((+2.4830230E+02) + (-4.2013166E-02) * X)$
 $F =$ SIGNIFICANCE OF γ = SIGNIFICANT
 $R =$ SIGNIFICANCE OF R = SIGNIFICANT
 $L =$ SIGNIFICANCE OF L = SIGNIFICANT
 $N =$ DEGREES OF FREEDOM = 1044
 $R =$ +3.3453545E+02
 $R =$ -4.9262007E-01
 $R =$ +1.8290310E+01
 $R =$ 1046
 $N =$ STORAGE CONDITIONS = AMB TEMP/RH

TEST CONDITIONS = 12 DEG RISE/MIN

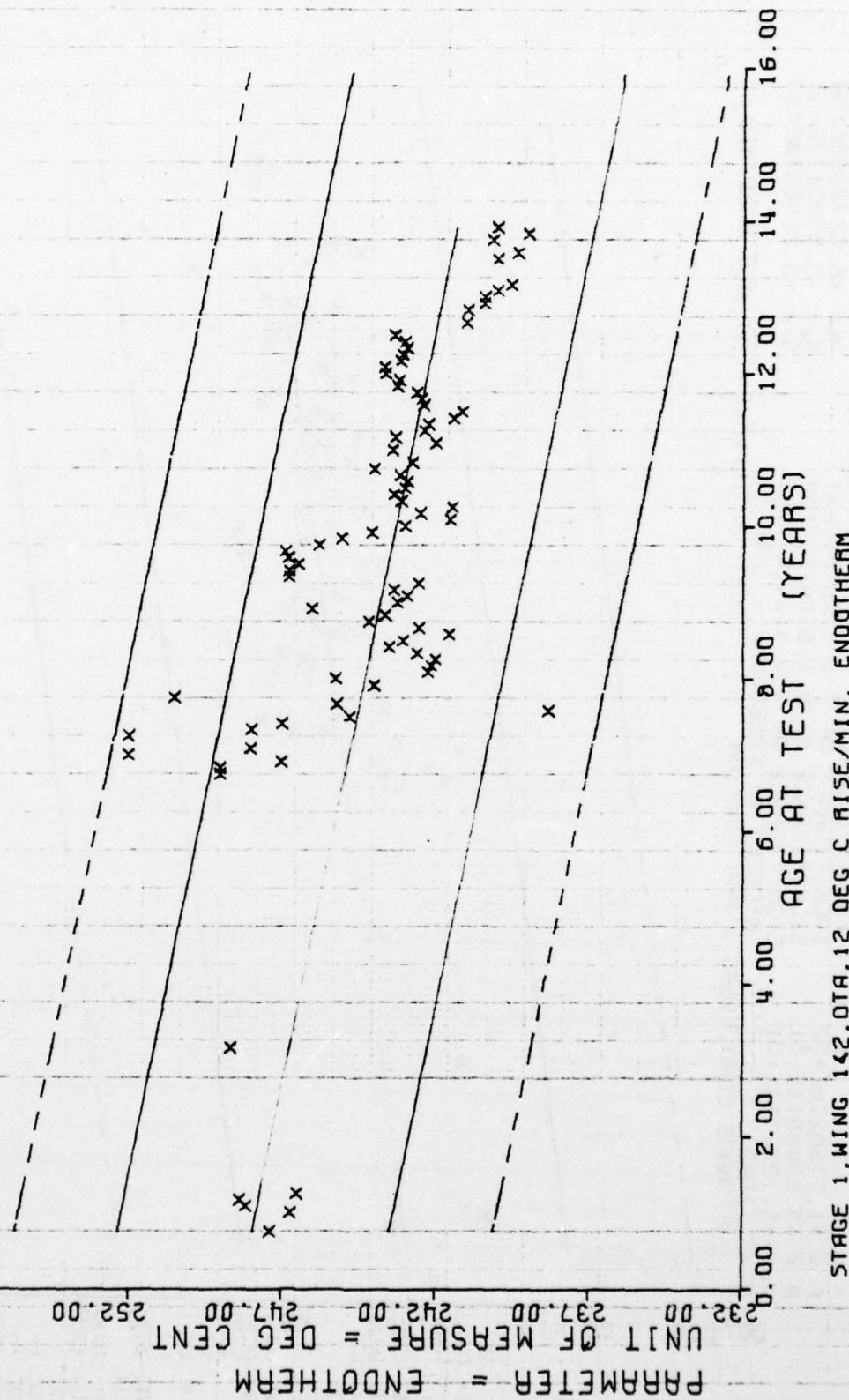


Figure 45

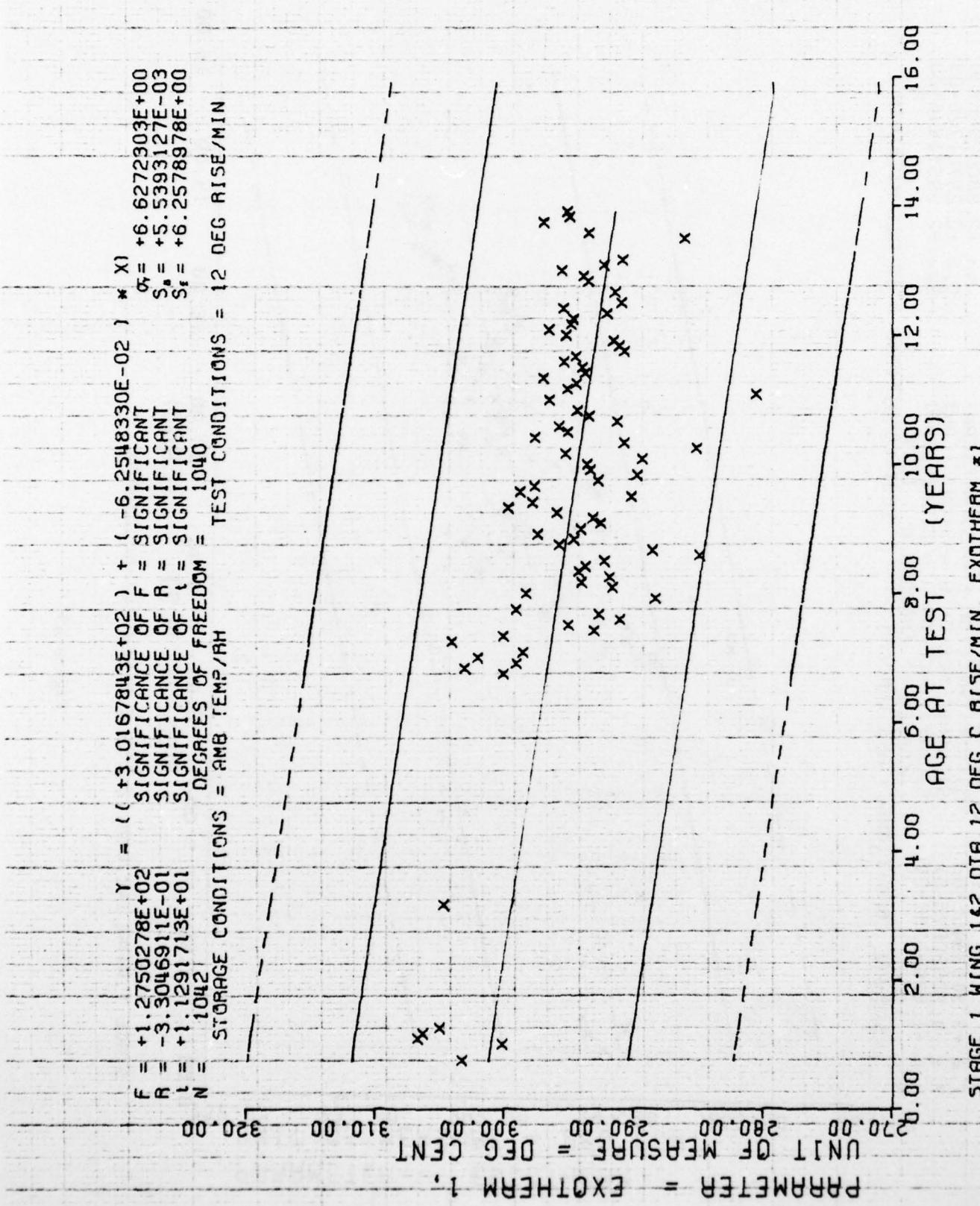


Figure 46

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
9, 0	16	102, 0	14	127, 0	12	155, 0	17
12, 0	24	103, 0	27	128, 0	6	156, 0	9
13, 0	12	104, 0	10	129, 0	3	157, 0	6
14, 0	16	105, 0	4	130, 0	4	158, 0	9
15, 0	8	106, 0	23	132, 0	5	162, 0	3
38, 0	3	107, 0	8	133, 0	4	163, 0	3
81, 0	1	108, 0	9	134, 0	16	165, 0	6
82, 0	1	109, 0	6	135, 0	23	166, 0	6
83, 0	1	110, 0	18	136, 0	25	167, 0	3
84, 0	1	111, 0	11	137, 0	29		
85, 0	2	112, 0	15	138, 0	42		
87, 0	1	113, 0	34	139, 0	37		
88, 0	1	114, 0	10	140, 0	20		
89, 0	1	115, 0	19	141, 0	8		
90, 0	4	116, 0	18	142, 0	8		
91, 0	2	117, 0	13	143, 0	20		
92, 0	5	118, 0	13	144, 0	11		
93, 0	4	119, 0	22	145, 0	12		
95, 0	4	120, 0	18	146, 0	21		
96, 0	6	121, 0	20	147, 0	12		
97, 0	10	122, 0	6	148, 0	17		
98, 0	14	123, 0	8	149, 0	3		
99, 0	6	124, 0	16	150, 0	6		
100, 0	12	125, 0	11	152, 0	6		
101, 0	13	126, 0	3	154, 0	20		

STAGE 1, WING 162.DTA, 12 DEG C RISE/MIN, EXOTHERM #2

This sample size summary is applicable to figure 47

$\gamma = ((+3.4172636E+02) + (-4.7405605E-02) * x) * 10^6$
 $F = +4.8341037E+01$ SIGNIFICANCE OF F = SIGNIFICANT
 $R = -2.1939027E-01$ SIGNIFICANCE OF R = SIGNIFICANT
 $\tau = +6.9527719E+00$ SIGNIFICANCE OF τ = SIGNIFICANT
 $N = 958$ DEGREES OF FREEDOM = 956
 STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = 12 DEG RISE/MIN

UNIT OF MEASURE = DEG CENT
 PARAMETER = EXOTHERM 2,
 300.00 320.00 340.00 360.00 380.00 400.00

- 74 -

STAGE 1, WING 142, DTA, 12 DEG C RISE/MIN, EXOTHERM #2

Figure 47

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
85.0	1	120.0	7	163.0	1
90.0	1	121.0	2		
92.0	2	123.0	3		
95.0	3	124.0	10		
96.0	2	125.0	1		
97.0	3	127.0	5		
98.0	3	128.0	6		
99.0	5	135.0	11		
100.0	2	136.0	9		
101.0	7	137.0	12		
103.0	9	138.0	19		
104.0	5	139.0	19		
106.0	5	140.0	10		
107.0	4	141.0	6		
109.0	7	142.0	6		
110.0	13	143.0	9		
111.0	5	144.0	8		
112.0	7	145.0	6		
113.0	11	146.0	6		
114.0	4	147.0	2		
115.0	5	148.0	8		
116.0	2	150.0	5		
117.0	5	152.0	2		
118.0	11	154.0	1		
119.0	10	158.0	5		

$\gamma = (+3.5951360E+02) + (+6.8747515E-02) * x$
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANCE OF } t = \text{SIGNIFICANT}$
 $N = \text{DEGREES OF FREEDOM} = 313$
 $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

UNIT OF MEASURE = DEG CENT
 PARAMETER = EXOTHERM 3,
 320.00 340.00 360.00 380.00 400.00 420.00

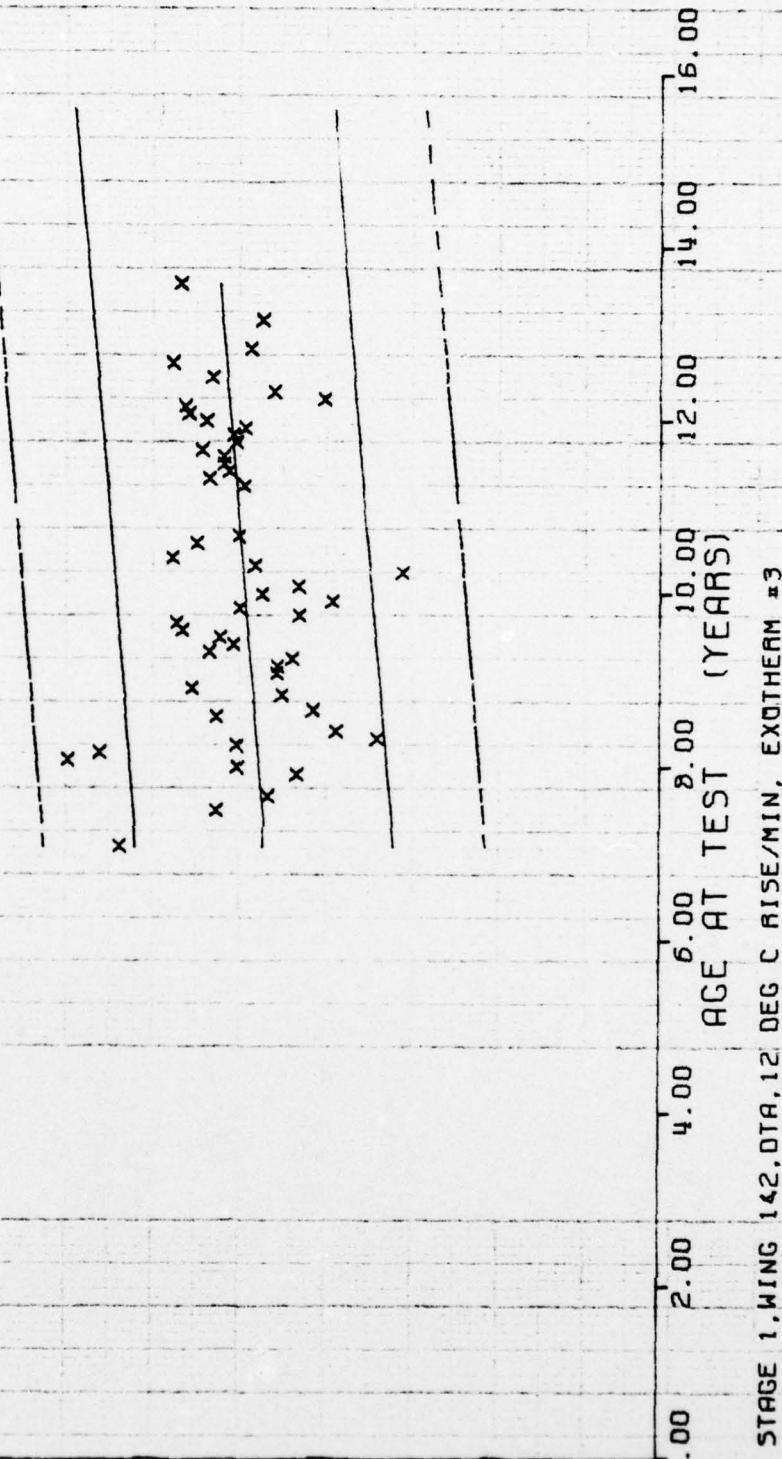


Figure 48

*** SAMPLE SIZE SUMMARY ***					
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
9.0	16	102.0	15	127.0	8
12.0	24	103.0	27	129.0	3
13.0	12	104.0	14	130.0	2
14.0	16	105.0	6	132.0	5
15.0	8	106.0	23	134.0	14
38.0	3	107.0	9	135.0	16
81.0	1	108.0	9	136.0	15
82.0	1	109.0	7	137.0	19
83.0	1	110.0	15	138.0	27
84.0	1	111.0	7	139.0	25
85.0	2	112.0	14	140.0	11
87.0	1	113.0	23	141.0	6
88.0	1	114.0	6	142.0	2
89.0	1	115.0	17	143.0	15
90.0	5	116.0	16	144.0	6
91.0	5	117.0	10	145.0	6
92.0	4	118.0	11	146.0	14
93.0	4	119.0	18	147.0	9
95.0	2	120.0	11	148.0	17
96.0	12	121.0	20	149.0	3
97.0	13	122.0	3	150.0	4
98.0	20	123.0	7	152.0	4
99.0	6	124.0	9	154.0	24
100.0	15	125.0	10	155.0	21
101.0	8	126.0	3	156.0	12

$F = +2.7393137E+00$ $\gamma = ((+3.7667115E+02) + (+2.6858185E-02) * x)$
 $R = +5.8055751E-02$ $F = \text{NOT SIGNIFICANT}$ $\sigma = +1.7662574E+01$
 $\sigma = +1.6227655E-02$ $R = \text{NOT SIGNIFICANT}$ $S = +1.6227655E-02$
 $\text{SIGNIFICANCE OF } R =$ NOT SIGNIFICANT $S_e = +1.7643664E+01$
 $\text{SIGNIFICANCE OF } \gamma =$ NOT SIGNIFICANT
 $\text{DEGREES OF FREEDOM} = 810$
 $N = 812$ $\text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$ $\text{TEST CONDITIONS} = 12 \text{ DEG RISE/MIN}$

$\text{UNIT OF MEASURE} = \text{DEG CENT}$
 $280.00 \quad 320.00 \quad 360.00 \quad 400.00 \quad 440.00 \quad 480.00$

PARAMETER = IGNITION TEMP

2.00 4.00 6.00 8.00 10.00 12.00 14.00 16.00

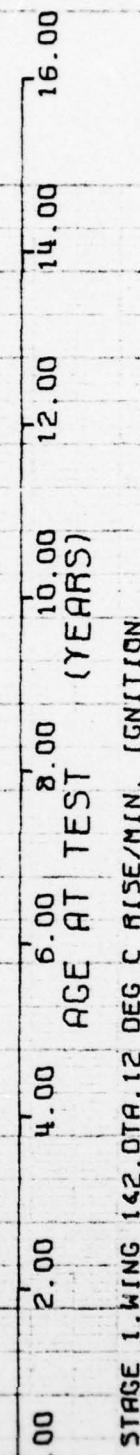


Figure 49

*** SAMPLE SIZE SUMMARY ***

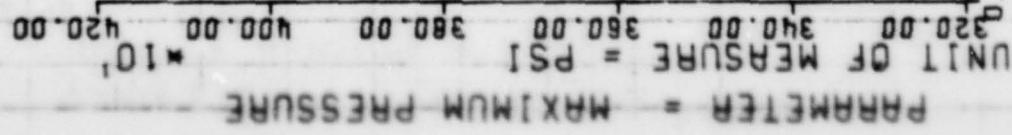
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
84.0	2	112.0	9	137.0	47	169.0	3
87.0	1	113.0	11	138.0	24	170.0	6
88.0	2	114.0	4	139.0	17	171.0	3
89.0	3	115.0	25	140.0	26	172.0	3
91.0	3	116.0	15	141.0	12		
92.0	1	117.0	17	142.0	24		
93.0	3	118.0	23	143.0	16		
94.0	2	119.0	18	144.0	7		
95.0	2	120.0	19	145.0	3		
96.0	1	121.0	12	146.0	4		
97.0	5	122.0	10	147.0	2		
98.0	5	123.0	36	148.0	2		
99.0	8	124.0	15	149.0	3		
100.0	7	125.0	30	153.0	3		
101.0	10	126.0	24	154.0	6		
102.0	6	127.0	39	155.0	15		
103.0	5	128.0	30	156.0	24		
104.0	10	129.0	27	157.0	3		
105.0	6	130.0	21	158.0	3		
106.0	7	131.0	27	160.0	3		
107.0	11	132.0	15	161.0	3		
108.0	16	133.0	11	163.0	6		
109.0	16	134.0	14	165.0	6		
110.0	4	135.0	19	166.0	6		
111.0	4	136.0	29	167.0	6		

STAGE 1, WING 1&2, PRESSURE TIME, MAXIMUM PRESSURE

This sample size summary is applicable to figures 50 and 51.

$\gamma = +1.0493737E+02$
 $F = +3.1904280E-01$
 $R = +1.0243894E+01$
 $t = 928$
 $N = 926$
 $F = \text{SIGNIFICANCE OF } F$
 $R = \text{SIGNIFICANCE OF } R$
 $t = \text{SIGNIFICANCE OF } t$
 $N = \text{DEGREES OF FREEDOM}$
 $\gamma = \text{STORAGE CONDITIONS} = \text{AMB TEMP/RH}$

$\gamma = (+3.3697632E+03) + (+3.1882157E+04) / X$
 $F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANT}$
 $t = \text{SIGNIFICANT}$
 $N = 926$
 $\gamma = \text{TEST CONDITIONS} = \text{TEST COND 500 PS}$

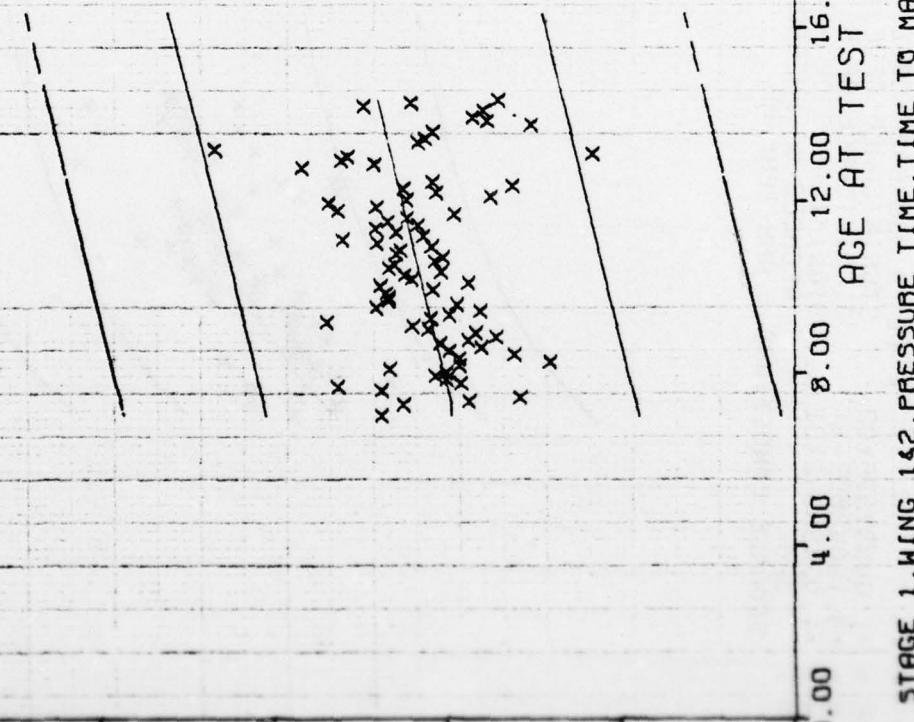


STAGE 1, WING 142, PRESSURE TIME, MAXIMUM PRESSURE

Figure 50

$\gamma = ((+6.0370125E-01) + (+3.9923188E-04) * X)$
 $F = \text{SIGNIFICANCE OF } F = \text{SIGNIFICANT}$
 $R = \text{SIGNIFICANCE OF } R = \text{SIGNIFICANT}$
 $S = \text{SIGNIFICANCE OF } S = \text{SIGNIFICANT}$
 $t = \text{DEGREES OF FREEDOM} = 925$
 $N = \text{STORAGE CONDITIONS} = \text{TEST COND 500 PS}$

PROGRAMMER = TIME TO MAX PRESS
 UNIT OF MEASURE = SECONDS
 0.48 0.58 0.64 0.72 0.80 0.88



STAGE 1, WING 1&2, PRESSURE TIME, TIME TO MAXIMUM PRESSURE

Figure 51

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
78.0	3	105.0	21	130.0	24	157.0	18
80.0	8	106.0	18	131.0	21	158.0	3
81.0	3	107.0	21	132.0	12	159.0	6
82.0	3	108.0	18	133.0	26	164.0	3
83.0	9	109.0	15	134.0	36	165.0	6
85.0	3	110.0	9	135.0	9	166.0	3
86.0	3	111.0	24	136.0	51	167.0	3
87.0	9	112.0	27	137.0	45	168.0	6
88.0	12	113.0	21	138.0	33	169.0	6
89.0	6	114.0	21	139.0	48		
90.0	3	115.0	18	140.0	18		
91.0	14	116.0	18	141.0	18		
92.0	15	117.0	45	142.0	9		
93.0	3	118.0	21	143.0	9		
94.0	3	119.0	18	144.0	9		
95.0	12	120.0	24	145.0	6		
96.0	9	121.0	18	146.0	3		
97.0	3	122.0	18	148.0	9		
98.0	9	123.0	26	149.0	3		
99.0	6	124.0	21	150.0	3		
100.0	6	125.0	24	152.0	6		
101.0	15	126.0	35	153.0	15		
102.0	18	127.0	23	154.0	6		
103.0	18	128.0	36	155.0	12		
104.0	15	129.0	29	156.0	15		

STAGE 1, WING AGE, TP-H1011, BURNING RATE 1000 PSI

This sample size summary is applicable to figure 52.

AD-A056 729

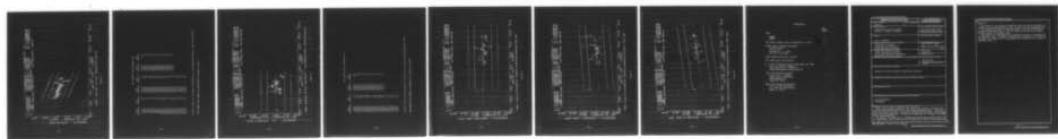
OGDEN AIR LOGISTICS CENTER HILL AFB UTAH PROPELLANT L--ETC F/G 21/9.2
PROPELLANT SURVEILLANCE REPORT. LGM-30 A AND B STAGE 1. TP-H101--ETC(U)

UNCLASSIFIED

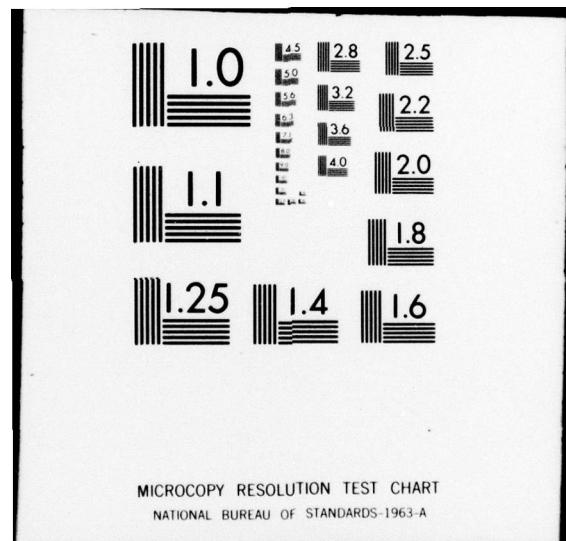
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$\gamma = ((+3.5642491E-01) + (-3.7065042E-04) * X)$
 $F = \text{SIGNIFICANCE OF }$
 $R = \text{SIGNIFICANCE OF }$
 $t = \text{SIGNIFICANCE OF }$
 $t^2 = \text{DEGREES OF FREEDOM} = 1275$
 $N = \text{STORAGE CONDITIONS = AMB TEMP/RH}$

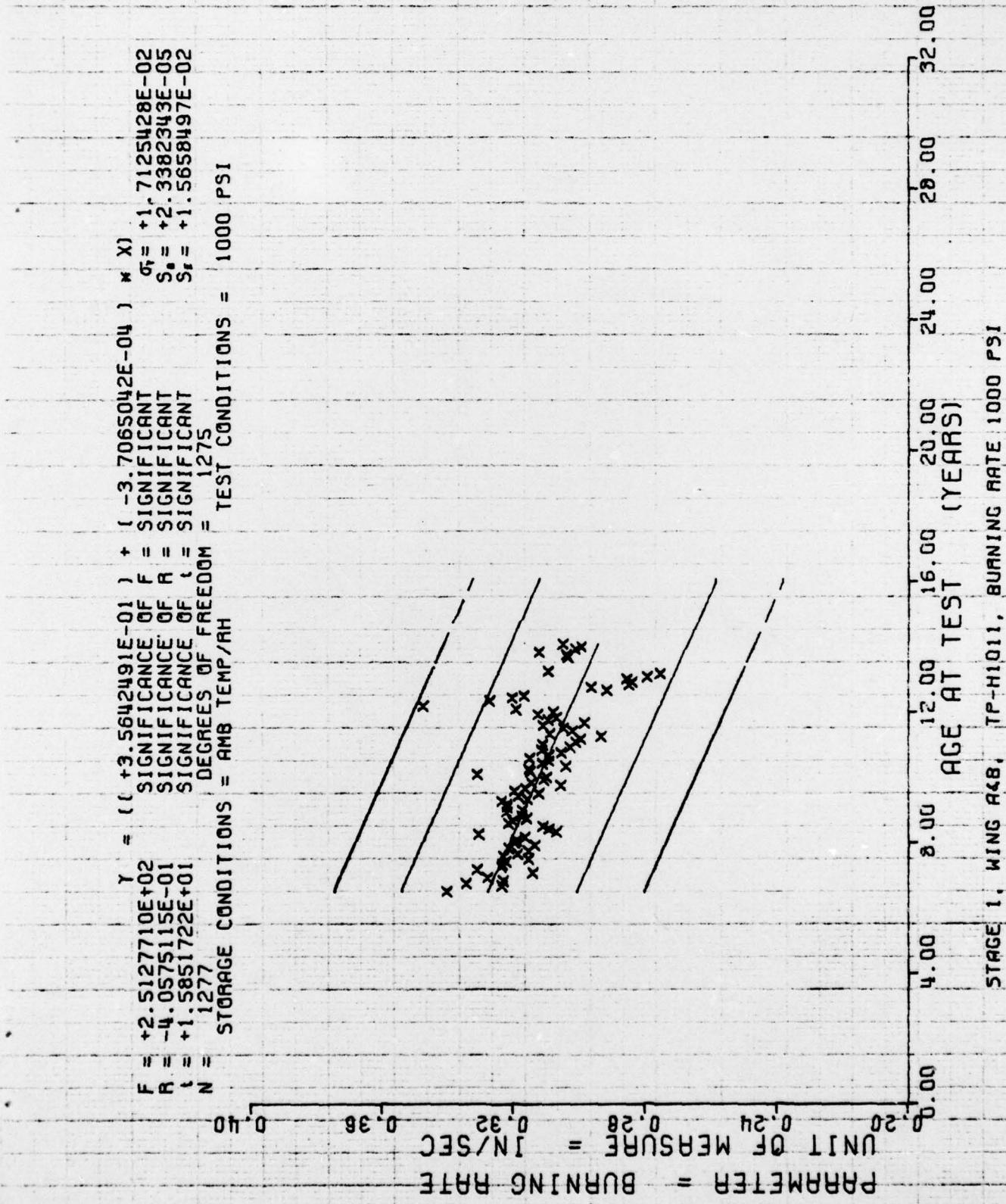


Figure 52

*** SAMPLE SIZE SUMMARY ***

AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
12.0	10	114.0	12	139.0	11
88.0	1	115.0	6	140.0	4
91.0	2	116.0	10	141.0	5
92.0	4	117.0	6	142.0	5
93.0	1	118.0	6	143.0	7
94.0	3	119.0	12	144.0	5
95.0	3	120.0	8	145.0	2
96.0	3	121.0	8	146.0	3
97.0	4	122.0	9	148.0	2
98.0	4	123.0	12	159.0	1
99.0	10	124.0	5	161.0	1
100.0	7	125.0	11	162.0	1
101.0	10	126.0	9	163.0	1
102.0	8	127.0	11	165.0	2
103.0	9	128.0	8	166.0	2
104.0	9	129.0	4	168.0	1
105.0	6	130.0	6	169.0	1
106.0	10	131.0	7	171.0	2
107.0	7	132.0	5		
108.0	12	133.0	9		
109.0	6	134.0	11		
110.0	8	135.0	8		
111.0	8	136.0	8		
112.0	12	137.0	8		
113.0	7	138.0	15		

STAGE 1 WING 1E2 TP-H1011 IGNITABILITY. IGN THRESHOLD POINT. 168 CAL/SQ CM/SEC

This sample size summary is applicable to figures 53 and 54.

$F = +7.0659453E-02$
 $R = -1.2788150E-02$
 $t = +2.6581845E-01$
 $N = 434$
 STORAGE CONDITIONS = AMB TEMP/RH

$\gamma = ((+6.1921518E+01) + (-5.0982735E-03) * X)$
 SIGNIFICANCE OF F = NOT SIGNIFICANT
 SIGNIFICANCE OF R = NOT SIGNIFICANT
 SIGNIFICANCE OF t = NOT SIGNIFICANT
 DEGREES OF FREEDOM = 432
 TEST CONDITIONS = 168 CAL/SQCM/SEC

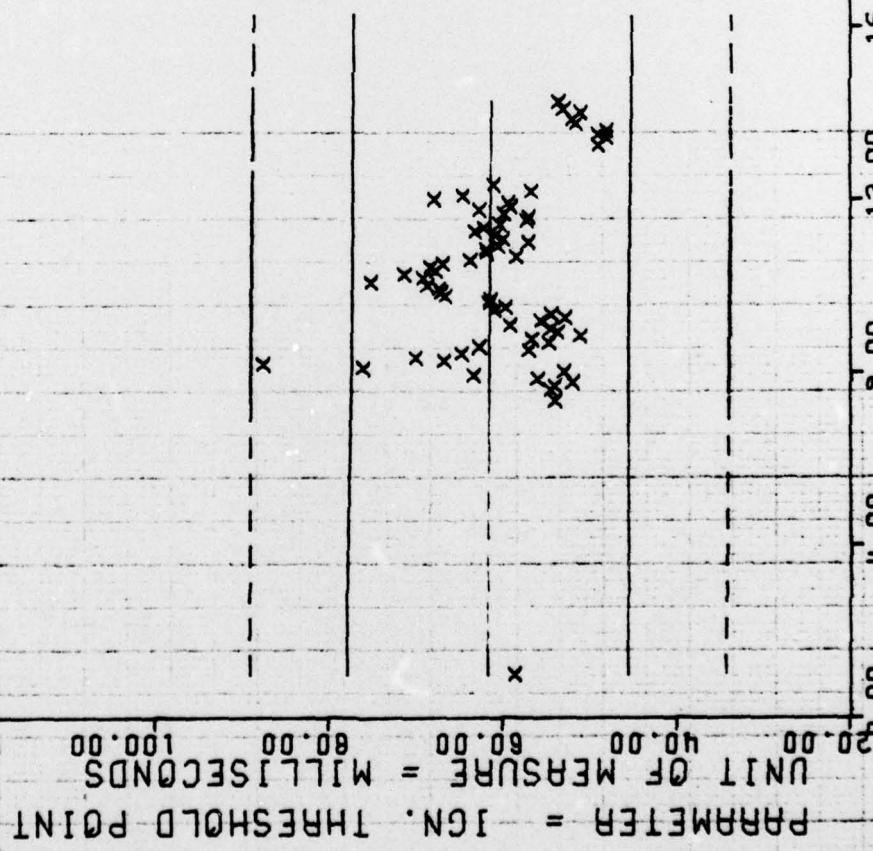


Figure 53

*** SAMPLE SIZE SUMMARY ***

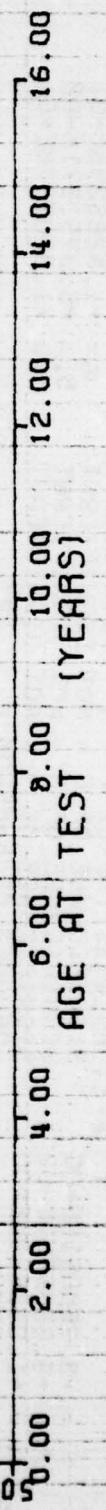
AGE (MONTHS)	NR SAMPLES	AGE (MONTHS)	NR SAMPLES
48.0	3	140.0	23
105.0	12	141.0	15
106.0	6	142.0	9
108.0	24	143.0	3
110.0	12	144.0	3
112.0	12	145.0	5
114.0	6	146.0	7
115.0	12	147.0	6
116.0	15	148.0	3
120.0	5	156.0	3
121.0	9	159.0	9
124.0	16	163.0	3
125.0	15	164.0	3
126.0	3	165.0	4
127.0	21	166.0	1
128.0	19	167.0	6
129.0	11	168.0	5
130.0	9		
131.0	9		
132.0	12		
135.0	7		
136.0	11		
137.0	14		
138.0	6		
139.0	22		

STAGE I WING 162 DIFFERENTIAL SCANNING CALORIMETER ENDOTHERM PEAK TEMP

This sample size summary is applicable to figures 55 and 56.

$Y = ((+5.1755606E+02) + (-1.9044618E-04) \times X)$
 $F = +1.294722E-03$ SIGNIFICANCE OF F = NOT SIGNIFICANT
 $R = -1.8058943E-03$ SIGNIFICANCE OF R = NOT SIGNIFICANT
 $s = +3.5982249E-02$ SIGNIFICANCE OF s = NOT SIGNIFICANT
 $N = 399$ DEGREES OF FREEDOM = 397
 $N =$ STORAGE CONDITIONS = AMB TEMP/RH TEST CONDITIONS = AMB TEMP/RH

UNIT OF MEASURE = DEG K
 PARAMETER = ENDOOTHERM PER K TEMP



STAGE I WING 142 DIFFERENTIAL SCANNING CALORIMETER ENDOOTHERM PEAK TEMP

Figure 54

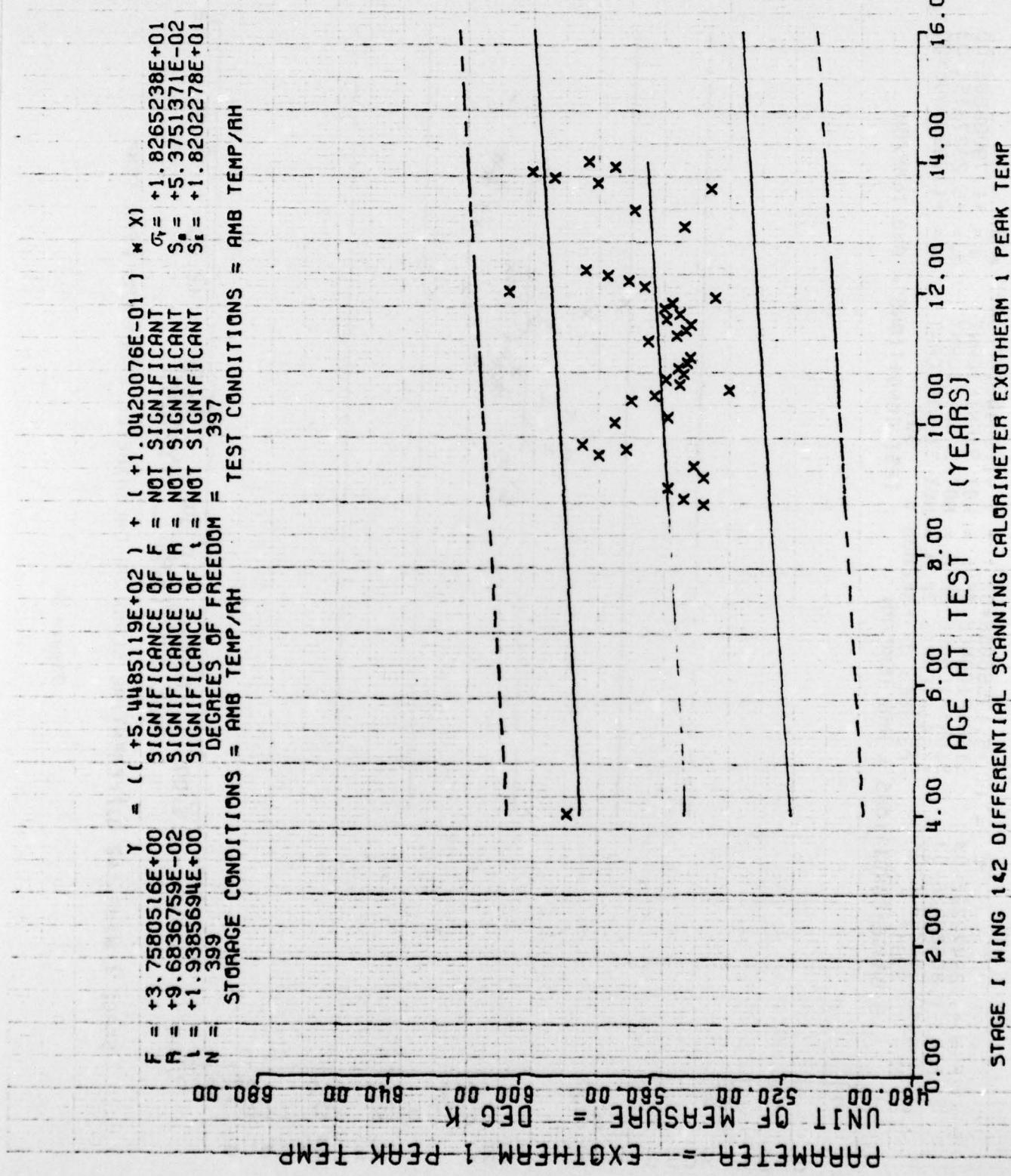
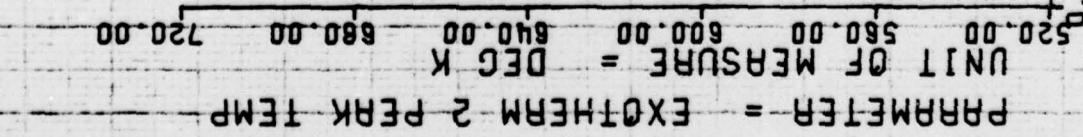


Figure 55

$\gamma = 1.2673663E+01$ $(+5.7944381E+02)$ $(+2.0886812E-01)$ $* X$
 $F = 1.2673663E+01$ SIGNIFICANCE OF $F =$ SIGNIFICANT
 $R = +1.7588632E-01$ SIGNIFICANCE OF $R =$ SIGNIFICANT
 $\sigma = +3.5600089E+00$ SIGNIFICANCE OF $\sigma =$ SIGNIFICANT
 $N = 399$ DEGREES OF FREEDOM = 397

TEST CONDITIONS = AMB TEMP/RH
STORAGE CONDITIONS = AMB TEMP/RH



STAGE 1 WING 142 DIFFERENTIAL SCANNING CALORIMETER EXOTHERM 2 PEAK TEMP

Figure 56

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1. REPORT NUMBER 393(78)	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
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7. AUTHOR(s) John A. Thompson		6. PERFORMING ORG. REPORT NUMBER
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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This report contains propellant test results from cartons of TP-H1011 bulk propellant representing LGM-30 A & B First Stage Minuteman Motors. This report is the fourteenth time that a statistical approach has been used to analyze First Stage bulk carton propellant. Testing was accomplished in accordance with MMEMP Project M82934C-WNL17514. The purpose of testing was to determine and provide early warning of any serious degradation trends occurring in the propellant for service life predictions.		

20 (con't)

An analysis of all parameters indicates that no potential problems are expected in the propellant for at least two years past the oldest data point.

Data stored in the G085 System were plotted utilizing the IBM 360-65 Computer and CAL-COMP Plotter. The data range at any age can be found by suitable inquiry of the G085 System.

Each point on the regression plot represents the mean of all samples at that particular age. The number of specimens at each point is indicated on the sample size summary sheet accompanying each regression plot or group of regression plots.